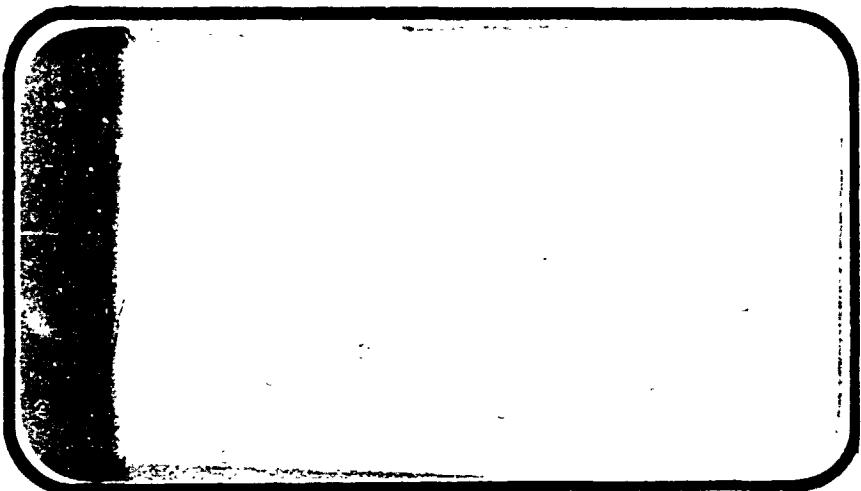




NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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ON A 0.010-SCALE MODEL OF THE
CONFIGURATION 3 SPACE SHUTTLE ORBITER AND
EXTERNAL TANK IN THE NASA/AMES (Chrysler
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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER
HOUSTON, TEXAS

DATA MANAGEMENT services
SPACE DIVISION  CHRYSLER
CORPORATION

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RESULTS OF INVESTIGATIONS ON A 0.010-SCALE MODEL
OF THE CONFIGURATION 3 SPACE SHUTTLE ORBITER
AND EXTERNAL TANK IN THE NASA/AMES RESEARCH CENTER
3.5-FOOT HYPERSONIC WIND TUNNEL (IA15)

By

M. T. Petrozzi and M. D. Milam, Rockwell International
J. A. Mellenthin, NASA Ames

Prepared under NASA Contract Number NAS9-13247

By

Data Management Services
Chrysler Corporation Space Division
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

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ABSTRACT

Experimental aerodynamic investigations were conducted in the NASA/Ames 3.5-Foot Hypersonic Wind Tunnel during the period of October 10 through October 15, 1973. The model used for this test was a 0.010-scale of the Configuration 3 Space Shuttle Orbiter and the External Tank.

Six-component aerodynamic force and moment data were recorded over an angle of attack range from -8° to $+30^\circ$ at 0° and 5° angles of sideslip. Data was also recorded during beta sweeps of -8° to $+10^\circ$ at angles of attack of -10° , 0° , and 30° . All testing was done at Mach 7.3.

Various elevon, rudder and orbiter to external tank attaching structures and fairings were tested to determine longitudinal and lateral-directional stability characteristics. Non-metric exhaust plumes were installed during a portion of the testing to determine the effects of the main propulsion system rocket plumes.

Base pressures on the external tank, which were monitored through tubing internally routed through the external tank and orbiter, were found to be questionable during the first 10 runs. Externally mounted tubing was installed prior to run 11 and the test series were completed using

* NASA/Ames

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that tubing configuration for the external tank base pressures. Subsequent to run 18, when the non-metric plumes were installed, all base pressures were monitored through externally routed tubing. See the DATA REDUCTION section for additional information.

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PLOT SCHEDULES:

- (A) CA, CN, CLM vs ALPHA, CN vs CLM
- (B) CY, CBL, CYN vs BETA, CY vs CYN
- (C) CY, CBL, CIN vs ALPHA
- (D) CN/A, CLM/A, XAC/LR vs ALPHA
- (E) CY/B, CBL/B, CYN/B, YAC/L vs BETA
- (F) DCN/DE, DCA/DE, DCIMDE vs ALPHA
- (G) DCY/DA, DCBLDA, DCYNDL vs ALPHA
- (H) DCY/DR, DCBLDR, DCYNDR vs BETA
- (I) DCY/DB, DCBLDB, DCYNDB, YAC/L vs ALPHA
- (J) DCY/DR, DCBLDR, DCYNDR vs ALPHA

NOMENCLATURE
General

| <u>SYMBOL</u> | <u>SADSAC SYMBOL</u> | <u>DEFINITION</u> |
|---------------|----------------------|--|
| a | | speed of sound; m/sec, ft/sec |
| C_p | CP | pressure coefficient; $(p_1 - p_\infty)/q$ |
| M | MACH | Mach number; V/a |
| p | | pressure; N/m ² , psf |
| q | Q(NSM) Q(PSF) | dynamic pressure; $1/2\rho V^2$, N/m ² , psf |
| RN/L | RN/L | unit Reynolds number; per m, per ft |
| v | | velocity; m/sec, ft/sec |
| α | ALPHA | angle of attack, degrees |
| β | BETA | angle of sideslip, degrees |
| ψ | PSI | angle of yaw, degrees |
| ϕ | PHI | angle of roll, degrees |
| ρ | | mass density; kg/m ³ , slugs/ft ³ |

Reference & C.G. Definitions

| | | |
|-------------------|------|---|
| Ab | | base area; m ² , ft ² |
| b | BREF | wing span or reference span; m, ft |
| c.g. | | center of gravity |
| $\frac{l}{c}$ REF | LREF | reference length or wing mean aerodynamic chord; m, ft |
| S | SREF | wing area or reference area; m ² , ft ² |
| | MRP | moment reference point |
| | XMRP | moment reference point on X axis |
| | YMRP | moment reference point on Y axis |
| | ZMRP | moment reference point on Z axis |

SUBSCRIPTS

| | |
|----------|-------------------|
| b | base |
| i | local |
| s | static conditions |
| t | total conditions |
| ∞ | free stream |

NOMENCLATURE (Continued)

Body-Axis System

| <u>SYMBOL</u> | <u>SADSAC SYMBOL</u> | <u>DEFINITION</u> |
|------------------------------|----------------------|---|
| C_N | CN | normal-force coefficient; $\frac{\text{normal force}}{qS}$ |
| C_A | CA | axial-force coefficient; $\frac{\text{axial force}}{qS}$ |
| C_Y | CY | side-force coefficient; $\frac{\text{side force}}{qS}$ |
| C_{A_b} | CAB | base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_b - p_\infty)/qS$ |
| C_{A_f} | CAF | forebody axial force coefficient, $C_A - C_{A_b}$ |
| C_m | CLM | pitching-moment coefficient; $\frac{\text{pitching moment}}{qS_{\text{REF}}}$ |
| C_n | CYN | yawing-moment coefficient; $\frac{\text{yawing moment}}{qS_b}$ |
| C_l | CBL | rolling-moment coefficient; $\frac{\text{rolling moment}}{qS_b}$ |
| <u>Stability-Axis System</u> | | |
| C_L | CL | lift coefficient; $\frac{\text{lift}}{qS}$ |
| C_D | CD | drag coefficient; $\frac{\text{drag}}{qS}$ |
| C_{D_b} | CDB | base-drag coefficient; $\frac{\text{base drag}}{qS}$ |
| C_{D_f} | CDF | forebody drag coefficient; $C_D - C_{D_b}$ |
| C_Y | CY | side-force coefficient; $\frac{\text{side force}}{qS}$ |
| C_m | CIM | pitching-moment coefficient; $\frac{\text{pitching moment}}{qS_{\text{REF}}}$ |
| C_n | CIN | yawing-moment coefficient; $\frac{\text{yawing moment}}{qS_b}$ |
| C_l | CSL | rolling-moment coefficient; $\frac{\text{rolling moment}}{qS_b}$ |
| L/D | L/D | lift-to-drag ratio; C_L/C_D |

NOMENCLATURE (Continued)

ADDITIONS TO STANDARD NOMENCLATURE

| <u>SYMBOL</u> | <u>SADSAC SYMBOL</u> | <u>DEFINITION</u> |
|---------------------|----------------------|---|
| δ_a | AILRON | aileron, total aileron deflection angle, degrees, (left aileron-right aileron)/2 |
| δ_e | ELEVON | elevon, surface deflection angle, positive deflection, trailing edge down; degrees |
| δ_r | RUDDER | rudder, surface deflection angle, positive deflection, trailing edge to the left; degrees |
| PL_1 | PLUMES | solid plumes (Ref. figure 2a), as a parameter PLUMES = 1 (plumes on), PLUMES = 0 (plumes off) |
| $C_{N\alpha}$ | CN/A | normal force coefficient derivative with ALPHA, per degree |
| $C_{m\alpha}$ | CLM/A | pitching moment coefficient derivative with ALPHA, per degree |
| X_{ac}/ℓ_{REF} | XAC/LR | pitch aerodynamic center, $-(CLM/A)/(CN/A)$ |
| $C_Y\beta$ | CY/B | side force coefficient derivative with BETA, per degree |
| $C_{l\beta}$ | CBL/B | rolling moment coefficient derivative with BETA, per degree |
| $C_{n\beta}$ | CYN/B | yawing moment coefficient derivative with BETA, per degree |
| Y_{ac}/ℓ_{REF} | YAC/L | yaw aerodynamic center $-(CYN/B)/(CY/B)$ for β sweeps, $-(DCYNDB)/(DCY/DB)$ for α sweeps |
| $C_{N\delta_e}$ | DCN/DE | normal force coefficient due to ELEVON, per degree |
| $C_{A\delta_e}$ | DCA/DE | axial force coefficient due to ELEVON, per degree |
| $C_{m\delta_e}$ | DCLMDE | pitching moment coefficient due to ELEVON, per degree |
| $C_{n\delta_a}$ | DCYNDA | yawing moment due to AILERON, per degree, (body axis) |

NOMENCLATURE (Concluded)

| <u>SYMBOL</u> | <u>SADSAC SYMBOL</u> | <u>DEFINITION</u> |
|--------------------|----------------------|--|
| $C_l \delta_a$ | DCBLDA | rolling moment due to aileron, per degree, (body axis) |
| $C_Y \delta_a$ | DCY/DA | side force due to aileron, per degree, (body axis) |
| $C_n \delta_r$ | DCYNDR | yawing moment due to RUDDER, per degree, (body axis) |
| $C_l \delta_r$ | DCBLDR | rolling moment due to RUDDER, per degree, (body axis) |
| $C_Y \delta_r$ | DCY/DR | side force due to RUDDER, per degree |
| $C_n \delta_\beta$ | DCYNDB | yawing moment due to BETA, per degree |
| $C_l \delta_\beta$ | DCBLDB | rolling moment due to BETA, per degree |
| $C_Y \delta_\beta$ | DCY/DB | side force due to BETA, per degree |

CONFIGURATIONS INVESTIGATED

The following summarizes configurations investigated and nomenclature used to designate their model components:

$OT = B_{19} C_7 E_{23} F_5 M_4 N_{24} N_8 R_5 V_7 W_{107} T_{10}$

$P_1 = PT_4 PT_5 PT_6; A_1 = AT_6 AT_7 AT_{11};$

$L = FL_3 FL_4; F = FR_1$

| <u>Config. Symbol</u> | <u>Component Description</u> | <u>Drawing Lines</u> |
|---------------------------|------------------------------|--------------------------|
| B_{19} | Body | VL70-000139 |
| C_7 | Canopy | VL70-000139 |
| E_{23} | Elevons | VL70-000139 |
| F_5 | Body Flap | VL70-000139 |
| M_4 | Orbital Maneuvering System | VL70-000139 |
| N_{24} | Orbiter SSME Nozzles | VL70-000140A |
| N_8 | OMS Nozzles | VL70-000089B |
| R_5 | Rudder | VL70-000140A |
| V_7 | Vertical Tail | VL70-000139 |
| W_{107} | Wing | VL70-000139B |
| T_{10} | External Tank | VL78-000041B |

| | | |
|-----------|--|-----------------------------|
| S_{12} | Boosters (Solid Rocket) | VL77-000036A |
| PL_1 | Solid Plumes | Defined in Model Dimen. |
| | Attach Structure (Simulated) | |
| AT_{11} | Front Orbiter to External Tank | VL72-000088D & 89 |
| AT_6 | Left Rear Orbiter to External Tank | VL72-000088D & 89 |
| AT_7 | Right Rear Orbiter to External Tank | VL72-000088D & 89 |
| AT_8 | Front SRB to External Tank | VL72-000106 |
| AT_9 | Rear SRB to External Tank | VL72-000106 |
| PT_4 | LO_2 Vent Line Fairing | VL78-000031A |
| PT_5 | LO_2 Feed Line | VL78-000031A |
| PT_6 | LH_2 Vent Line | VL78-000031A |
| | Feed Lines (from External Tank to Orbiter) | |
| FL_3 | LO_2 Feedlines | VL78-000050 |
| FL_4 | LH_2 Feedline | VL78-000050 |
| | SRB Protuberances | |
| PS_1 | Electrical Tunnel Fairing | Sketch "SRB Electr. Tunnel" |
| PS_2 | Attach Ring | VL77-000036A |
| PS_3 | Separation Rocket Fairing | VL77-000036A |
| FR_1 | Umbilical Door Fairing | VL78-000050 |

TEST FACILITY

The NASA-Ames 3.5-Foot Hypersonic Wind Tunnel is a closed-circuit, blowdown-type tunnel capable of operating at nominal Mach numbers of 5, 7, and 10 at pressures to 1800 psia and temperatures to 3400°R for run times to four minutes. The major components of the facility include a gas storage system where the test gas is stored at 3000 psi, a storage heater filled with aluminum-oxide pebbles capable of heating the test gas to 3400°R, axisymmetric contoured nozzles with exit diameters of 42 inches for generating the desired Mach number, and a 900,000 ft³ vacuum storage system which operates to pressures of 0.3 psia. The test section itself is an open-jet type enclosed within a chamber approximately 12-feet in diameter and 40-feet in length, arranged transversally to the flow direction.

A model support system is provided that can pitch models through an angle-of-attack range of -20 to +18 degrees, in a vertical plane, about a fixed point of rotation on the tunnel centerline. This rotation point is adjustable from 1 to 5 feet from the nozzle exit plane. The model normally is out of the test stream (strut centerline 37-inches from tunnel centerline) until the tunnel test conditions are established after which it is inserted. Insertion time is adjustable to as little as 1/2 second and models may be inserted at any strut angle.

A high-speed, analog-to-digital data acquisition system is used to record test data on magnetic tape. The present system is equipped to measure and record the outputs from 80 transducers in addition to 20 channels of tunnel parameters.

DATA REDUCTION

Aerodynamic forces and moments have been reduced to coefficient form based on the following reference values:

s_{ref} = total theoretical wing projected area = 0.2690 ft²

l_{ref} = body length = 12.903 in

b_{ref} = total wing span = 9.3668 in

The moments have been reduced about a reference moment center located at the external tank STA 9.89 (this is orbiter STA 2.38) on the external tank center line.

All data are corrected for model base pressure effects. The groupings of the manifolded pressures along with their designated symbols of the effective base area, magnitude of the base areas, pressure coefficient symbols, base/cavity axial-force coefficient definitions, and the orifice number assignments are listed as follows:

Runs 1 through 18:

| <u>Base Area Name</u> | <u>Area Desig.</u> | <u>Area, Numerical Value - Sq. in</u> | <u>Pressure Coefficient Symbol</u> | <u>Pressure Orifice Number(s)</u> |
|---------------------------|--------------------|---------------------------------------|------------------------------------|-----------------------------------|
| Orbiter Upper Base | A_{BU} | 1.31 | $C_{P_{BU}}$ | 3 |
| Orbiter Lower Base | A_{BL} | 1.97 | $C_{P_{BL}}$ | 4 |
| OMS Upper (Recessed) Base | A_{OU} | 0.80 | $C_{P_{OU}}$ | 1 |
| OMS Lower (Extended) Base | A_{OL} | 0.52 | $C_{P_{OL}}$ | 2 |

| | | | | |
|---------------------------|----------|------|------------|---|
| Orbiter Balance Cavity | A_{BC} | 1.78 | C_P_{BC} | 7 |
|---------------------------|----------|------|------------|---|

| | | | | |
|------------------------------|-----------|------|-----------------|---|
| External Tank Base, inner | A_{ETI} | 3.99 | $C_P_{B_{ETI}}$ | 5 |
|------------------------------|-----------|------|-----------------|---|

| | | | | |
|------------------------------|----------------|------|-----------------|---|
| External Tank Base, outer | $A_{BC_{ETO}}$ | 4.32 | $C_P_{B_{ETO}}$ | 6 |
|------------------------------|----------------|------|-----------------|---|

Runs subsequent to no.18:

| | | | | |
|-----------------------|----------|------|------------|---|
| Orbiter Upper Base | A_{BU} | 2.30 | C_P_{BU} | 1 |
|-----------------------|----------|------|------------|---|

| | | | | |
|-----------------------|----------|------|------------|---|
| Orbiter Lower Base | A_{BL} | 2.30 | C_P_{BL} | 2 |
|-----------------------|----------|------|------------|---|

| | | | | |
|---------------------------|----------|------|------------|---|
| Orbiter Balance Cavity | A_{BC} | 1.78 | C_P_{BC} | 3 |
|---------------------------|----------|------|------------|---|

| | | | | |
|------------------------------|-----------|------|-----------------|---|
| External Tank Base, inner | A_{ETI} | 3.99 | $C_P_{B_{ETI}}$ | 4 |
|------------------------------|-----------|------|-----------------|---|

| | | | | |
|------------------------------|----------------|------|-----------------|---|
| External Tank Base, outer | $A_{BC_{ETO}}$ | 4.32 | $C_P_{B_{ETO}}$ | 5 |
|------------------------------|----------------|------|-----------------|---|

TABLE I.

TABLE II.
TEST : ARC 3.5-175 (IA 15) DATA SET/RUN NUMBER COLLATION SUMMARY

TEST: ARC 3.5-175 (IA 15)

DATA SET/RUN NUMBER COLLATION SUMMARY

DATE : 15 OCTOBER 73

| TEST RUN NUMBERS | | | | | | | | | |
|---------------------|---------------------------|--------|--------------------|----------|--------------|--|--|--|--|
| DATA SET IDENTIFIER | CONFIGURATION | SCHED. | PARAMETERS/VALUES | NO. RUNS | MACH NUMBERS | | | | |
| REG002 | $\phi T + L + F, 1/1, +F$ | C 0 | Se 0, SR Pla 0 off | 1 | 2 | | | | |
| 003 | +F | T 0 | 0 0, T -20 | 1 | 3 | | | | |
| 004 | +F | T 0 | 0 -40, 0 | 1 | 4 | | | | |
| 005 | +F | T 0 | +15 | 1 | 5 | | | | |
| 006 | | T 0 | +15 | 1 | 6 | | | | |
| 007 | | T 0 | -40 | 1 | 7 | | | | |
| 008 | | T 0 | -20 | 1 | 8 | | | | |
| 009 | | T 0 | 10 | 1 | 9 | | | | |
| 010 | | T 0 | 0 | 1 | 10 | | | | |
| 011 | | T 0 | 0 | 1 | 11 | | | | |
| 012 | +F | T 5 | -20 | 1 | 12 | | | | |
| 013 | +F | T 5 | 0 | 1 | 13 | | | | |
| 014 | +F | T 0 | B 0 | 1 | 14 | | | | |
| 015 | +F | T 0 | -20 | 1 | 15 | | | | |
| 016 | | T 0 | 0 | 1 | 16 | | | | |
| 017 | | T 30 | 0 | 1 | 17 | | | | |
| 018 | | T -10 | 0 | 1 | 18 | | | | |
| 019 | +F | T 0 | 0 ON | 1 | 19 | | | | |
| 020 | +F | T 0 | -20 | 1 | 20 | | | | |

TEST: ARC 3.5-175 (IA15)

TABLE II. (CONTINUED)
DATA SET/RUN NUMBER COLLATION SUMMARY

DATE : 15 OCTOBER 73

TABLE III. - MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY - R19

GENERAL DESCRIPTION: Fuselage, Configuration 3, per Rockwell Lines
VL70-000139B.

NOTE: Identical to B₁₇ except forebody.

Model Scale = 0.010

DRAWING NUMBER: VL70-000139B

| <u>DIMENSIONS:</u> | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> |
|------------------------|-------------------|--------------------|
| Length - IN. | <u>1290.3</u> | <u>12.903</u> |
| Max. Width - IN. | <u>267.6</u> | <u>2.676</u> |
| Max. Depth - IN. | <u>244.5</u> | <u>2.445</u> |
| Fineness Ratio | <u>4.82175</u> | <u>4.82175</u> |
| Area - FT ² | | |
| Max. Cross-Sectional | <u>386.67</u> | <u>0.03867</u> |
| Planform | | |
| Wetted | | |
| Base | | |

TABLE III. - Continued.

MODEL COMPONENT: Canopy - C7

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

Model Scale = 0.010

DRAWING NUMBER VL70-000139

| <u>DIMENSION:</u> | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|---|-------------------|--------------------|
| Length ($X_0 = 433$ to $X_0 = 670$) - in FS | <u>237</u> | <u>2.370</u> |
| Max Width | | |
| Max Depth ($Z_0 =$ to $Z_0 = 501$) - in FS | | |
| Fineness Ratio | | |
| Area | | |
| Max Cross-Sectional | | |
| Planform | | |
| Wetted | | |
| Base | | |

TABLE III. - Continued.

MODEL COMPONENT: ELEVON - E23GENERAL DESCRIPTION: Configuration 3 per WL07 Rockwell Lines
VL70-000139B, data for (1) of (2) sidesModel Scale = 0.010DRAWING NUMBER: VL70-000139B

| DIMENSIONS: | FULL-SCALE | MODEL SCALE |
|---|----------------|----------------|
| Area - FT ² | <u>205.52</u> | <u>0.02055</u> |
| Span (equivalent) - IN. | <u>353.34</u> | <u>3.533</u> |
| Inb'd equivalent chord | <u>114.78</u> | <u>1.147</u> |
| Outb'd equivalent chord | <u>55.00</u> | <u>0.550</u> |
| Ratio movable surface chord/ total surface chord | | |
| At Inb'd equiv. chord | <u>.208</u> | <u>.208</u> |
| At Outb'd equiv. chord | <u>.400</u> | <u>.400</u> |
| Sweep Back Angles, degrees | | |
| Leading Edge | <u>0.00</u> | <u>0.00</u> |
| Tailing Edge | <u>-10.24</u> | <u>-10.24</u> |
| Hingeline | <u>0.00</u> | <u>0.00</u> |
| Area Moment (Normal to hinge line)- FT ³ | <u>1548.07</u> | <u>.001548</u> |
| Product of Area Moment | | |

TABLE III. - Continued.

MODEL COMPONENT: F5 Body Flap

GENERAL DESCRIPTION: 3 Configuration per Rockwell Lines VL70-000139

Scale Model = 0.010

DRAWING NUMBER VL70-000139

| <u>DIMENSION:</u> | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|------------------------|-------------------|--------------------|
| Length - in | <u>84.70</u> | <u>0.8470</u> |
| Max Width - in | <u>267.6</u> | <u>2.676</u> |
| Max Depth | | |
| Fineness Ratio | | |
| Area - Ft ² | | |
| Max Cross-Sectional | | |
| Planform | <u>142.5</u> | <u>0.01425</u> |
| Wetted | | |
| Base | <u>38.0958</u> | <u>.00380958</u> |

TABLE III. - Continued.

MODEL COMPONENT: OMS Pod - M₄

GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VL70-000139

NOTE: M₄ identical to M3, except intersection to fuselage.

Model Scale = 0.010,

DRAWING NUMBER VL70-000139

| <u>DIMENSION:</u> | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|------------------------|-------------------|--------------------|
| Length - IN | 346.0 | 3.460 |
| Max Width - IN | 108.0 | 1.080 |
| Max Depth - IN | 113.0 | 1.130 |
| Fineness Ratio | | |
| Area - FT ² | | |
| Max Cross-Sectional | | |
| Planform | | |
| Wetted | | |
| Base | | |

TABLE III. - Continued.

MODEL COMPONENT: NOZZLES - N8

GENERAL DESCRIPTION: Basic CMS Nozzle of Configuration 2A per Rockwell Lines

VL70-008306 and VL70-000089"B". Intersection of nozzle exit plane and
nozzle centerline at $X_0 = 1570.75$, $Y_0 = \pm 99.25$, $Z_0 = 507.25$

MODEL SCALE = 0.010

DRAWING NO. VL70-008306, VL70-000089"B", SS-A00092

| DIMENSIONS | FULL SCALE | MODEL SCALE |
|---------------------------------|------------|-------------|
| Mach No. _____ | | |
| Length ~ in. | | |
| Gimbal Point to Exit Plane | _____ | _____ |
| Throat to Exit Plane | _____ | _____ |
| Diameter~in. | | |
| Exit | 50.00 | 0.500 |
| Throat | N/A | N/A |
| Inlet | 28.00 | 0.280 |
| Area ~ ft ² ./Nozzle | | |
| Exit | 13.635 | 0.00136 |
| Throat | _____ | _____ |
| Gimbal Point (station)~in. | | |
| X | 1518.0 | 15 .180 |
| Y | +88.0 | 0.880 |
| Z | 492.0 | 4.920 |
| Null Position~deg. | | |
| Pitch | 15°49' | 15°49' |
| Yaw (Outb'd) | +12°17' | +12°17' |

MODEL COMPONENT: MPS NOZZLES - N 24 TABLE III. - Continued.

GENERAL DESCRIPTION: Configuration 3A MPS Nozzles

MODEL SCALE = 0.010

DRAWING NO. VL70-000140A, VL70-005030A

| <u>DIMENSIONS</u> | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|---------------------------------|-------------------|--------------------|
| Mach No. _____ | _____ | _____ |
| Length ~in. | _____ | _____ |
| Gimbal Point to Exit Plane | _____ | _____ |
| Throat to Exit Plane | _____ | _____ |
| Diameter ~in. | _____ | _____ |
| Exit | <u>91.000</u> | <u>0.910</u> |
| Throat | _____ | _____ |
| Inlet | _____ | _____ |
| Area ~ft ² . /Nozzle | _____ | _____ |
| Exit | <u>45.16585</u> | <u>0.00452</u> |
| Throat | _____ | _____ |
| Gimbal Point (station) ~in. | _____ | _____ |
| Upper Nozzle | _____ | _____ |
| X | <u>1445</u> | <u>14.450</u> |
| Y | <u>0</u> | <u>0</u> |
| Z | <u>443</u> | <u>4.430</u> |
| Lower Nozzles | _____ | _____ |
| X | <u>1468.16996</u> | <u>14.68170</u> |
| Y | <u>+53.00000</u> | <u>+5.530</u> |
| Z | <u>342.63938</u> | <u>3.42640</u> |
| Null Position ~deg. | _____ | _____ |
| Upper Nozzle | _____ | _____ |
| Pitch | <u>16°</u> | <u>16°</u> |
| Yaw | <u>0°</u> | <u>0°</u> |
| Lower Nozzles | _____ | _____ |
| Pitch | <u>10°</u> | <u>10°</u> |
| Yaw (outboard) | <u>-5.5°</u> | <u>-5.5°</u> |

TABLE III. - Continued.

MODEL COMPONENT: Solid Plume - PLI

GENERAL DESCRIPTION: SSME simulated plumes from N24 nozzles to represent
all 3 engines at $M = 5.5$ during exit trajectory

MODEL SCALE = 0.010

DRAWING NUMBER:

COORDINATES:

Ratio of local plume radius
to nozzle exit plane internal
radius

Ratio of local axial distance
from nozzle exit plane to nozzle
exit plane internal radius

| | |
|-------|--------|
| 1.053 | 0.057 |
| 1.943 | 1.122 |
| 2.772 | 2.250 |
| 3.497 | 3.341 |
| 4.450 | 4.912 |
| 5.421 | 6.642 |
| 5.905 | 7.566 |
| 6.389 | 8.529 |
| 7.321 | 10.496 |
| 7.861 | 11.699 |
| 8.136 | 12.330 |
| 8.672 | 13.602 |
| 8.937 | 14.367 |
| 9.204 | 14.912 |
| 9.464 | 15.569 |
| | |
| | |
| | |
| | |
| | |
| | |
| | |
| | |

DIMENSIONS:

Nozzle Exit Radius, in.

FULL SCALEMODEL SCALE

45.2

0.452

TABLE III. - Continued.

MODEL COMPONENT: RUDDER - R5

GENERAL DESCRIPTION: 2A, 3 and 3A Configuration per Rockwell Lines

VL70-000095

Model Scale = 0.010

DRAWING NUMBER: VL70-000095

| DIMENSIONS: | FULL-SCALE | MODEL SCALE |
|---|------------|-------------|
| Area - FT ² | 106.38 | 0.0106 8 |
| Span (equivalent) - IN. | 201.0 | 2.010 |
| Inb'd equivalent chord | 91.585 | 0.916 |
| Outb'd equivalent chord | 50.833 | 0.508 |
| Ratio movable surface chord/ total surface chord | | |
| At Inb'd equiv. chord | 0.400 | 0.400 |
| At Outb'd equiv. chord | 0.400 | 0.400 |
| Sweep Back Angles, degrees | | |
| Leading Edge | 34.83 | 34.83 |
| Tailing Edge | 26.25 | 26.25 |
| Hingeline | 34.83 | 34.83 |
| Area Moment (Normal to hinge line)- FT ³ | 526.13 | 000526 |
| Product of Area and Mean Chord | | |

TABLE III. - Continued.

MODEL COMPONENT: BOOSTFR SOLID ROCKET MOTOR - S12GENERAL DESCRIPTION: Configuration 3A, Data for (1) or (2) sides,
per Rockwell Lines VL77-000036AModel Scale = 0.010DRAWING NUMBER VL72-000086A
VL77-000036A

| <u>DIMENSION:</u> | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|---|-------------------|--------------------|
| Length (Includes Nozzle) - IN. | <u>1741.0</u> | <u>17.410</u> |
| Max Width (Tank Dia) - IN. | <u>142.3</u> | <u>1.423</u> |
| Max Depth (Aft Shroud) - IN. | <u>192.0</u> | <u>1.920</u> |
| Fineness Ratio | <u>9.06771</u> | <u>9.06771</u> |
| Area - FT ² | | |
| Max Cross-Sectional | <u>201.06193</u> | <u>0.0201</u> |
| Planform | | |
| Wetted | | |
| Base | | |
| WP of BSRM Centerline (Z _T) - IN. | <u>400</u> | <u>4.00</u> |
| FS of BSRM Nose (X _T) - IN. | <u>200</u> | <u>2.00</u> |

TABLE III. - Continued.

MODEL COMPONENT: EXTERNAL TANK - T10

GENERAL DESCRIPTION: External Oxygen Hydrogen Tank, 3 Configuration,
per Rockwell Lines VL78-000041 and VL72-000088

Model Scale = 0.010

VL72-000088

DRAWING NUMBER

VL78-000041

DIMENSION:

FULL SCALE

MODEL SCALE

Length - IN (Nose @ X_T = 309)

1865

18.65

Max Width (Dia) - IN.

324

3.24

Max Depth

-

-

Fineness Ratio

5.75617

5.75617

Area - FT²

Max Cross-Sectional

572.555

0.0573

Planform

Wetted

Base

WP of Tank Centerline (X_T) IN.

400.0

4.00

TABLE III. - Continued.

MODEL COMPONENT: VERTICAL - V7

GENERAL DESCRIPTION: Centerline vertical tail, double wedge airfoil with rounded leading edge.

NOTE: Same as V5, but with manipulator housing removed.

Model Scale = 0.010

DRAWING NUMBER: VL70-000139

| DIMENSIONS: | FULL-SCALE | MODEL SCALE |
|-----------------------------|------------|-------------|
| <u>TOTAL DATA</u> | | |
| Area (Theo) Ft ² | 425.92 | 0.0425 |
| Planform | | |
| Span (Theo) In | 315.72 | 3.157 |
| Aspect Ratio | 1.675 | 1.675 |
| Rate of Taper | 0.507 | 0.507 |
| Taper Ratio | 0.404 | 0.404 |
| Sweep Back Angles, degrees | | |
| Leading Edge | 45.000 | 45.000 |
| Trailing Edge | 26.249 | 26.249 |
| 0.25 Element Line | 41.130 | 41.130 |
| Chords: | | |
| Root (Theo) WP | 268.50 | 2.685 |
| Tip (Theo) WP | 103.47 | 1.085 |
| MAC | 199.31 | 1.998 |
| Fus. Sta. of .25 MAC | 1463.50 | 14.635 |
| W. P. of .25 MAC | 635.522 | 6.355 |
| B. L. of .25 MAC | 0.00 | 0.00 |
| Airfoil Section | | |
| Leading Wedge Angle Deg | 10.000 | 10.000 |
| Trailing Wedge Angle Deg | 14.920 | 14.920 |
| Leading Edge Radius | 2.0 | 0.020 |
| Void Area - Ft ² | 13.17 | 0.131 |
| Blanketed Area | 0.00 | 0.00 |

TABLE III. - Continued.

MODEL COMPONENT: WING-W 107GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines VI.70-000139BNOTE: Same as W103, except cuff, airfoil and incidence angle.Model Scale = 0.010

| <u>TEST NO.</u> | <u>DWG. NO.</u> | <u>VI.70-000139B</u> |
|--|-------------------|----------------------|
| <u>DIMENSIONS:</u> | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> |
| TOTAL DATA | | |
| Area (Theo.) Ft ² | | |
| Planform | 2690.00 | 0.2690 |
| Span (Theo) In. | 936.68 | 9.3668 |
| Aspect Ratio | 2.265 | 2.265 |
| Rate of Taper | 1.177 | 1.177 |
| Taper Ratio | 0.200 | 0.200 |
| Dihedral Angle, degrees (@ TE of Elevon) | 3.500 | 3.500 |
| Incidence Angle, degrees | 0.500 | 0.500 |
| Aerodynamic Twist, degrees | +3.000 | +3.000 |
| Sweep Back Angles, degrees | | |
| Leading Edge | 45.000 | 45.000 |
| Trailing Edge | -10.24 | -10.24 |
| 0.25 Element Line | 35.209 | 35.209 |
| Chords: | | |
| Root (Theo) B.P.O.O. | 689.24 | 6.892 |
| Tip, (Theo) B.P. | 137.85 | 1.378 |
| MAC | 474.81 | 4.748 |
| Fus. Sta. of .25 MAC | 1136.89 | 11.3689 |
| W.P. of .25 MAC | 299.20 | 2.992 |
| B.L. of .25 MAC | 182.13 | 1.8213 |
| EXPOSED DATA | | |
| Area (Theo) Ft ² | | |
| Span, (Theo) In. BP108 | 1752.29 | 0.1752 |
| Aspect Ratio | 720.68 | 7.2068 |
| Taper Ratio | 2.058 | 2.058 |
| Chords | 0.2451 | 0.2451 |
| Root BP108 | 562.40 | 5.6240 |
| Tip 1.00 b | 137.85 | 1.3785 |
| MAC | 393.03 | 3.9303 |
| Fus. Sta. of .25 MAC | 1185.31 | 11.8531 |
| W.P. of .25 MAC | 300.20 | 3.002 |
| B.L. of .25 MAC | 251.76 | 2.518 |
| Airfoil Section (Rockwell Mod NASA) | | |
| XXXX-64 | | |
| Root $\frac{b}{2}$ = | 0.10 | 0.10 |
| Tip $\frac{b}{2}$ = | 0.12 | 0.12 |
| Data for (1) or (2) Sides | | |
| Leading Edge Cuff | | |
| Planform Area Ft ² | 118.333 | 0.0118 |
| Leading Edge Intersects Fus M. L. @ Sta | 500 | 5.00 |
| Leading Edge Intersects Wing @ Sta | 1083.4 | 10.834 |

TABLE III. - Continued.

MODEL COMPONENT: Attach Structure - AT₆

GENERAL DESCRIPTION: Right Rear, Orbiter to External Tank

MODEL SCALE = 0.010

DRAWING NO. VL72-000088B + VL72-000089 NOTE: Use first drawing for location and second drawing for detail of struts

DIMENSIONS:

First Strut

| | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|------------------------------------|-------------------|--------------------|
| Diameter in. (Approx.) | 1 | 0.010 |
| Fwd Location, in. (Attach to Orb.) | | |
| X _o | | |
| X _s | 1307 | 13.07 |
| | 2058 | 20.58 |

Approximate Aft Location, in. (Attach to Orb.)

| | | |
|----------------|------|-------|
| X _o | 1107 | 11.07 |
| X _s | 1858 | 18.58 |

(Note: This strut is the mirror image of Strut AT₇)

Second Strut

| | | |
|-------------------------|------|-------|
| Diameter, in. (Approx.) | 1 | 0.010 |
| Location, in. | | |
| X _o | 1307 | 13.07 |
| X _s | 2058 | 20.58 |

(Note: This is a Cross-Brace Strut)

TABLE III. - Continued.

MODEL COMPONENT: Attach Structure - AT7

GENERAL DESCRIPTION: Left Rear, Orbiter to External Tank

MODEL SCALE = 0.010

DRAWING NO. VL7-000088B + VL72-000089 NOTE: Use first drawing for location and second drawing for detail of struts.

DIMENSIONS:FULL SCALE MODEL SCALEFORWARD ATTACH POINTS

Orbiter to Tank

| | | |
|------------------------|------|-------|
| Number of Struts | 1 | 1 |
| Diameter in. (Approx.) | 1 | 0.010 |
| Location in. | | |
| X_o | 1307 | 13.07 |
| X_T | 2058 | 20.58 |

Orbiter to SRB

| | | |
|------------------|--|--|
| Number of Struts | | |
| Diameter in. | | |
| Location in. | | |
| X_o | | |
| X_s | | |

Tank to SRB

| | | |
|------------------|--|--|
| Number of Struts | | |
| Diameter in. | | |
| Location in. | | |
| X_p | | |
| X_s | | |

AFT ATTACH POINTS

Orbiter to Tank

| | | |
|------------------------|------|-------|
| Number of Struts | 1 | 1 |
| Diameter in. (Approx.) | 1 | 0.010 |
| Location in. (Approx.) | | |
| X_o | 1107 | 11.07 |
| X_T | 1858 | 18.58 |

Orbiter to SRB

| | | |
|------------------|--|--|
| Number of Struts | | |
| Diameter in. | | |
| Location in. | | |
| X_o | | |
| X_s | | |

Tank to SRB

| | | |
|------------------|--|--|
| Number of Struts | | |
| Diameter in. | | |
| Location in. | | |

TABLE III. - Continued.

MODEL COMPONENT: Attach Structure - AT8GENERAL DESCRIPTION: Front, SRB to External TankMODEL SCALE = 0.010DRAWING NO. VL72-00106DIMENSIONS:FULL SCALEMODEL SCALEFORWARD ATTACH POINTSOrbiter to Tank

Number of Struts

Diameter in.

Location in.

 X_o X_T

Orbiter to SRB

Number of Struts

Diameter in.

Location in.

 X_o X_S

Tank to SRB

Number of Struts (3 to each SRB)

Diameter in. (Approx)

Location in.

 X_T X_S

6

6

5/6

0.06

947

9.47

404

4.04

TABLE III. - Continued.

MODEL COMPONENT: Attach Structure - AT9

GENERAL DESCRIPTION: Rear, SRB to External Tank

MODEL SCALE = 0.010

DRAWING NO. VL72-00

DIMENSIONS:FULL SCALE MODEL SCALEFORWARD ATTACH POINTS

Orbiter to Tank

Number of Struts

Diameter in.

Location in.

 X_o X_T

| | |
|-------|-------|
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |

Orbiter to SRB

Number of Struts

Diameter in.

Location in.

 X_o X_s

| | |
|-------|-------|
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |

Tank to SRB

Number of Struts (3 to each SRB)

Diameter in. (Approx.)

Location in.

 X_T X_s

| | |
|-------|-------|
| 6 | 6 |
| 576 | 0.06 |
| 2058 | 20.58 |
| 1515 | 15.15 |
| _____ | _____ |

AFT ATTACH POINTS

Orbiter to Tank

Number of Struts

Diameter in.

Location in.

 X_o X_T

| | |
|-------|-------|
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |

Orbiter to SRB

Number of Struts

Diameter in.

Location in.

 X_o X_s

| | |
|-------|-------|
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |

Tank to SRB

Number of Struts

Diameter in.

Location in.

 X_T X_s

| | |
|-------|-------|
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |
| _____ | _____ |

TABLE III. - Continued.

MODEL COMPONENT: Attach Structure - AT11

GENERAL DESCRIPTION: Front, Orbiter to External Tank

MODEL SCALE = 0.010

DIMENSIONS:

| | <u>FULL SCALE</u> | <u>MODEL SCALE</u> |
|----------------------|-----------------------------------|--------------------|
| Number of Struts | <u>2</u> | <u>2</u> |
| Width of Each Strut | <u>12$\frac{1}{2}$</u> | <u>.125</u> |
| Length of Each Strut | <u>25</u> | <u>.250</u> |
| Location | | |
| X _O | <u>391.0</u> | <u>3.91</u> |
| X _T | <u>1132.0</u> | <u>11.32</u> |

NOTE: Configuration (AT11) is the same as configuration AT₅ except legs are 12 $\frac{1}{2}$ by 25 instead of 6 inches diameter.

TABLE III. - Concluded.

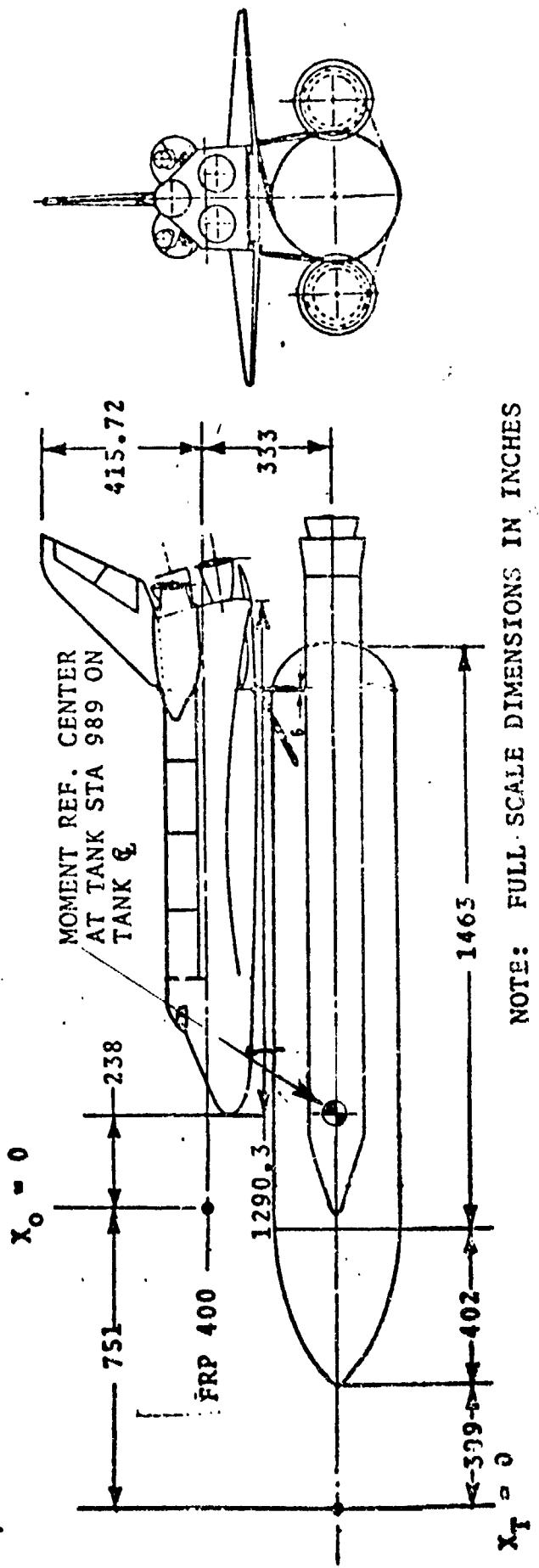
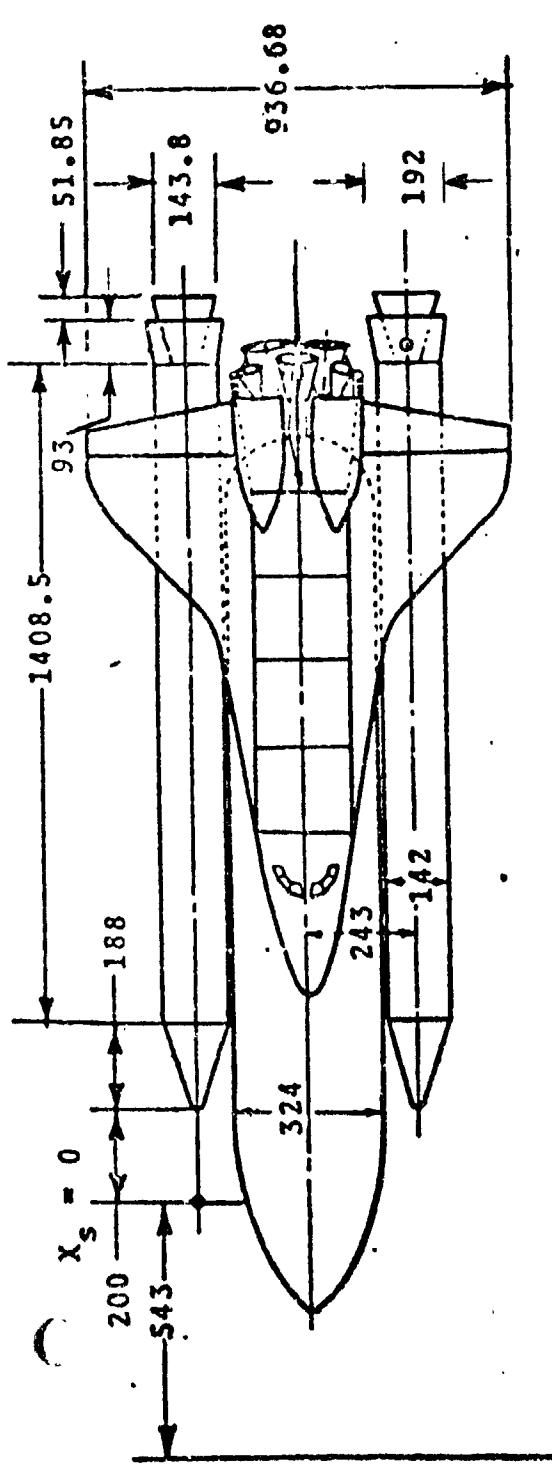
MODEL COMPONENTS: PT₄, PT₅, PT₆, FL₃, FL₄, PS₁, PS₂, PS₃, FR.

GENERAL DESCRIPTON: General dimensional data not applicable. See
description in "Configurations Investigated" section and in figure 2.

Notes:

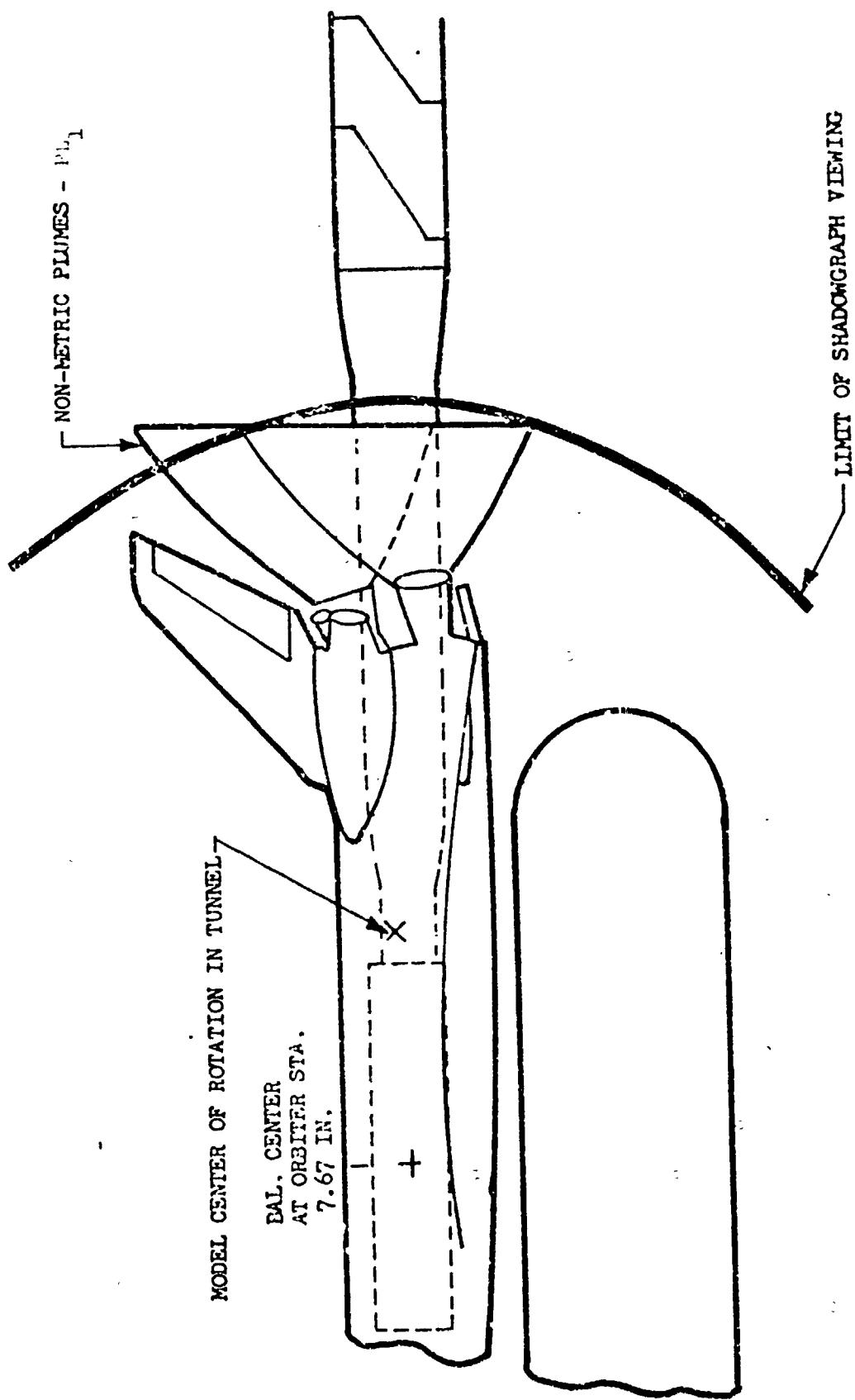
1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrow
 2. For clarity, origins of wind and stability axes have been displaced from the center of gravity
-

Figure 1. - Axis systems.



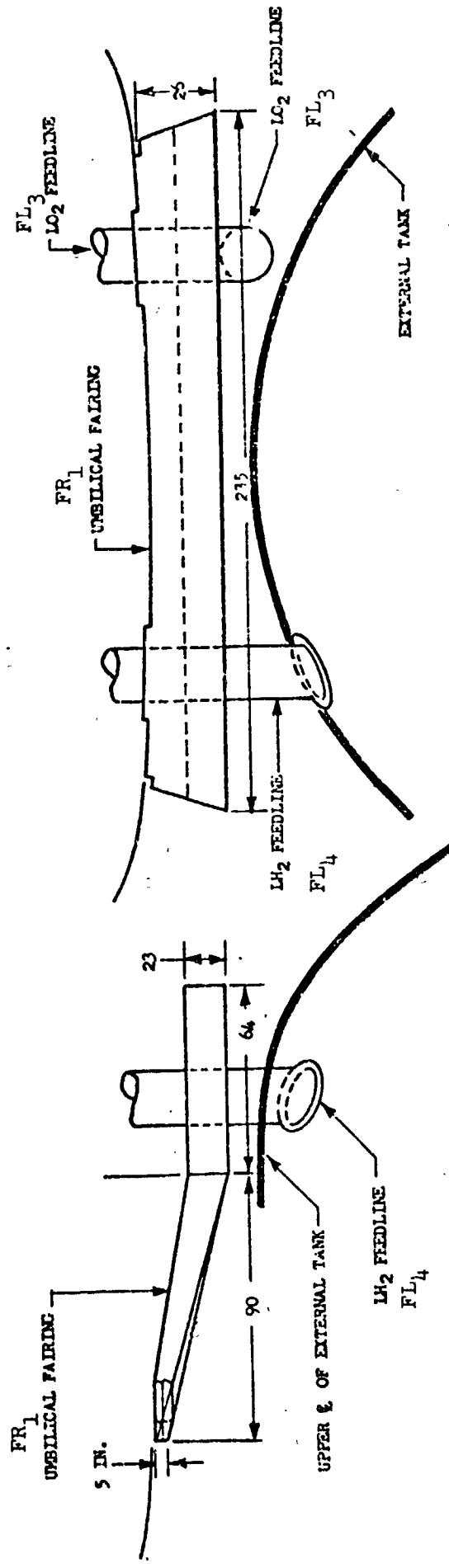
a. Integrated Vehicle Configuration 3 (Mated)

Figure 2. - Model sketches.



b. Model Installation with Non-Metric Plumes Included

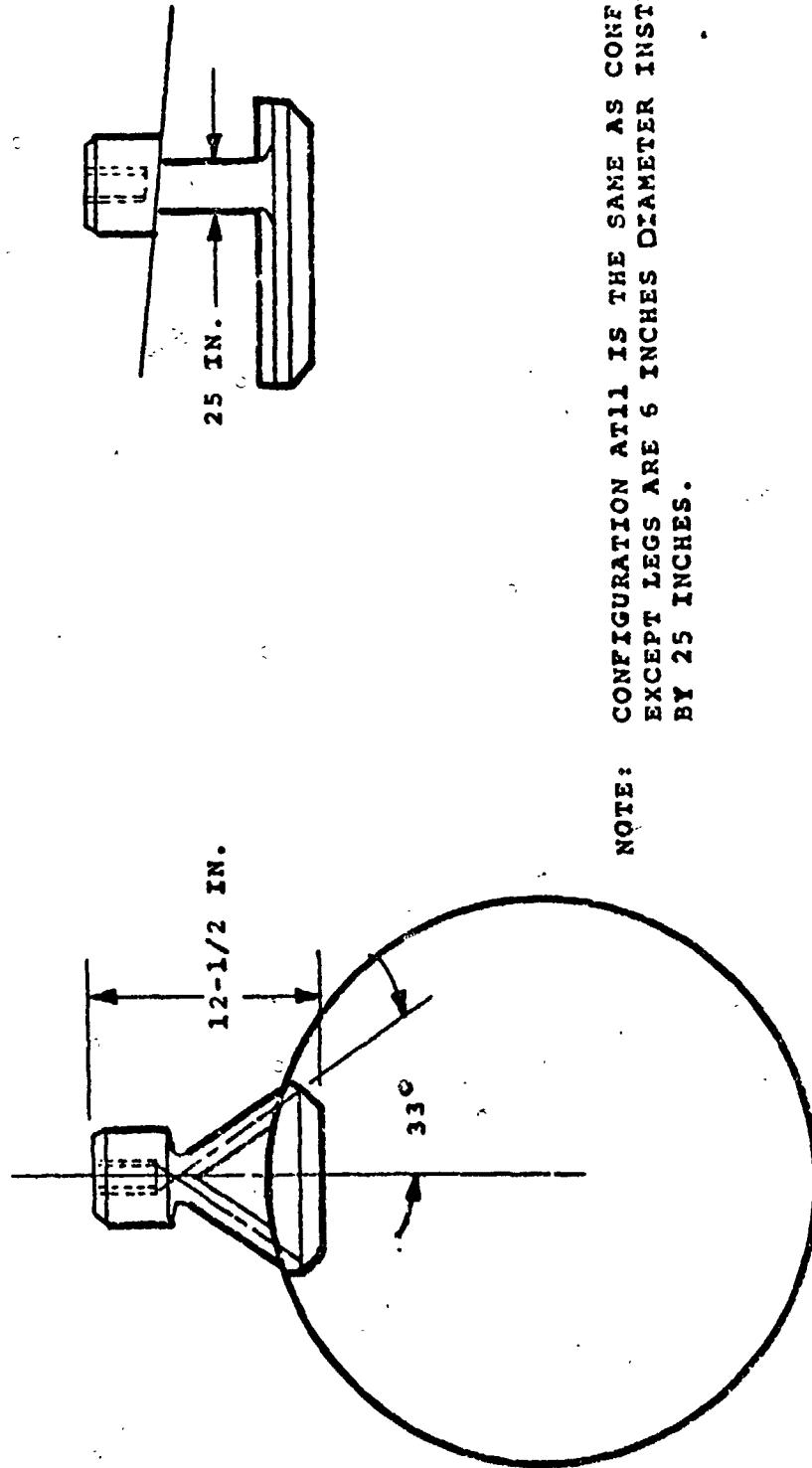
Figure 2. - Continued.



c. Umbilical Fairing on Orbiter (FR₁)

Figure 2. - Continued.

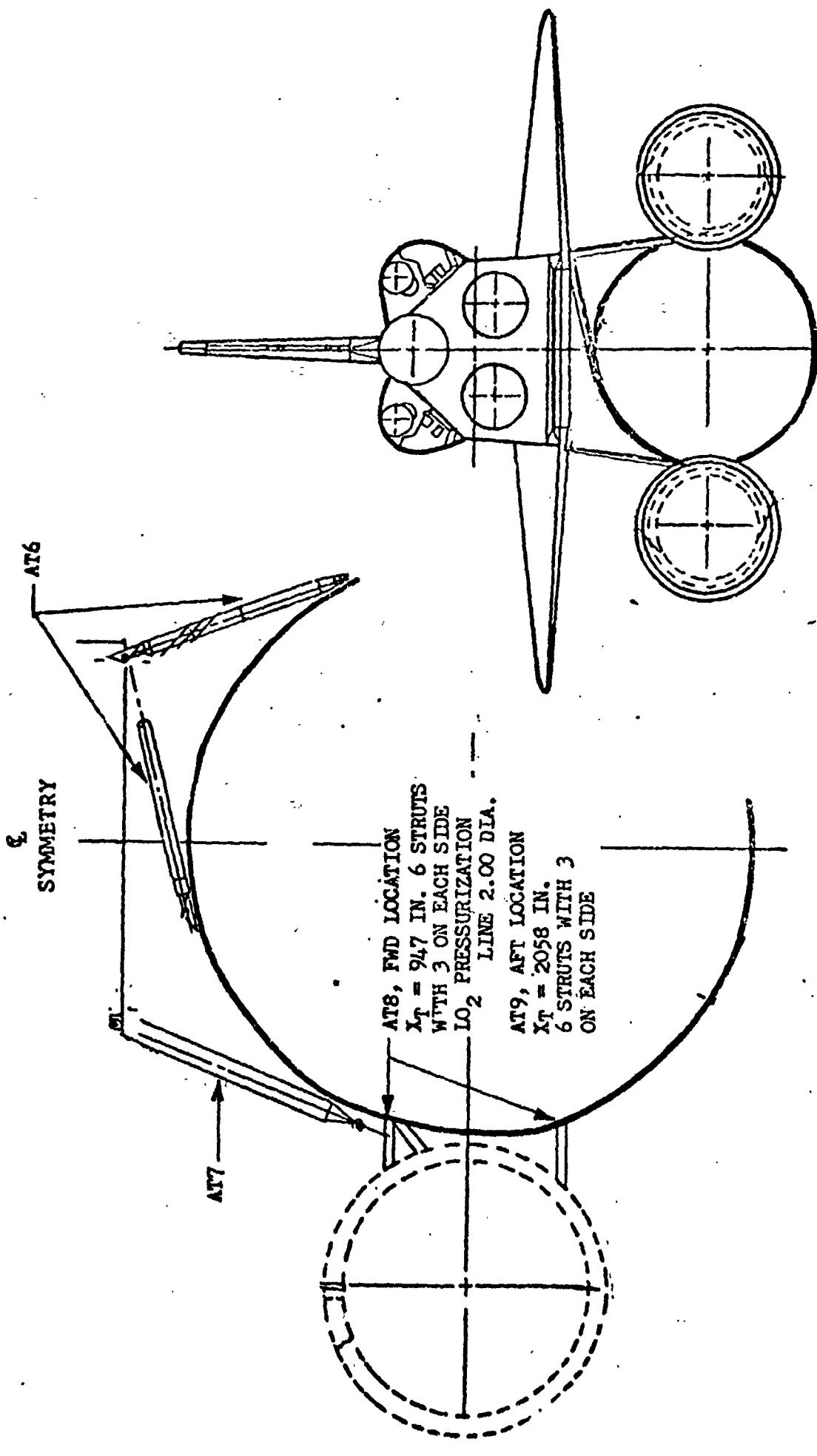
CONFIGURATION AT11



NOTE: CONFIGURATION AT11 IS THE SAME AS CONFIGURATION AT5
EXCEPT LEGS ARE 6 INCHES DIAMETER INSTEAD OF 12-1/2
BY 25 INCHES.

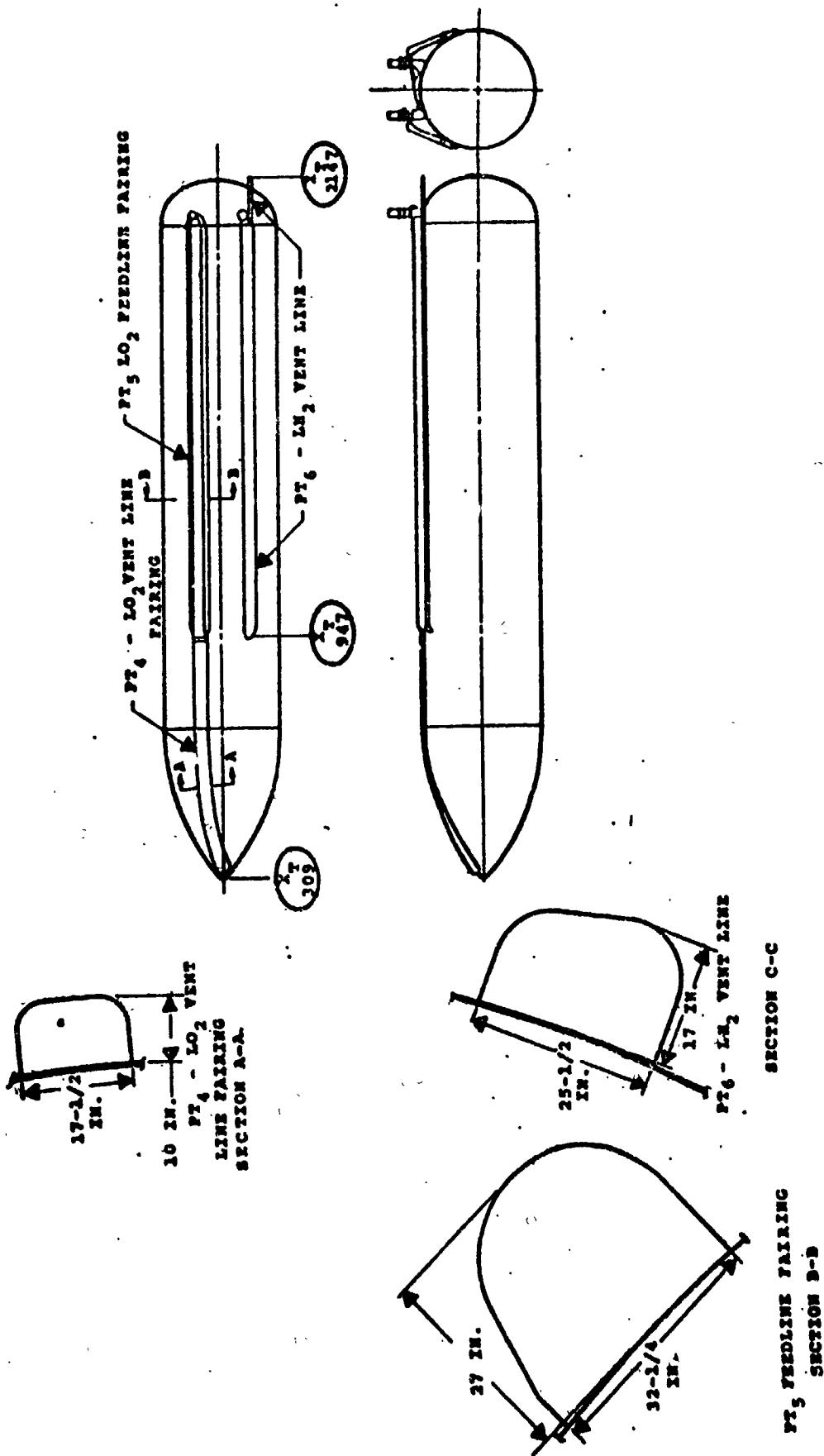
d. FORWARD Attachment of the External Tank to the Orbiter

Figure 2. - Continued.



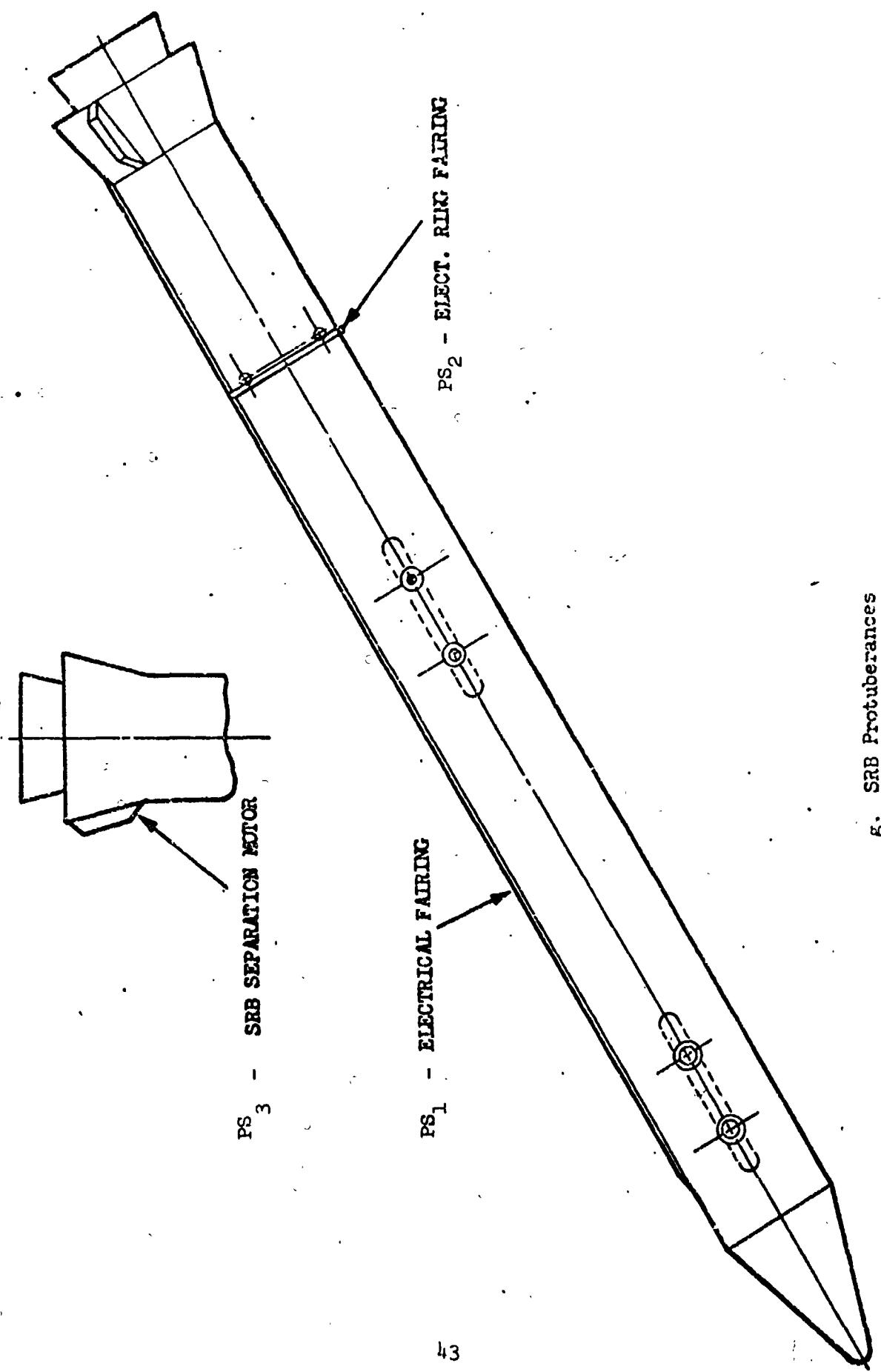
e. Attach Structure - VL72-000089 Configuration 3A

Figure 2. - Continued.



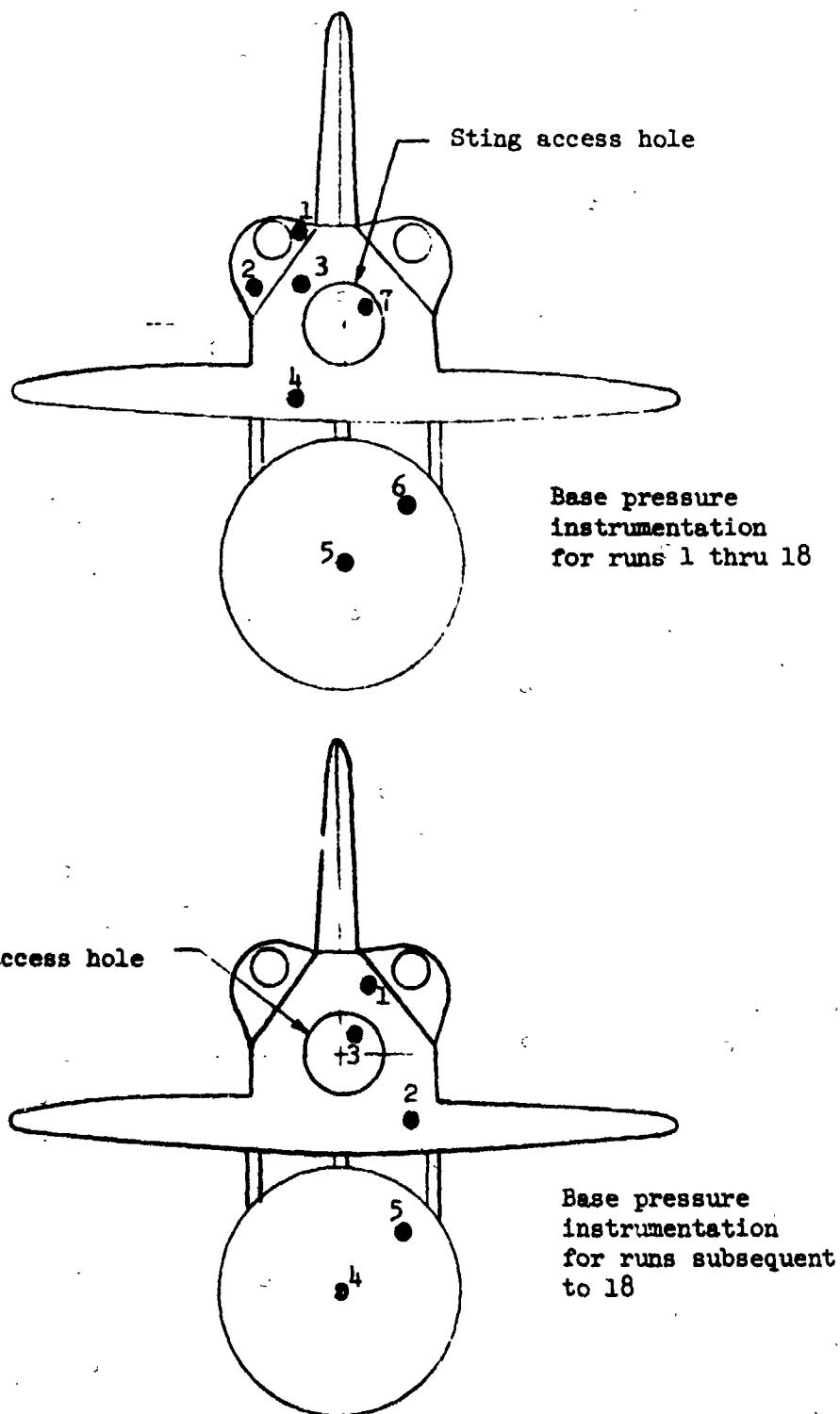
f. External Tank Protuberances

Figure 2. - Continued.



g. SRB Protuberances

Figure 2. - Continued.



h. Base pressure orifice locations.

Figure 2. - Concluded.



(a) Close-up 3/4 rear view of 0.010-scale orbiter model 139B mounted on external tank

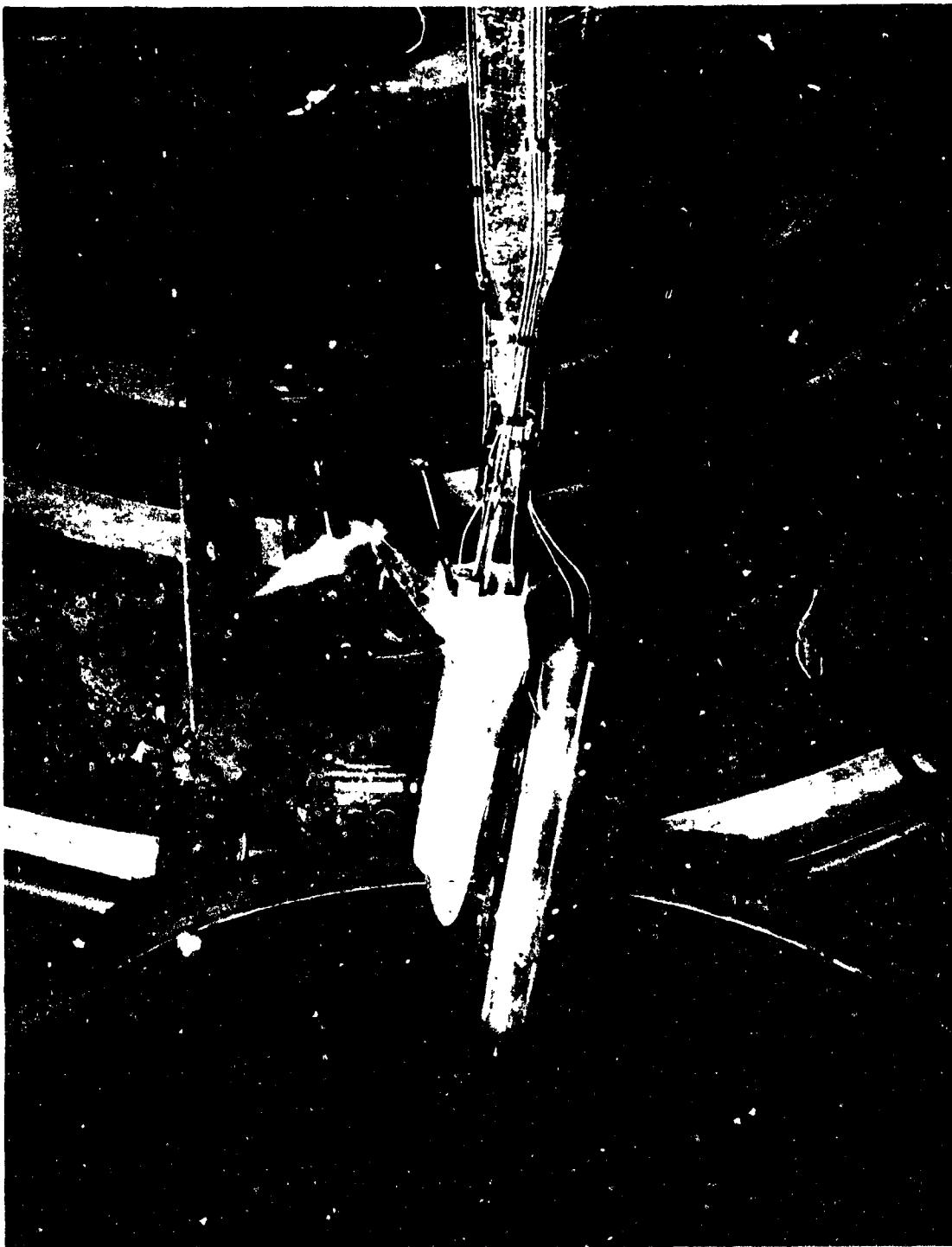
Figure 3. - Model photographs.

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR



(b) Side view of 0.010-scale orbiter model 139B, external tank, sting, and strut

Figure 3. - Continued.



(c) Side view of 0.010-scale orbiter model 139B, external tank, and sting.

Figure 3. - Concluded.

DATA FIGURES

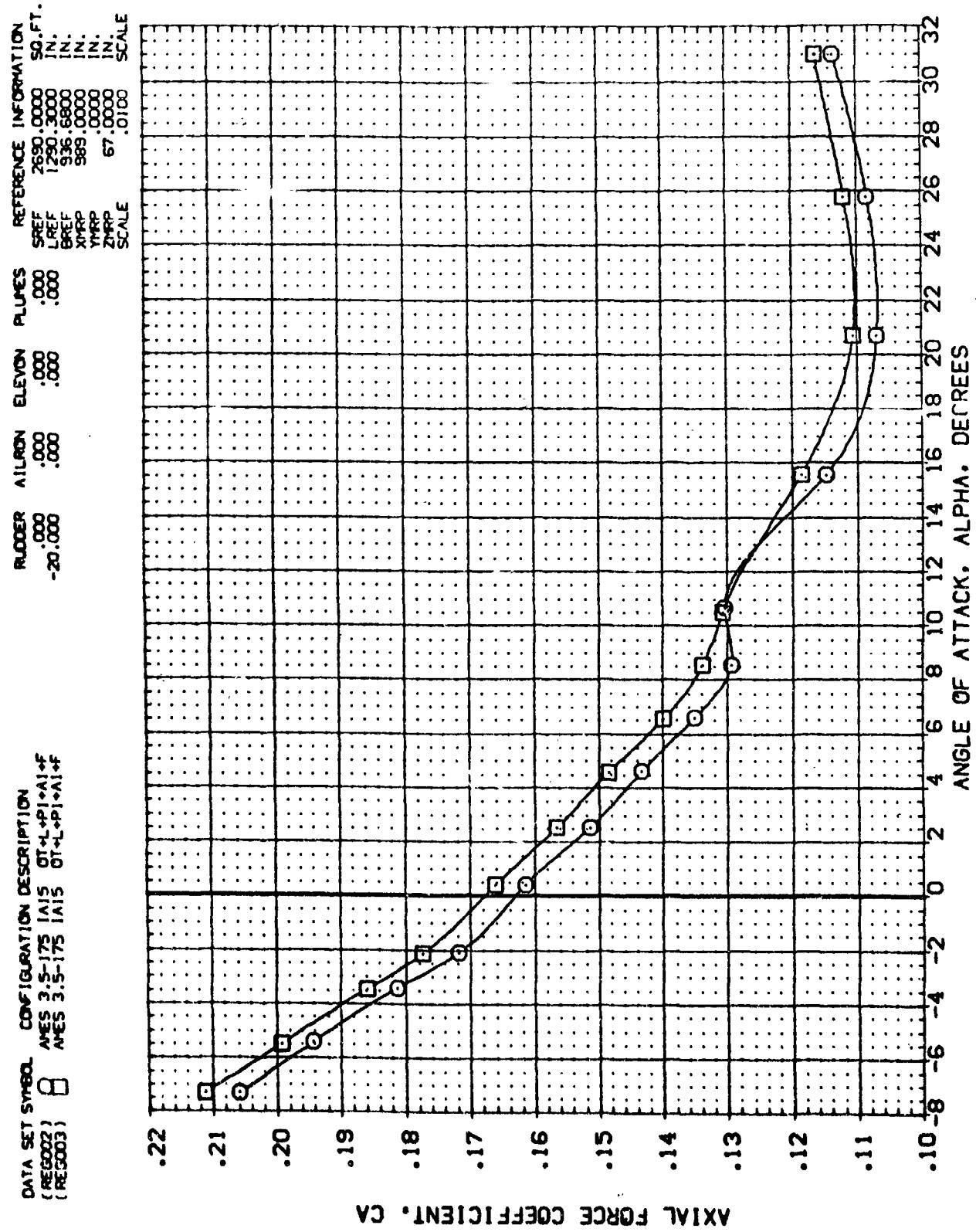


FIG. 4 RUDDER DEFLECTION WITH FAIRING, LONGITUDINAL.
($\text{MACH} = 7.32$)

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REG002) ASES 3.5-175 LAYS DT-LP1-A14F
 (REG003) ASES 3.5-175 LAYS DT-LP1-A14F

| | RUDDER | AIRTON | ELEVON | PONES | REFERENCE INFORMATION |
|--|---------|--------|--------|-------|-----------------------|
| | .000 | .000 | .000 | .000 | SREF 2630.0000 SQ.FT. |
| | -20.000 | .000 | .000 | .000 | LREF 1290.3000 IN. |
| | | | | | BREF 936.6800 IN. |
| | | | | | XRP 989.0000 IN. |
| | | | | | YRP .0000 IN. |
| | | | | | ZRP 67.0000 IN. |
| | | | | | SCALE .0100 |

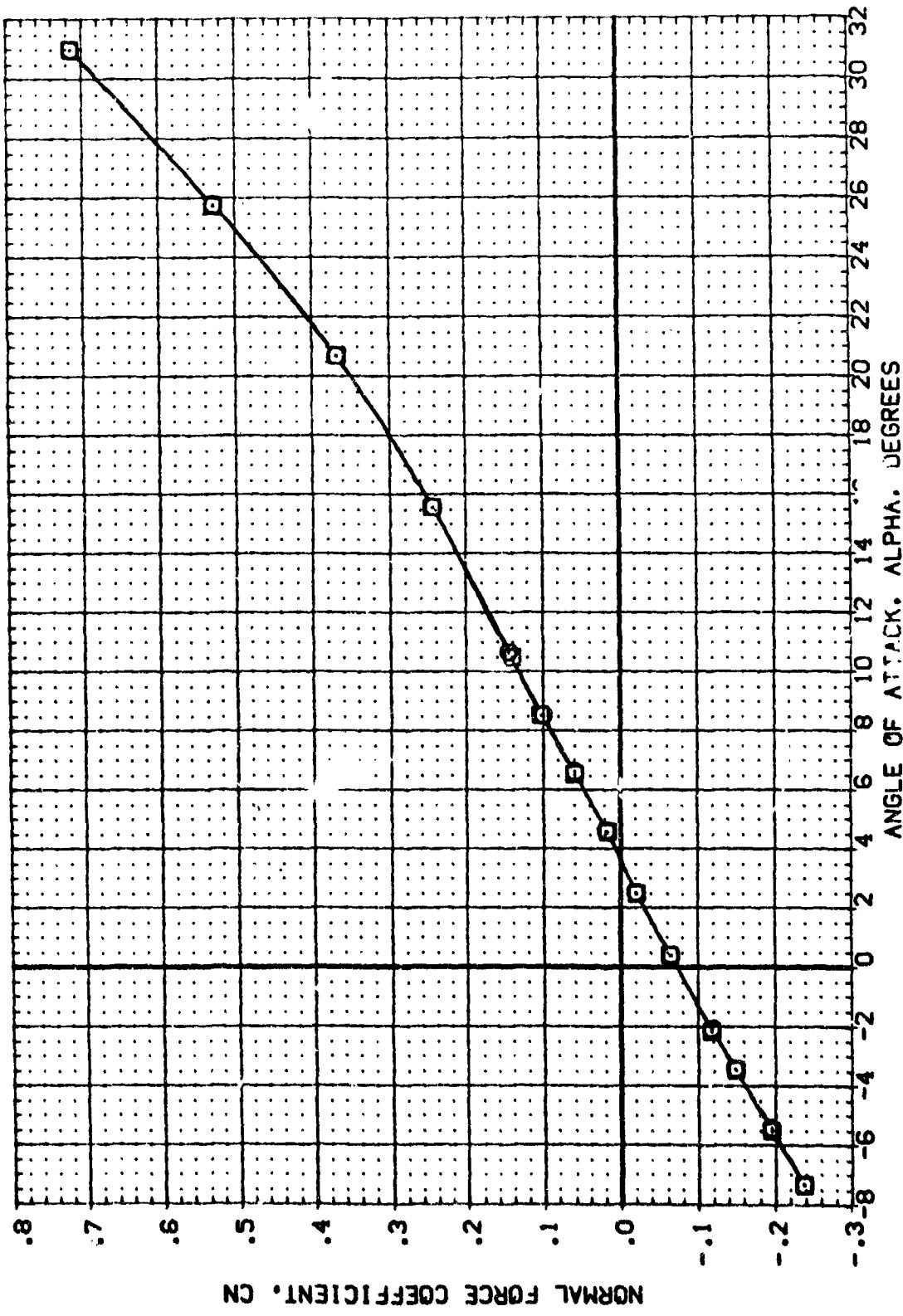
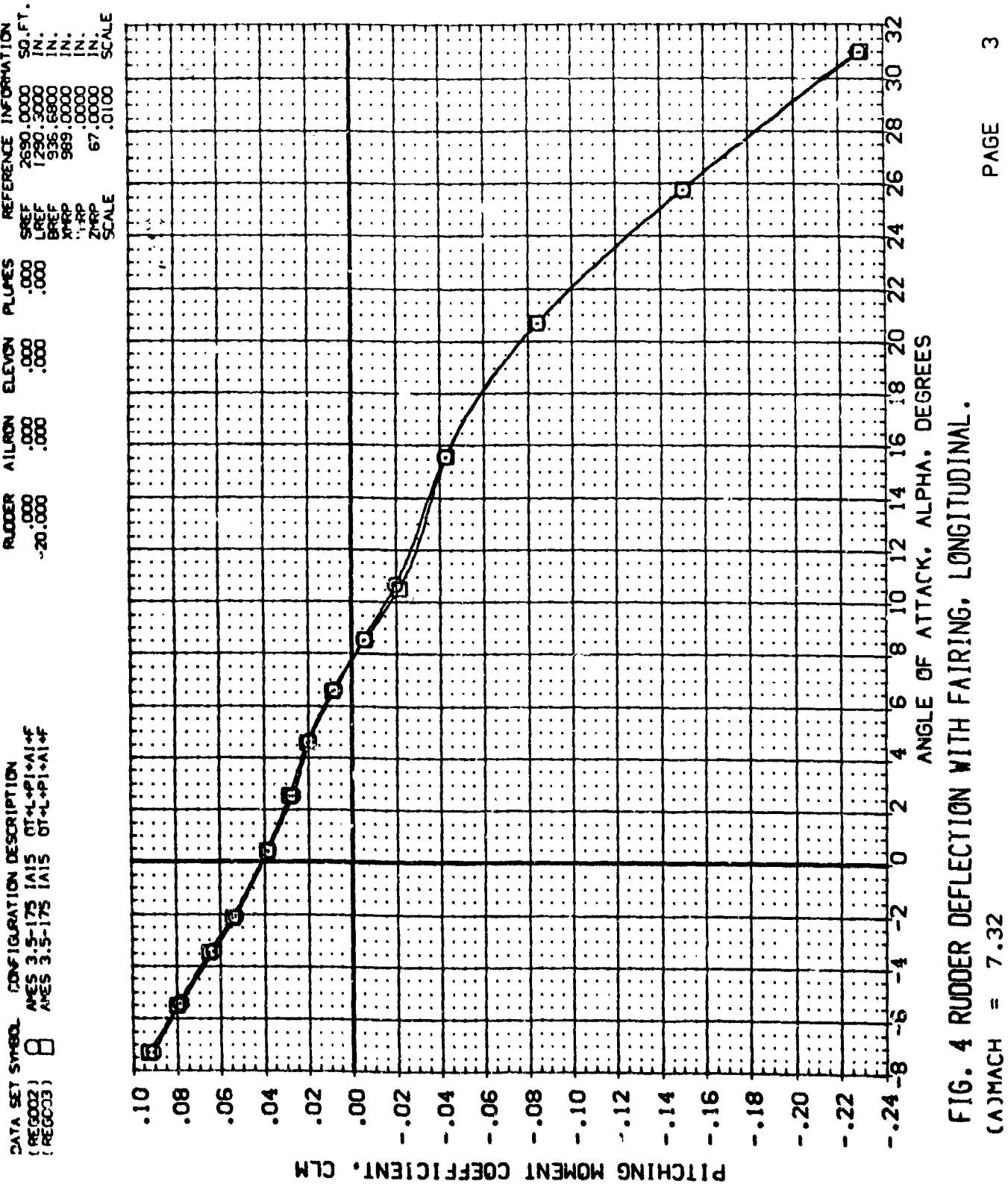


FIG. 4 RUDDER DEFLECTION WITH FAIRING, LONGITUDINAL.
 (A)MACH = 7.32



PAGE 3

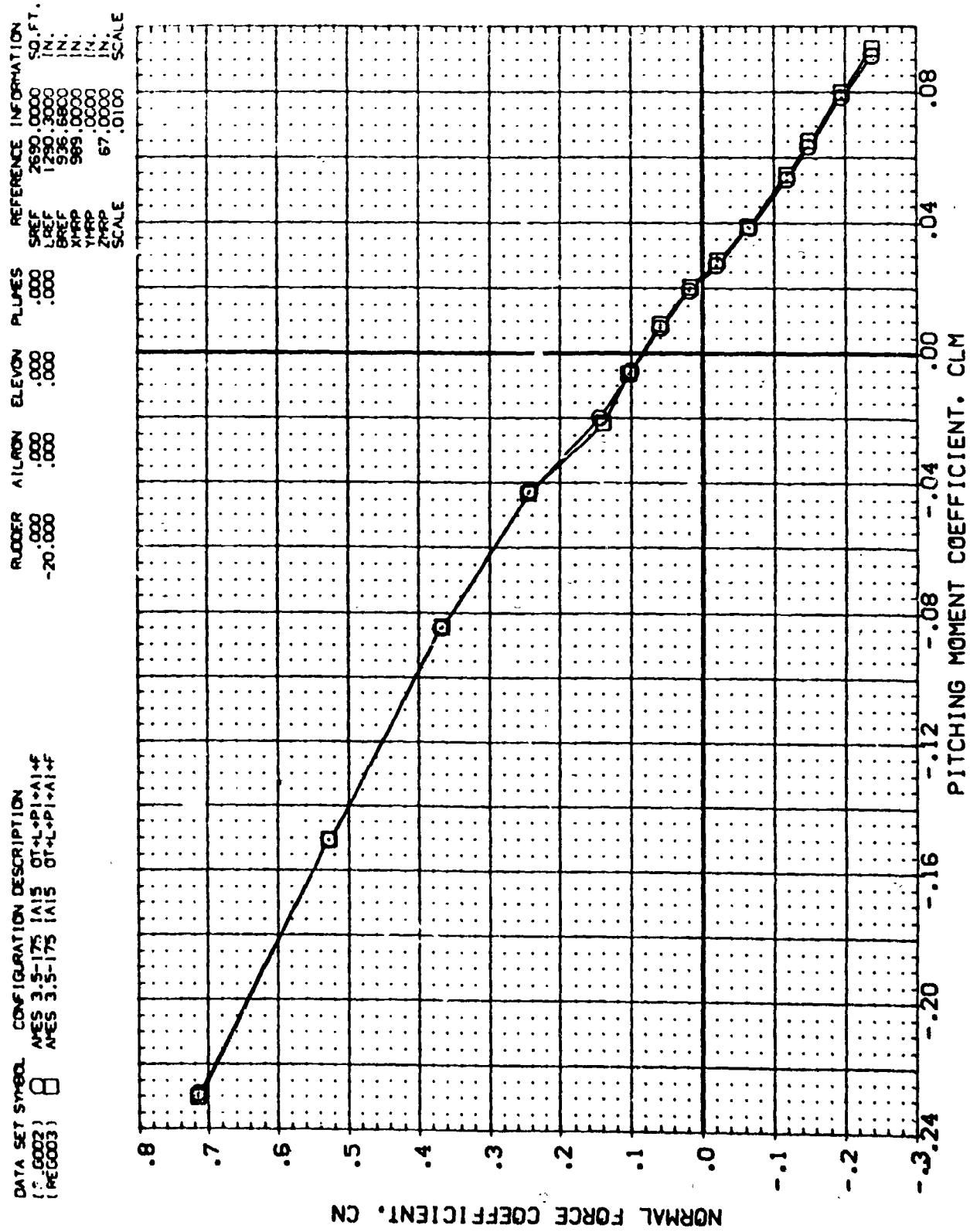


FIG. 4 RUDDER DEFLECTION WITH FAIRING, LONGITUDINAL.
 (A)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REGO14) AMES 3.5-175 LALS OT-L+P1+A145
 (REGO15) AMES 3.5-175 LALS OT-H+P1+A145

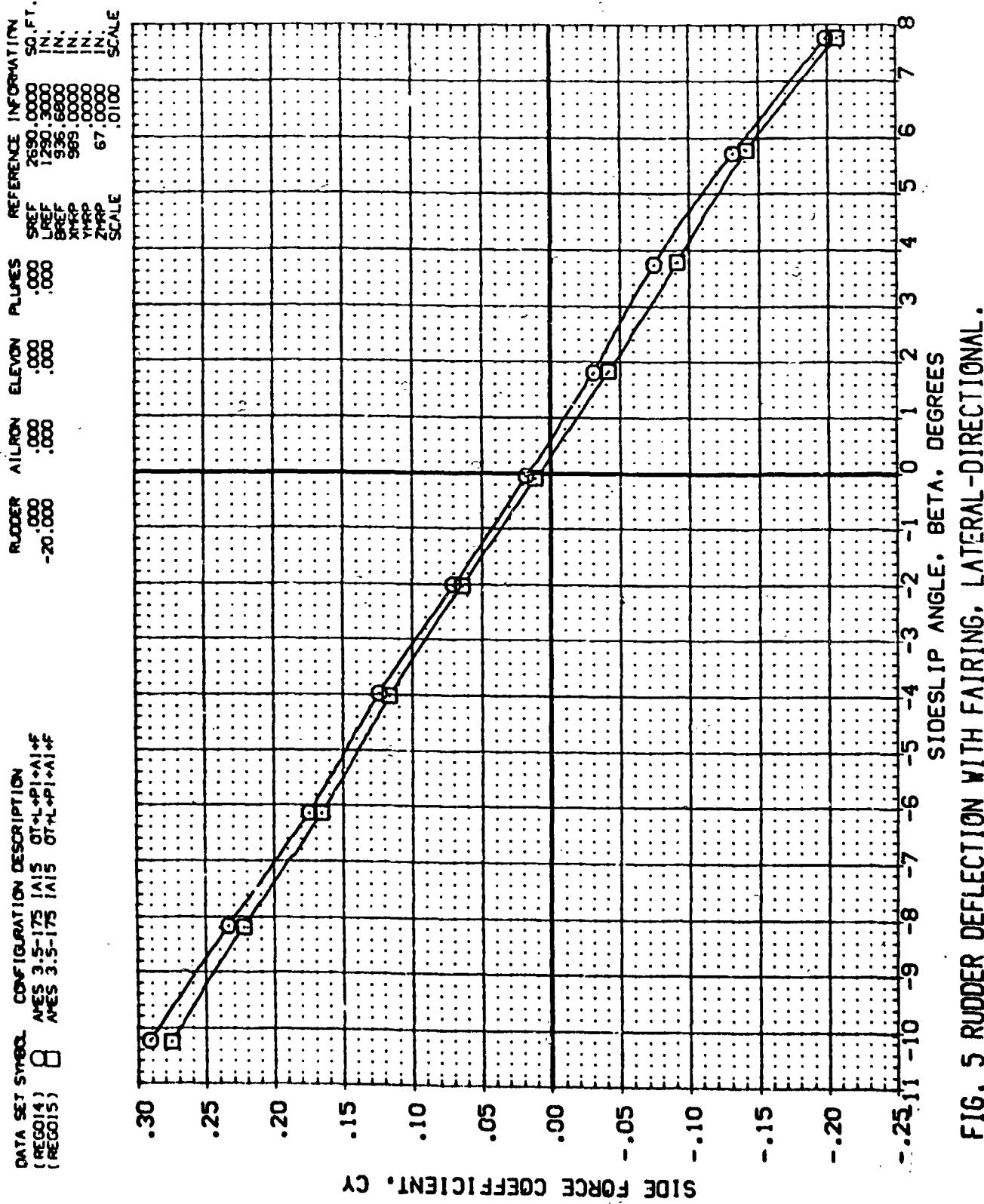


FIG. 5 RUDDER DEFLECTION WITH FAIRING, LATERAL-DIRECTIONAL.
 $(\lambda MACH = 7.32$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
(REG01) ARES 3.5-175 TAIS OTL-PI-ALF
(REG01) ARES 3.5-175 TAIS OTL-PI-ALF

Rudder Alalon ELEVON PLATES REFERENCE INFORMATION
RUDER :000 :000 :000 SREF 2690.0000 SQ.FT.
-20.000 :000 :000 LREF 1290.3000 IN.
BREF 936.6000 IN.
XHMP 989.0000 IN.
YHMP .0000 IN.
ZHMP 67.0000 IN.
SCALE .0100 IN.

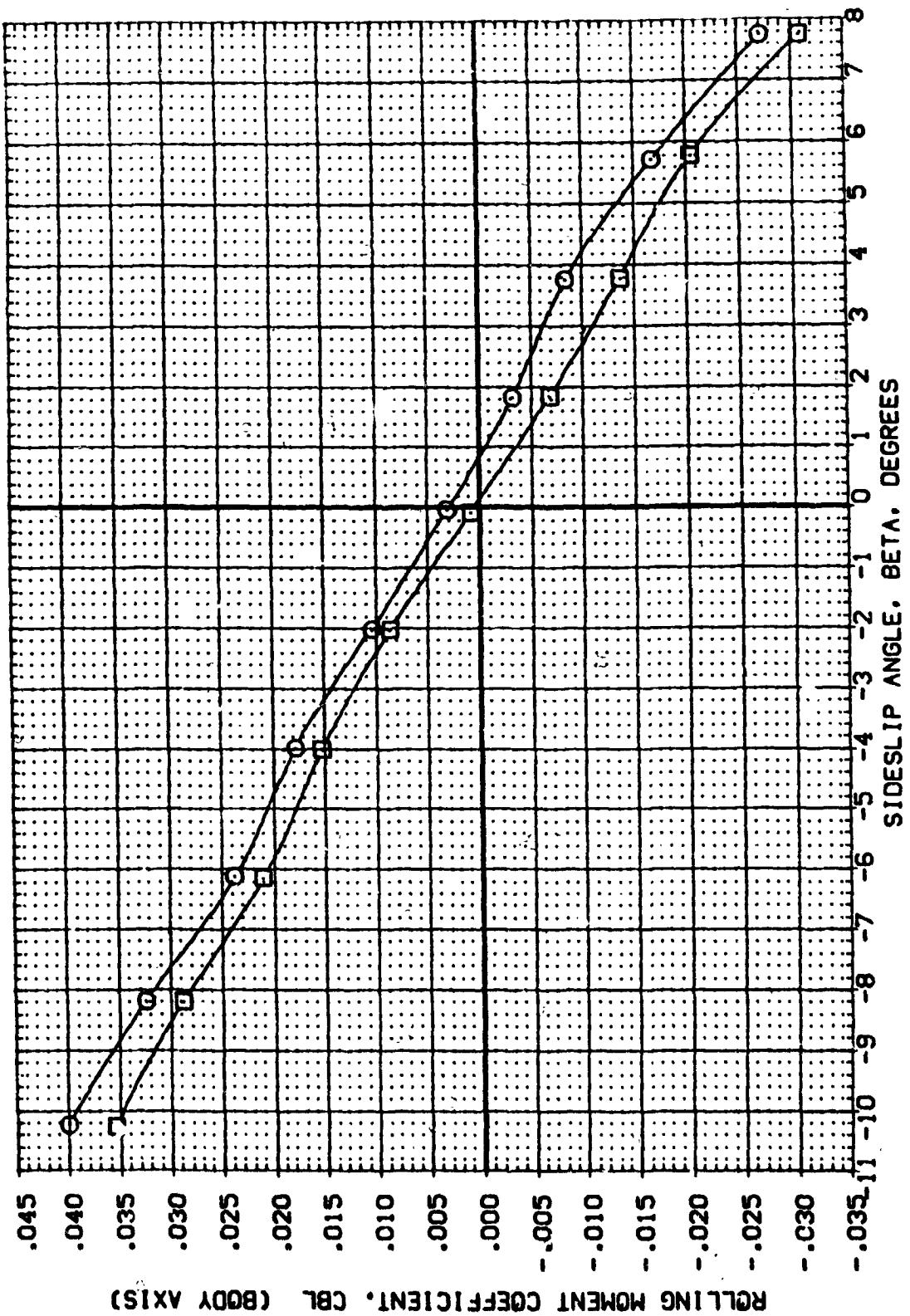


FIG. 5 RUDDER DEFLECTION WITH FAIRING, LATERAL-DIRECTIONAL.

(A)MACH = 7.32

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DATA SET SYMBOL: CONFIGURATION DESCRIPTION
 (REGD14) 8 ANES 3.5-175 TA15 DT-LAP-A14F
 (REGD15) 8 ANES 3.5-175 TA15 DT-LAP-A14F

RUDDER AIRON ELEVON PLATES REFERENCE INFORMATION
 .000 .000 .000 SREF 2690.0000 SQ.FT.
 -20.000 .000 .000 LREF 1290.3000 IN.
 BREF 1936.6800 IN.
 XTRP 989.0000 IN.
 YTRP 67.0000 IN.
 ZTRP .0100 SCALE

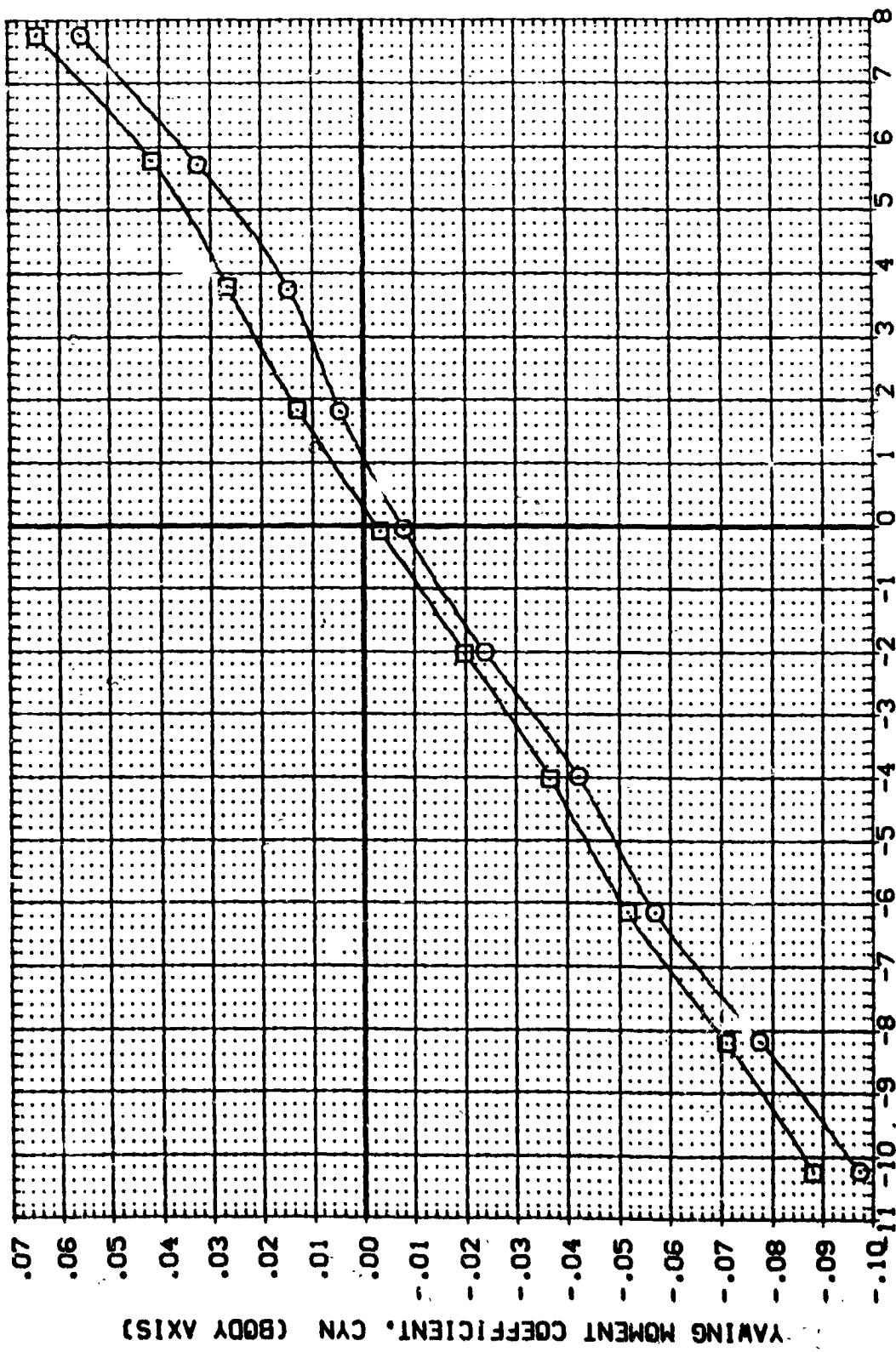


FIG. 5 RUDDER DEFLECTION WITH FAIRING, LATERAL-DIRECTIONAL.
 (A)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION
(REGO14) 8 MACH 3.5-175 LARIS OT-L-PI-AI-OF
(REGO15) 8 MACH 3.5-175 LARIS OT-L-PI-AI-OF

RUDDER AILRDN ELEVON PLUMES REFERENCE INFORMATION
.000 .000 .000 SREF 2690.0000 SQ.FT.
-20.000 .000 .000 LREF 1290.3000 IN.
BREF 975.6800 IN.
XHPP 389.0000 IN.
ZHPP 67.0000 IN.
SCALE .0100

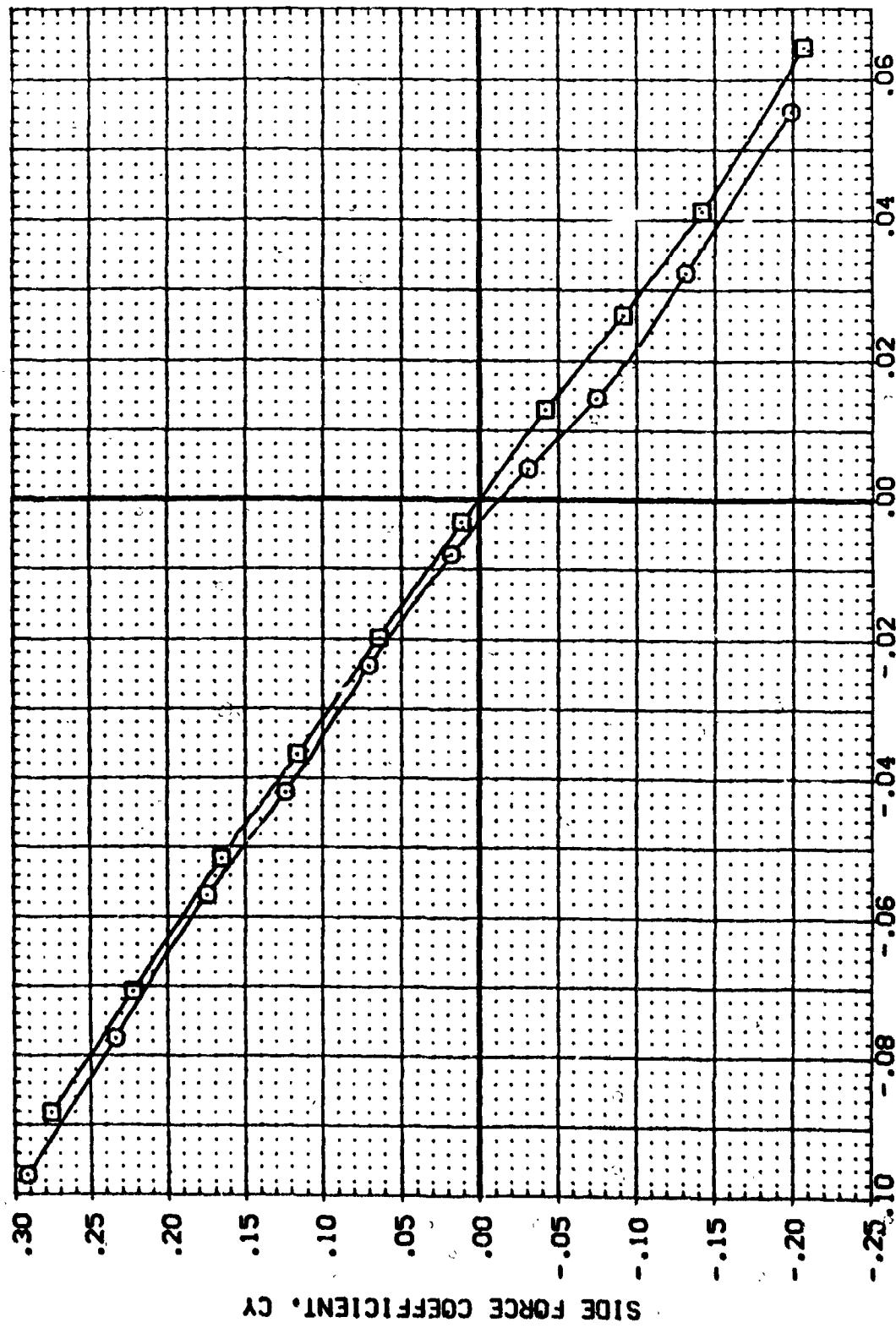


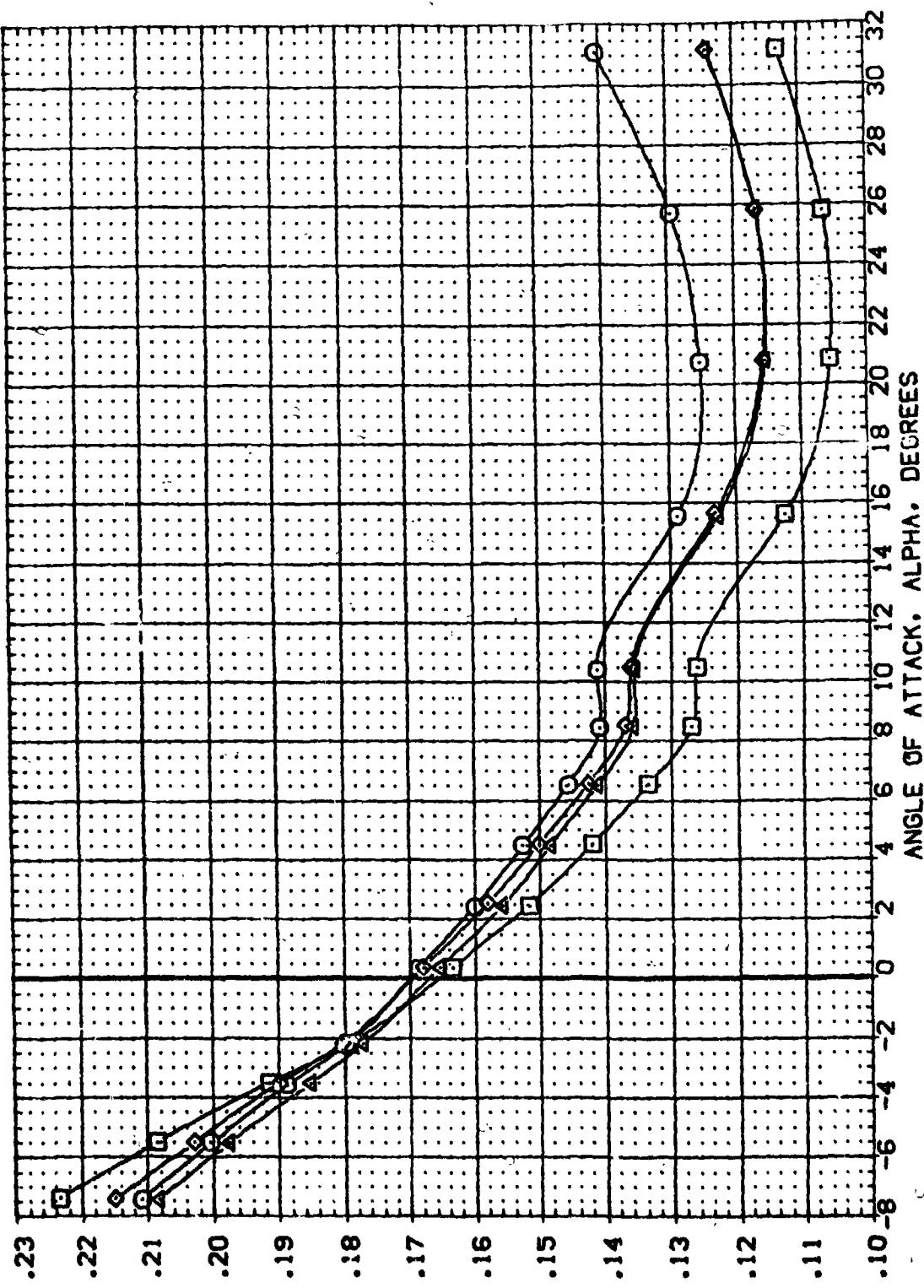
FIG. 5 RUDDER DEFLECTION WITH FAIRING, LATERAL-DIRECTIONAL.
(MACH = 7.32)

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DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | | |
|----------|---|---------------|------|-----------|
| (REG006) | □ | ALLES 3.5-175 | AIIS | DT-L-SP-A |
| (REG007) | ○ | ALLES 3.5-175 | AIIS | DT-4-SP-A |
| (REG008) | × | ALLES 3.5-175 | AIIS | DT-4-SP-B |
| (REG010) | △ | ALLES 3.5-175 | AIIS | DT-L-SP-B |

RUDDER ALARM ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 15,000 .000 SREF 2690.0000 SQ.FT.
 .000 .000 -10,000 .000 LREF 1290.3000 IN.
 .000 .000 -20,000 .000 BREF 936.6800 IN.
 .000 .000 .000 XHPP 369.0000 IN.
 .000 .000 .000 YHPP 67.0000 IN.
 SCALE .0100



AXIAL FORCE COEFFICIENT. C_A

FIG. 6 ELEVON DEFLECTION WITHOUT FAIRING, LONGITUDINAL.

(A)MACH = 7.32

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DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | |
|----------|--------------|-----|----------|
| (REG006) | AMES 3.5-175 | IAS | OT-L-P+A |
| (REG007) | AMES 3.5-175 | IAS | OT-L-P+A |
| (REG008) | AMES 3.5-175 | IAS | OT-L-P+A |
| (REG010) | AMES 3.5-175 | IAS | OT-L-P+A |

RUDDER ALIASN ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 15.000 .000 SREF 2690.0000 50.FT.
 .000 .000 -40.000 .000 LREF 1290.3000 IN.
 .000 .000 -20.000 .000 BREF 935.6800 IN.
 .000 .000 0.000 XREF 569.0000 IN.
 .000 .000 57.0000 IN.
 SCALE .0100 SCALE

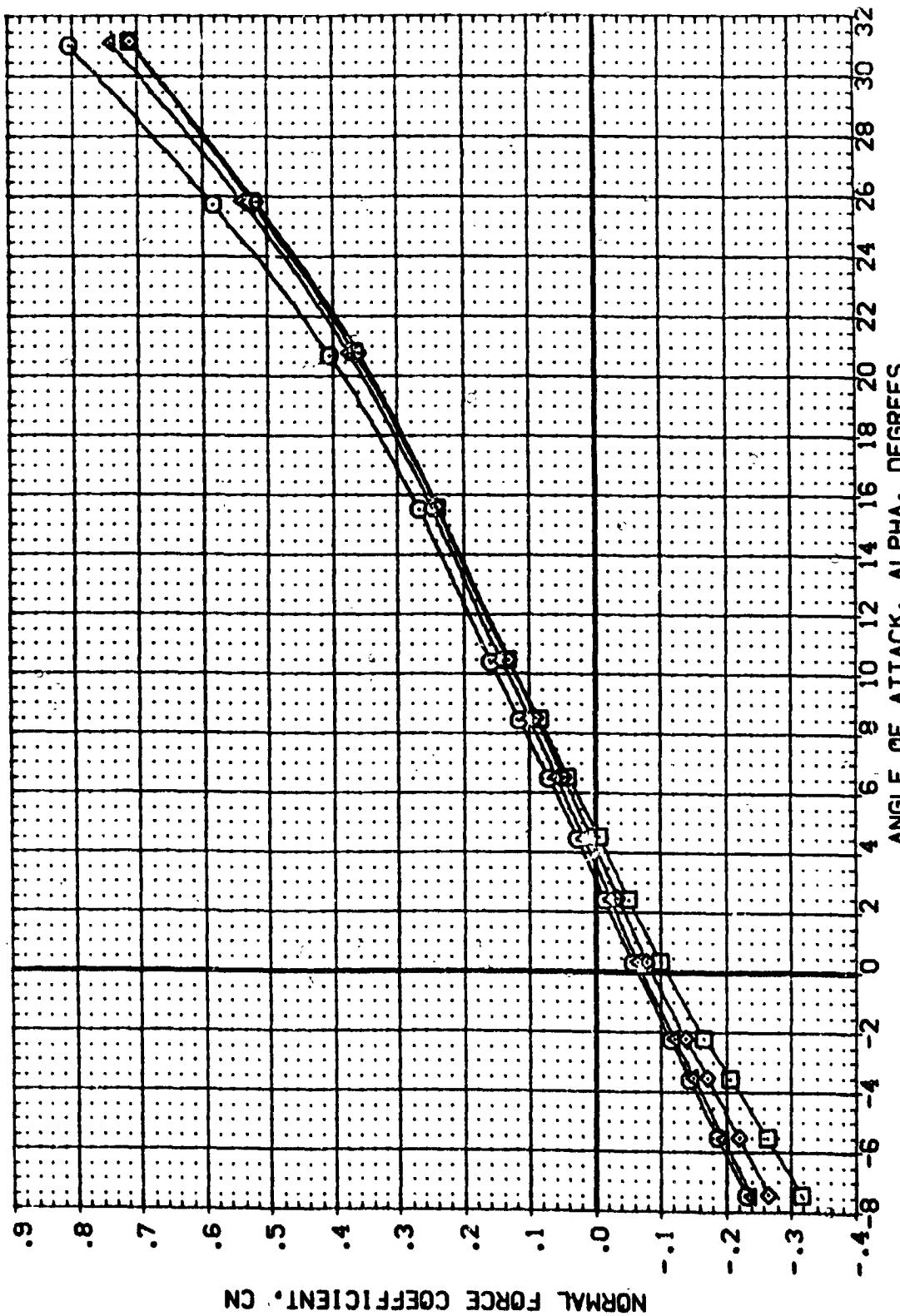


FIG. 6 ELEVON DEFLECTION WITHOUT FAIRING, LONGITUDINAL.
 (A)MACH = 7.32

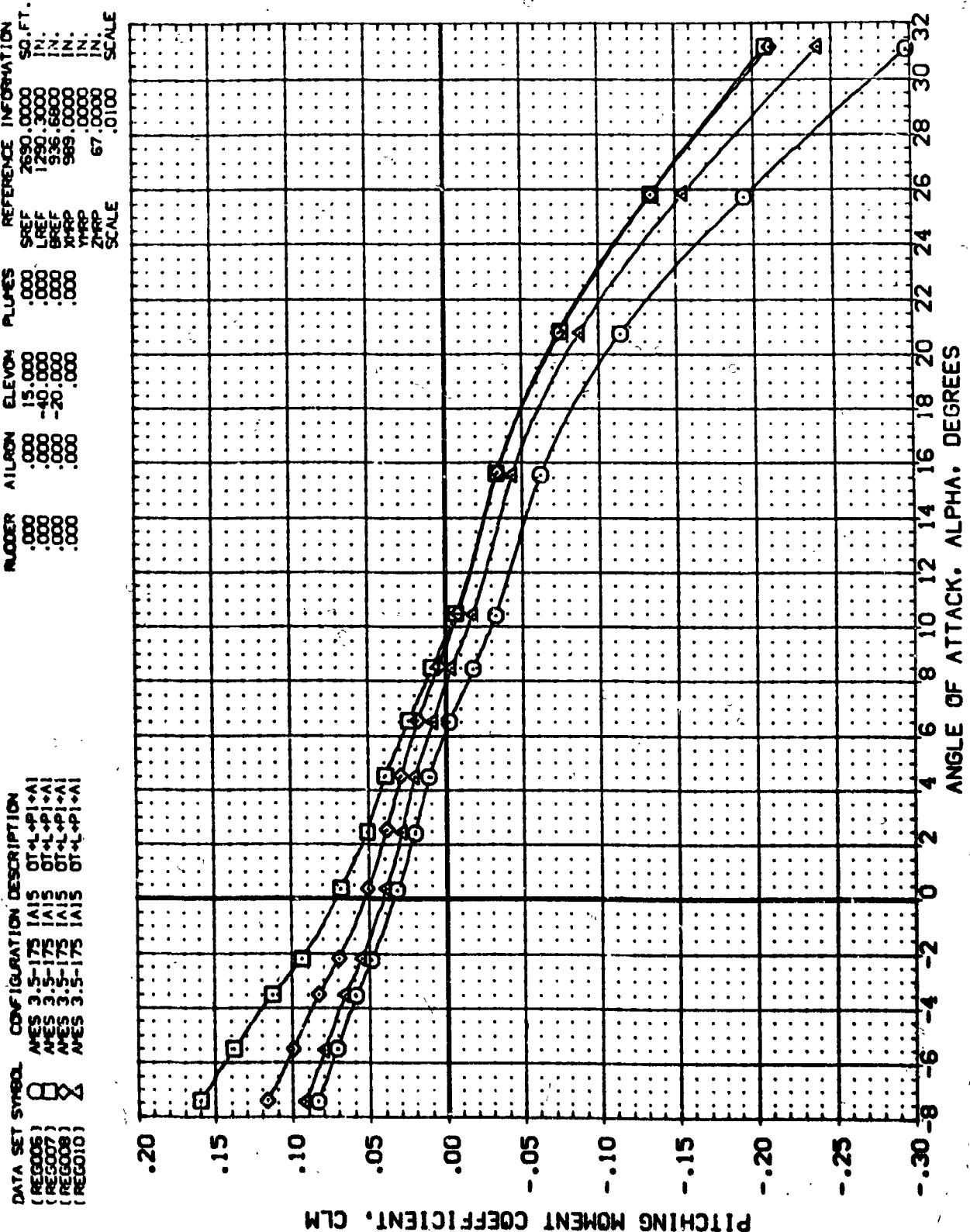


FIG. 6 ELEVON DEFLECTION WITHOUT FAIRING, LONGITUDINAL.
(A)MACH = 7.32

DATA SET SYMBOL. CONFIGURATION DESCRIPTION

| | | | | |
|----------|---|-------------|------|-------------|
| (REG006) | □ | AES 3.5-175 | TA15 | OT-L-4P1-A1 |
| (REG007) | ○ | AES 3.5-175 | TA15 | OT-L-4P1-A1 |
| (REG008) | × | AES 3.5-175 | TA15 | OT-L-4P1-A1 |
| (REG010) | △ | AES 3.5-175 | TA15 | OT-L-4P1-A1 |

RUDDER AIRLON ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 15.000 .000 SREF 2650.0000 SQ.FT.
 .000 .000 -40.000 .000 LREF 1290.3000 IN.
 .000 .000 -20.000 .000 BREF 936.6900 IN.
 .000 .000 .000 .000 XHPP 989.0000 IN.
 .000 .000 .000 .000 YHPP 67.0000 IN.
 .000 .000 .000 .000 ZHPP 67.0000 IN.
 SCALE .0100

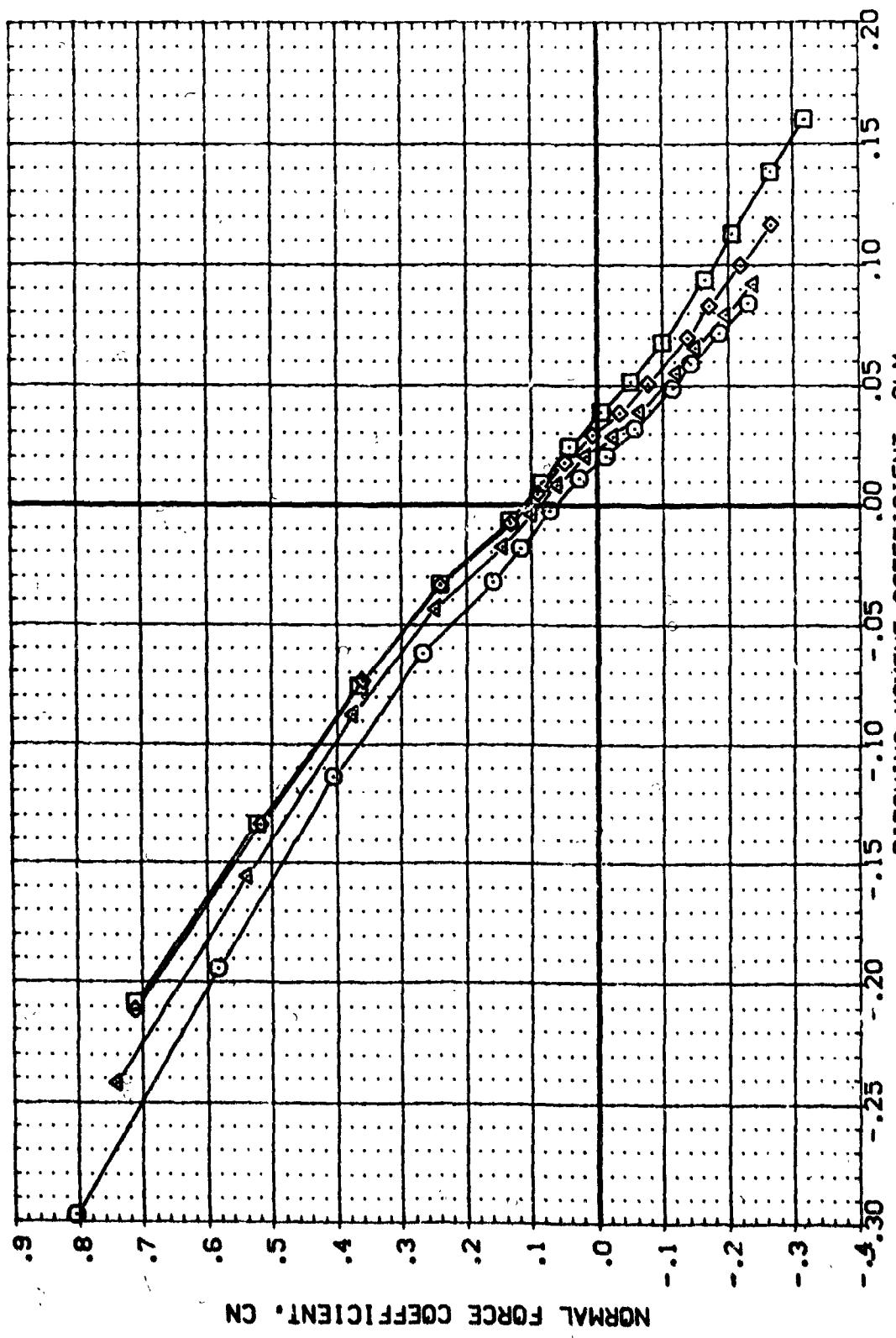


FIG. 6 ELEVON DEFLECTION WITHOUT FAIRING, LONGITUDINAL.

(A) MACH = 7.32

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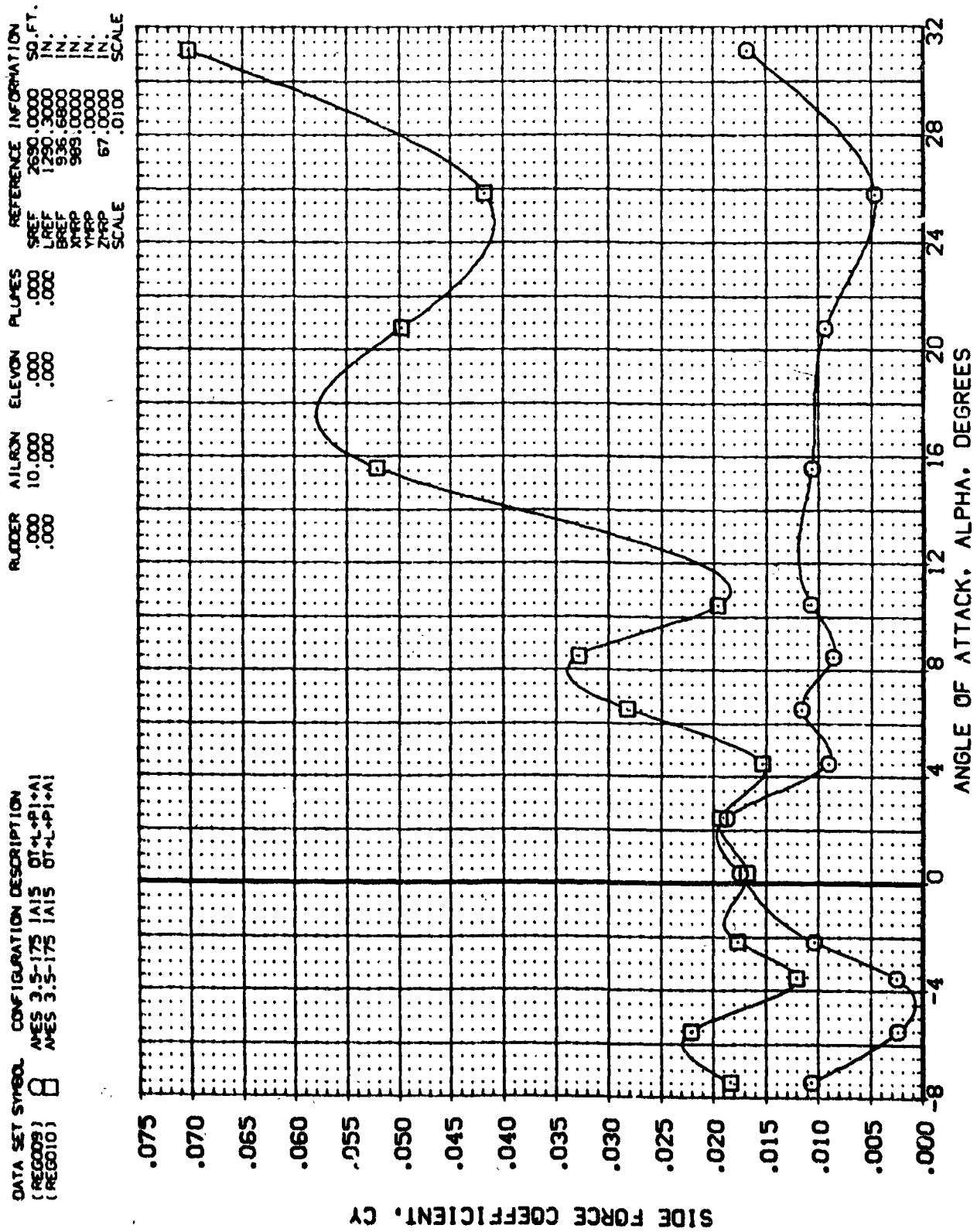


FIG. 7 AILERON DEFLECTION FAIRING, LATERAL-DIRECTIONAL.

(A)MACH = 7.32

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DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RECD09) 8 ANES 3-S-175 1A15 OT-L-AI
 (RECD10) 8 ANES 3-S-175 1A15 OT-L-AI

| RUDER | AILERON | ELEVON | PLUNES | REFERENCE INFORMATION |
|-------|---------|--------|--------|-----------------------|
| :000 | 10.000 | :000 | :000 | SREF 2680.0000 SQ.FT. |
| :000 | :000 | :000 | :000 | LREF 1250.3000 IN. |
| | | | | BREF 935.6800 IN. |
| | | | | XHPP 369.0000 IN. |
| | | | | YHPP 67.0000 IN. |
| | | | | ZHPP .0100 SCALE |

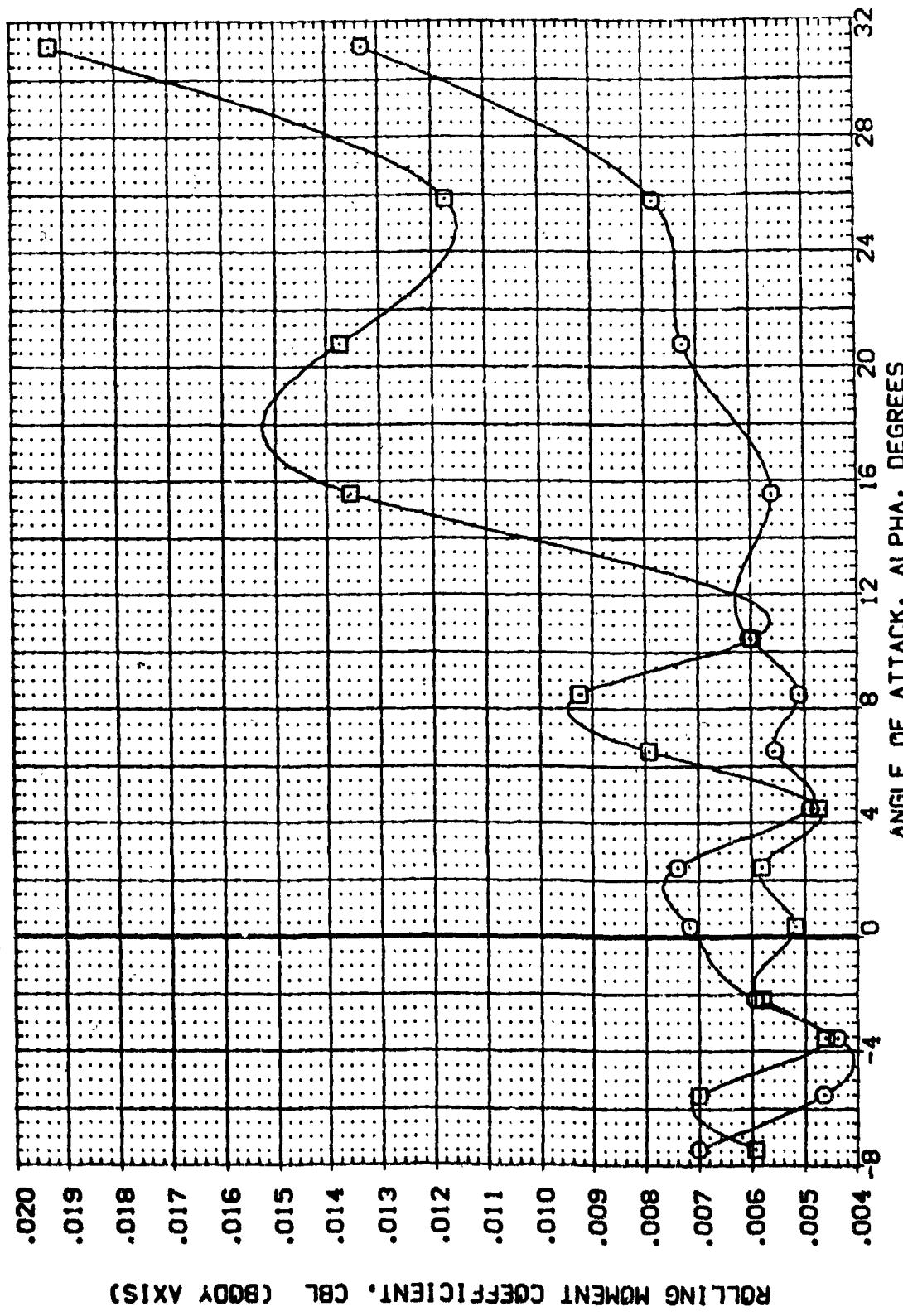


FIG. 7 AILERON DEFLECTION FAIRING, LATERAL-DIRECTIONAL.
 (AJMACH = 7.32)

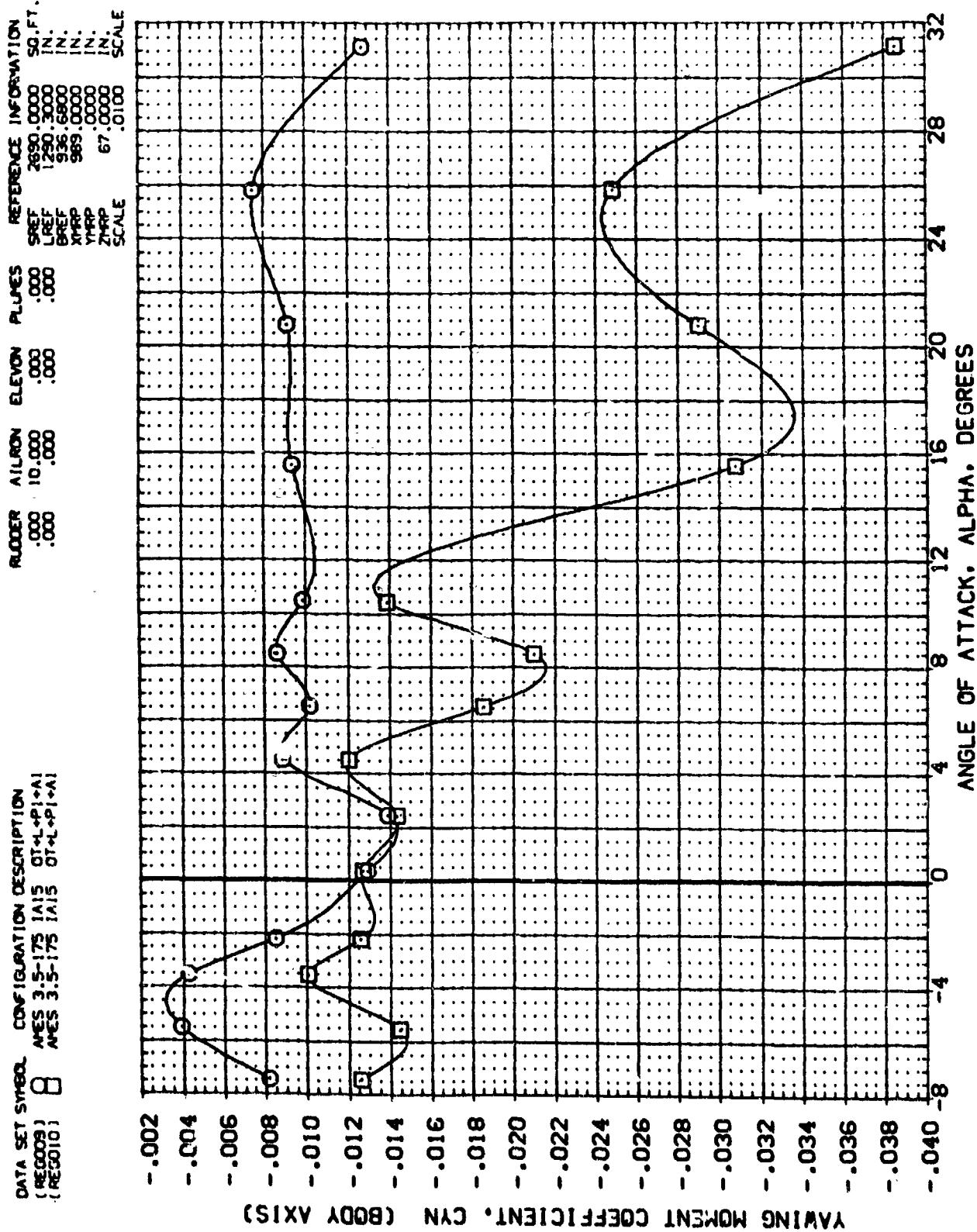


FIG. 7 AILERON DEFLECTION FAIRING, LATERAL-DIRECTIONAL.

(MACH = 7.32)

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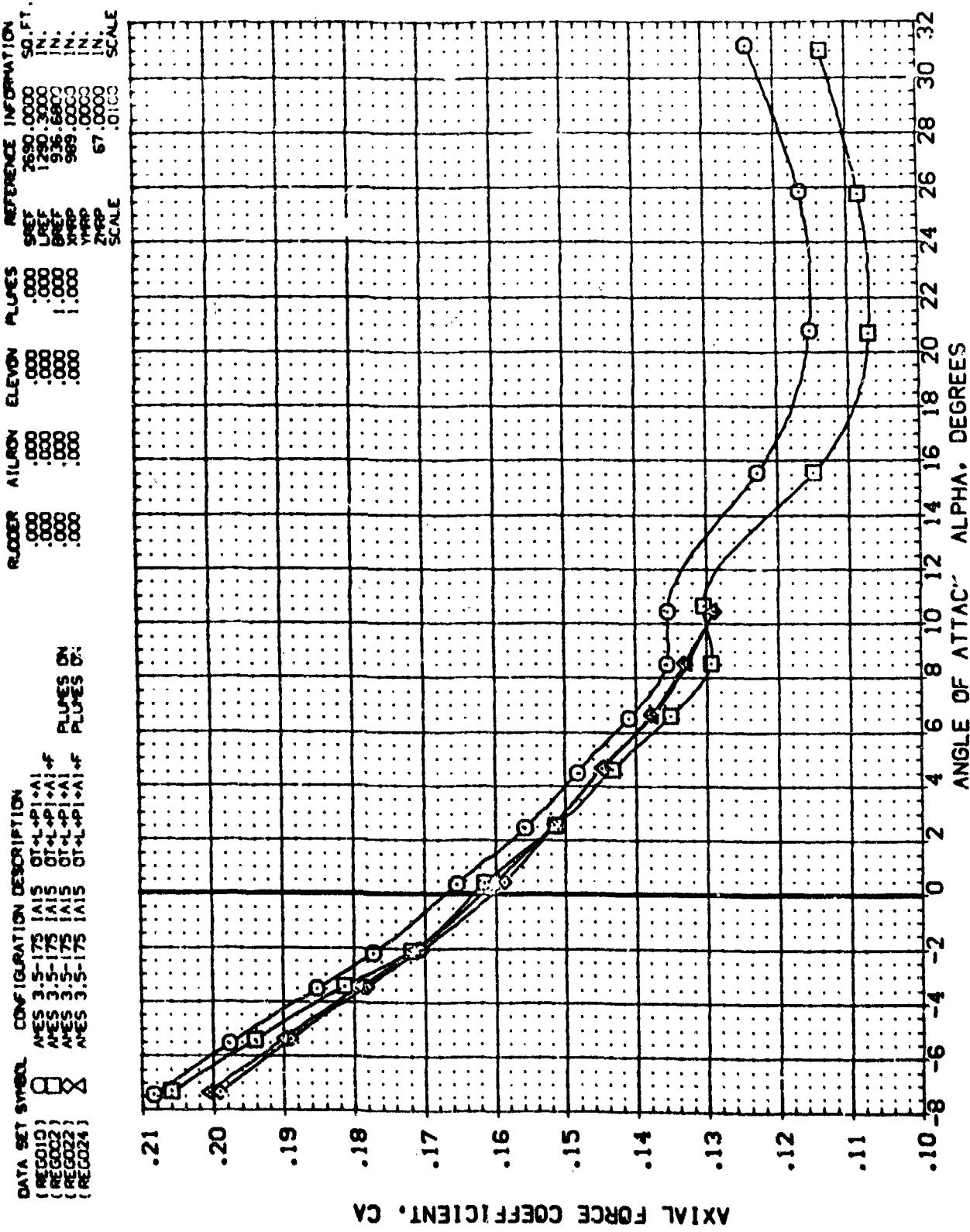


FIG. 8 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LONGITUDINAL.
(A)MACH = 7.32

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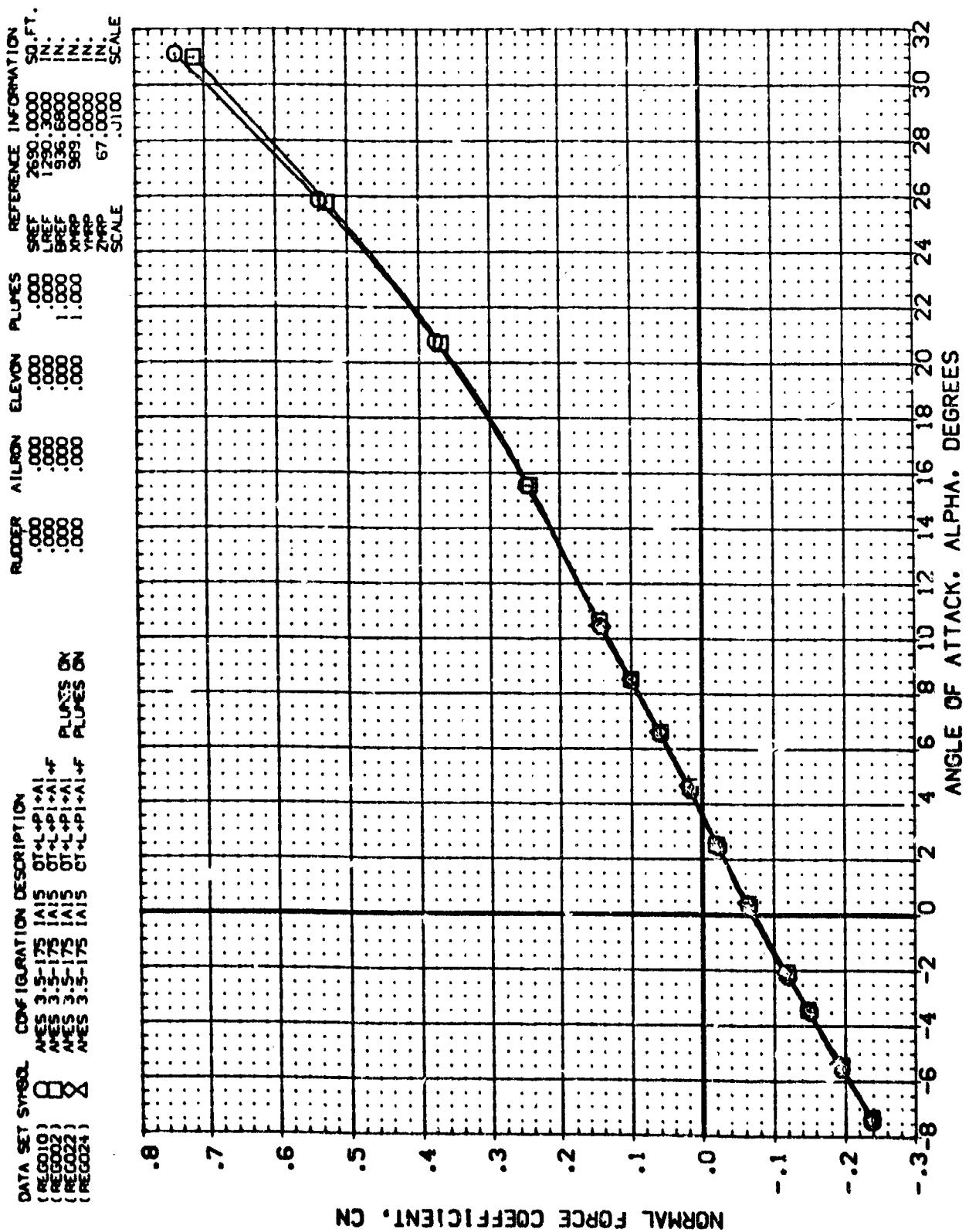


FIG. 8 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LONGITUDINAL.

(A)_{MACH} = 7.32

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DATA SET SYMBOL CONFIGURATION DESCRIPTION
 ANES 3.5-173 TAIS DT+L+P1+A1
 ANES 3.5-173 TAIS DT+L+P1+A1
 ANES 3.5-173 TAIS DT+L+P1+A1
 ANES 3.5-173 TAIS DT+L+P1+A1
 PLATES ON
 (REC010) (REC021)
 (REC022) (REC024)

RUDDER AIRRON ELEVON PLATES
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 SC. FT.
 SREF 2690 .0000
 LREF 1290 .0000
 BREF 1936 .0000
 XMRP 989 .0000
 YMRP 67 .0000
 ZMRP .0100
 SCALE

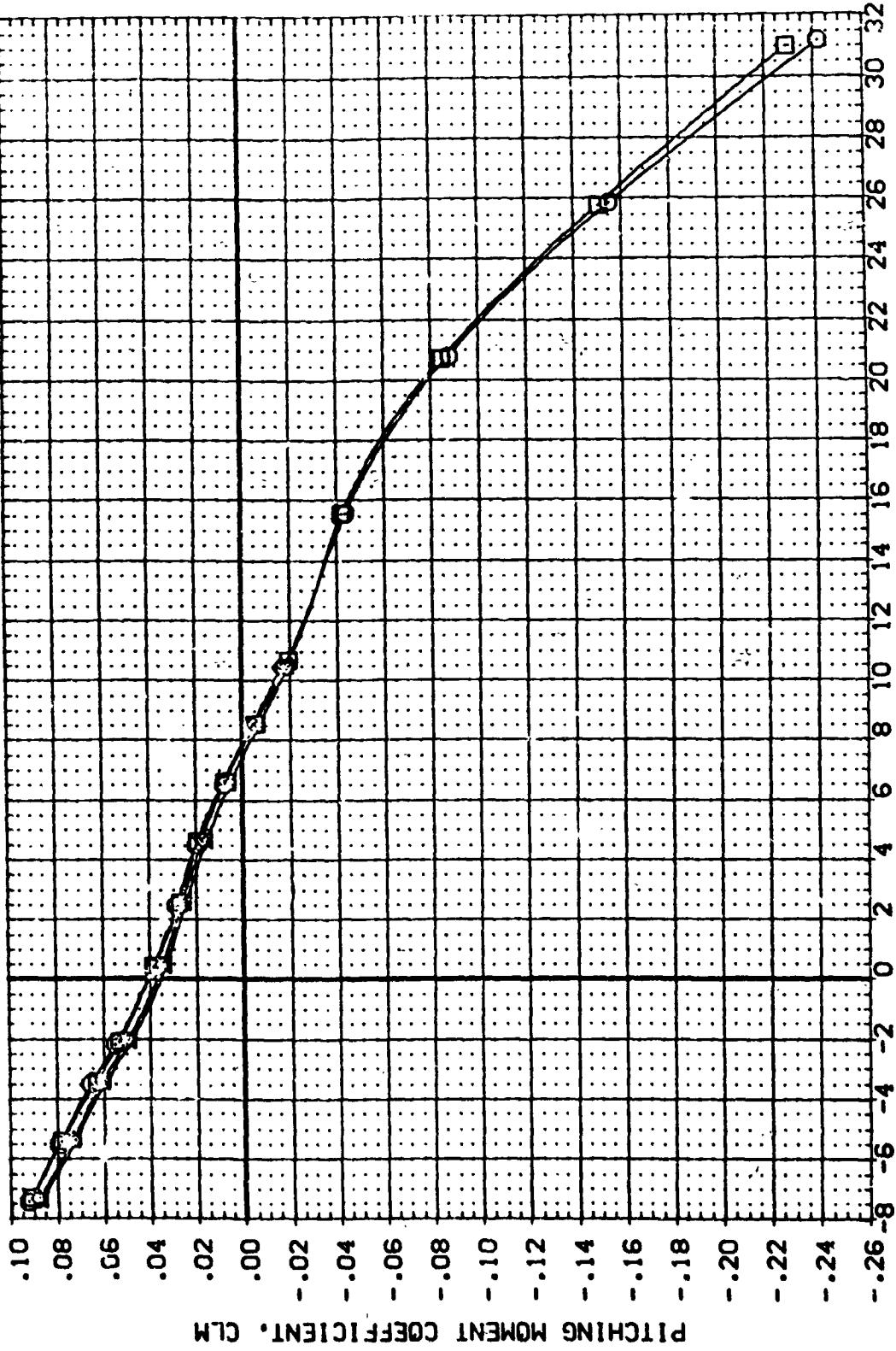


FIG. 8 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LONGITUDINAL.

(A)MACH = 7.32

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| DATA SET SYMBOL | CONFIGURATION DESCRIPTION |
|-----------------|----------------------------|
| (REF010) | AMES 3.5-75 AILS DT+L+P+A1 |
| (REF002) | AMES 3.5-75 AILS DT+L+P+A1 |
| (REF022) | AMES 3.5-75 AILS DT+L+P+A1 |
| (REF024) | AMES 3.5-75 AILS DT+L+P+A1 |

PLATES ON PLATES ON PLATES ON PLATES ON

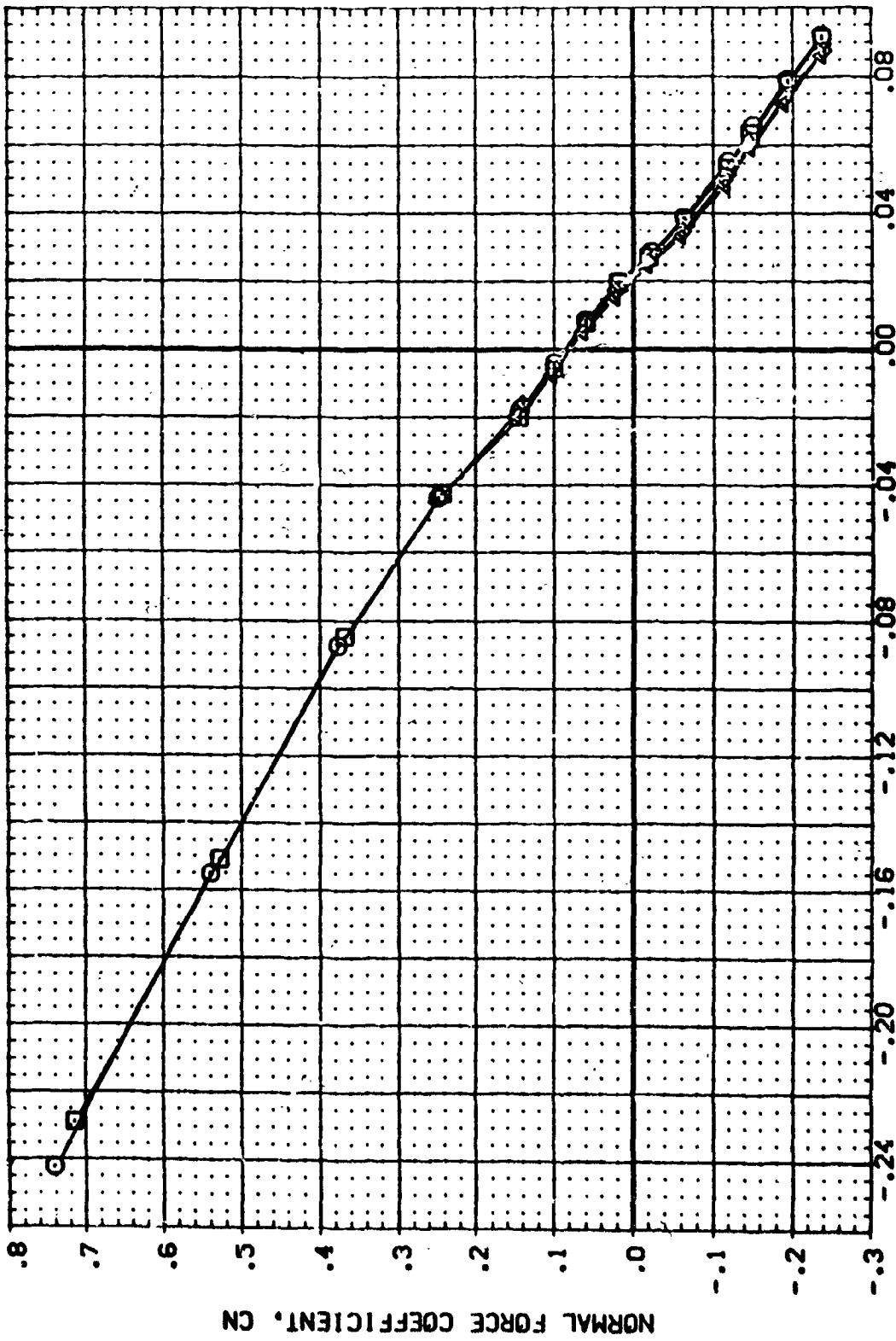


FIG. 8 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LONGITUDINAL.
 $(\Delta)MACH = 7.32$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RECD04) ASES 3.5-175 A15 DT-L-A15
 (RECD05) ASES 3.5-175 A15 DT-L-A15
 (RECD06) ASES 3.5-175 A15 DT-L-A15
 (RECD07) ASES 3.5-175 A15 DT-L-A15

RUDDER ALIUDN ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 -40.000 .000 SHEF 2690.0000 SO.FT.
 .000 .000 -15.000 .000 LREF 1250.3000 IN.
 .000 .000 15.000 .000 BREF 936.6800 IN.
 .000 .000 -40.000 .000 XREF 999.0000 IN.
 .000 .000 67.0000 .000 YREF 67.0000 IN.
 .000 .000 .000 .000 ZREF .0100 IN.
 SCALE

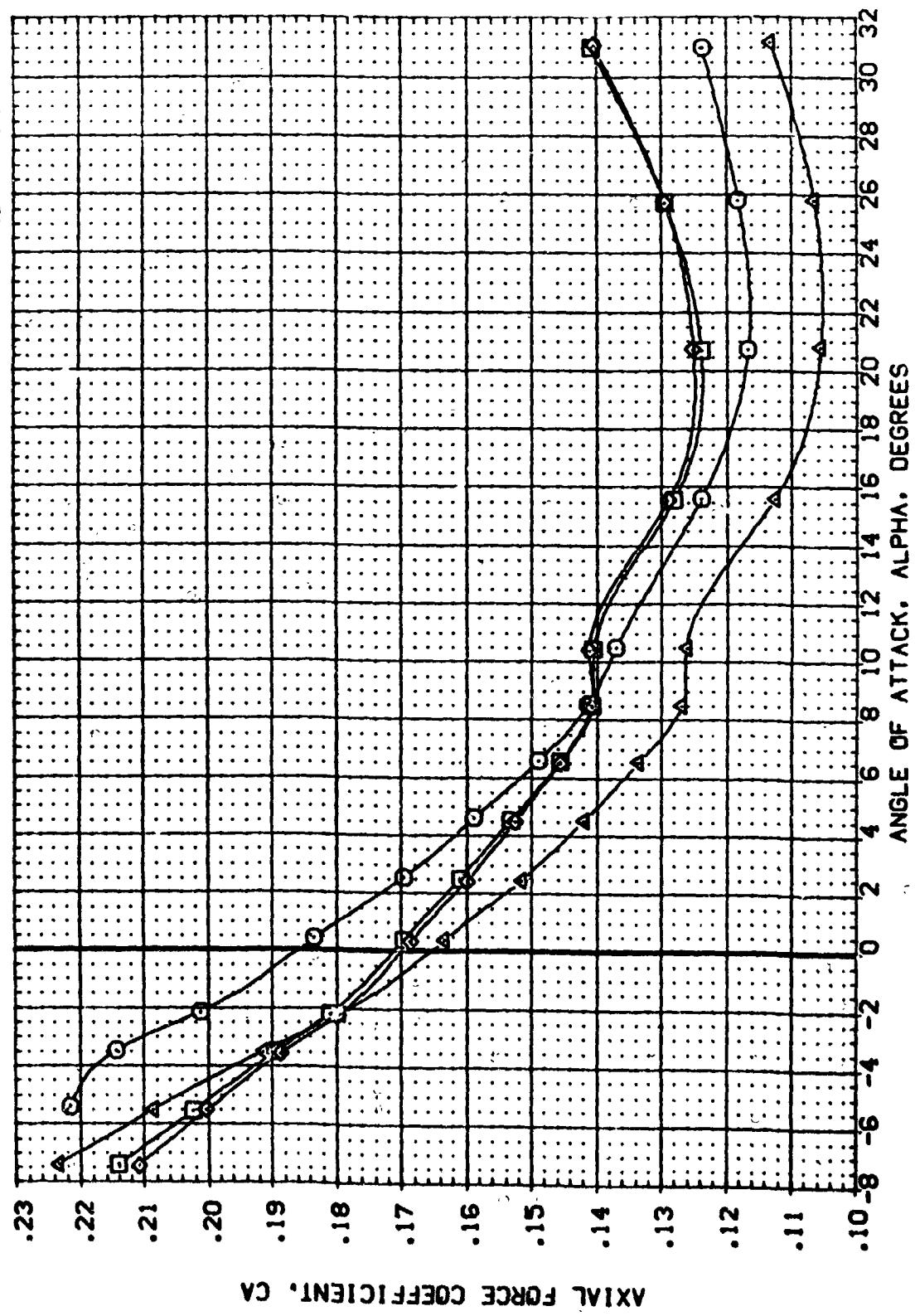


FIG. 9 WITH AND WITHOUT FAIRING, DE = 15 AND -40 DEGS.. LONGITUDINAL.
 (A)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | |
|----------|--------------|------|--------------|
| (REF004) | AMES 3.5-175 | 1A15 | DT+L+P1+A1+P |
| (REF005) | AMES 3.5-175 | 1A15 | DT+L+P+AI+P |
| (REF006) | AMES 3.5-175 | 1A15 | DT+L+P1+A1 |
| (REF007) | AMES 3.5-175 | 1A15 | DT+L+P1+A1 |

RUDDER ALIEN PLATES ELEVON
 .000 .000 -40.000
 .000 .000 15.000
 .000 .000 15.000
 .000 .000 -40.000
 .000 .000 .000
 REFERENCE INFORMATION
 SREF 2680.0000 SQ.FT.
 LREF 1280.3000 IN.
 BREF 936.6800 IN.
 XHPP 989.0000 IN.
 YHPP 67.0000 IN.
 ZHPP .0100 SCALE

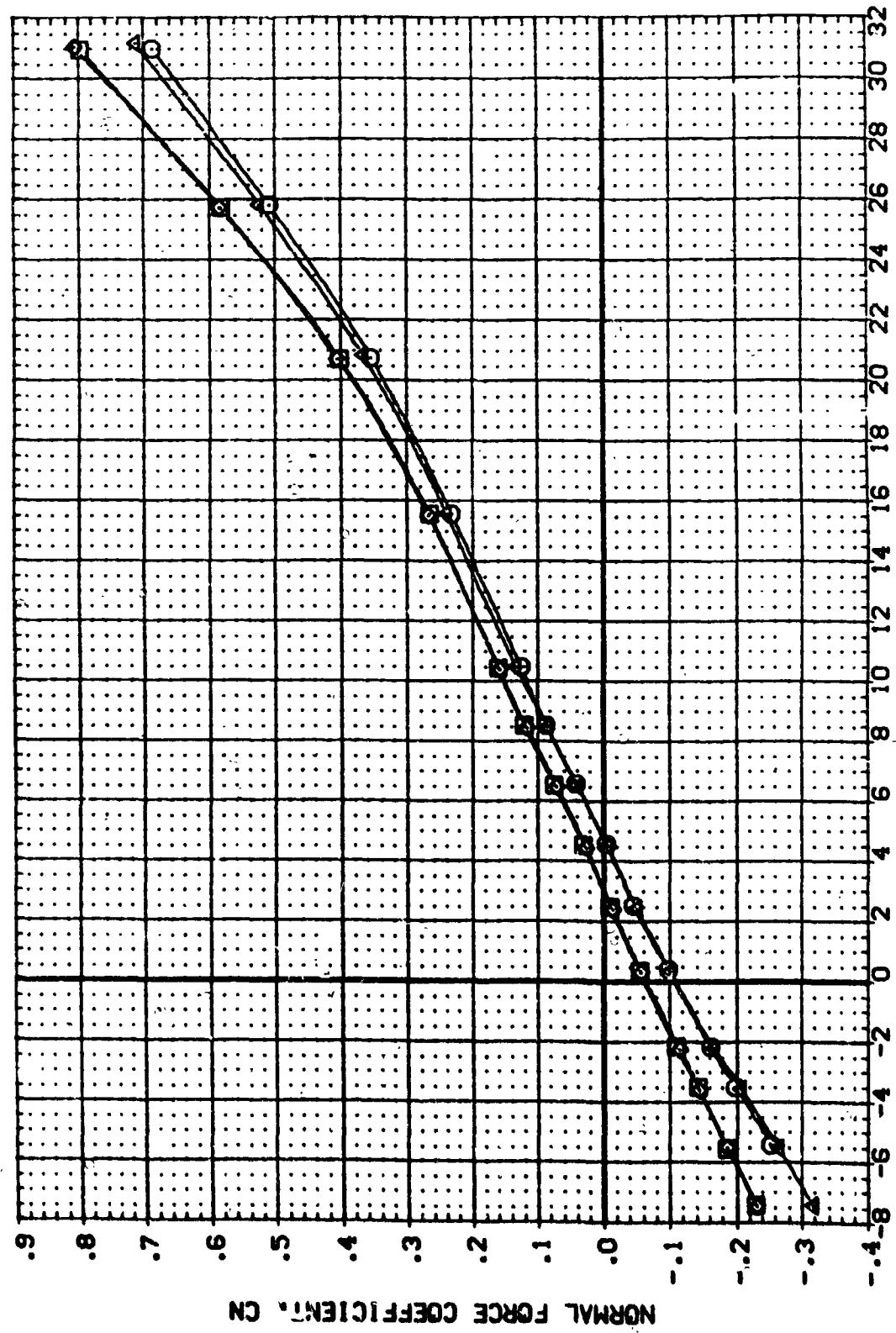


FIG. 9 WITH AND WITHOUT FAIRING. DE = 15 AND -40 DEGS., LONGITUDINAL.
 (A)MACH = 7.32

| DATA SET SYMBOL | CONFIGURATION DESCRIPTION |
|-----------------|-----------------------------|
| (REG004) | ANES 3.5-175 AIS OT+L+p+A+f |
| (REG005) | ANES 3.5-175 AIS OT+L+p+A+f |
| (REG006) | ANES 3.5-175 AIS OT+L+p+A+f |
| (REG007) | ANES 3.5-175 AIS OT+L+p+A+f |

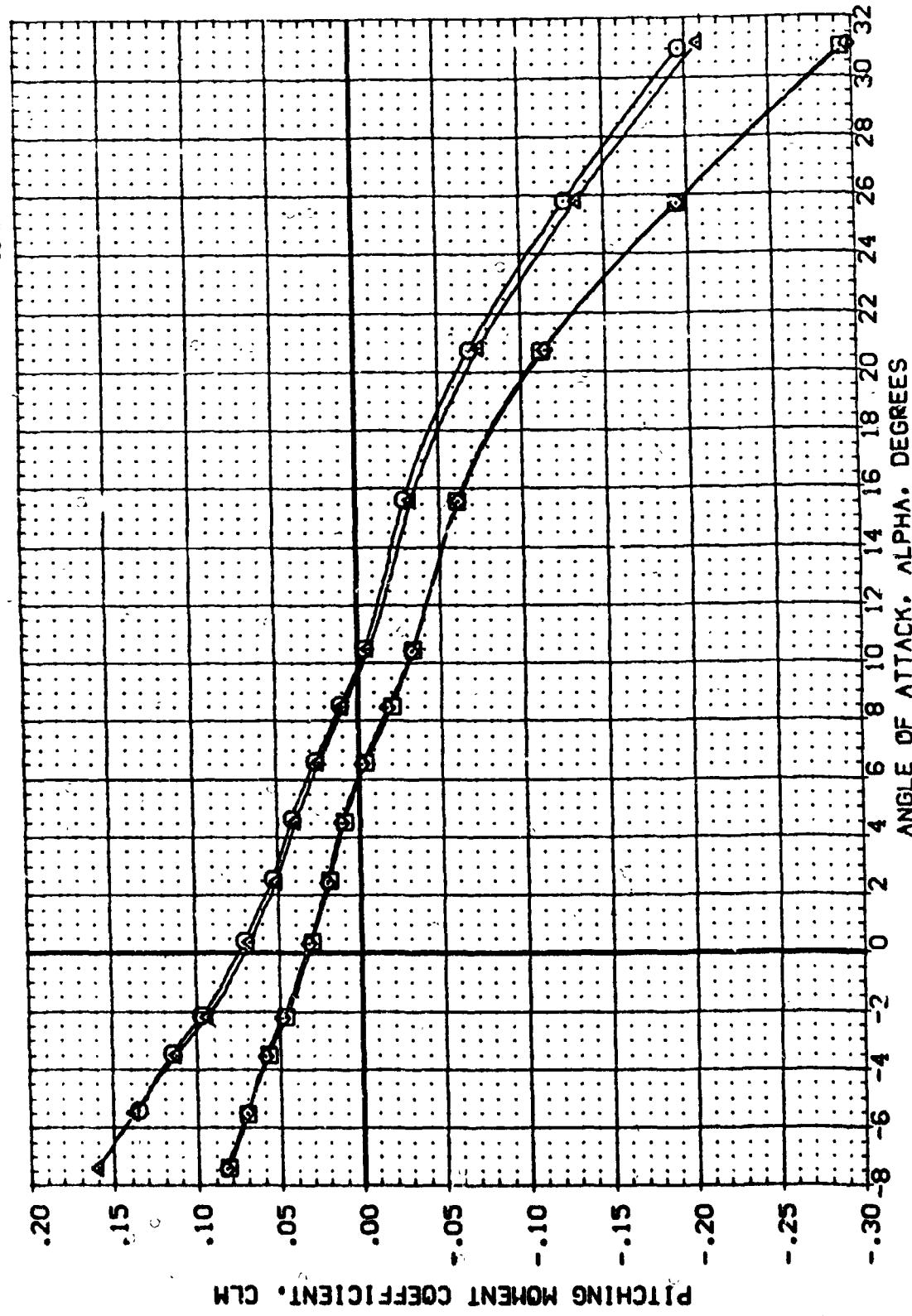


FIG. 9 WITH AND WITHOUT FAIRING, DE = 15 AND -40 DEGS., LONGITUDINAL.

$C_{\text{MACH}} = 7.32$

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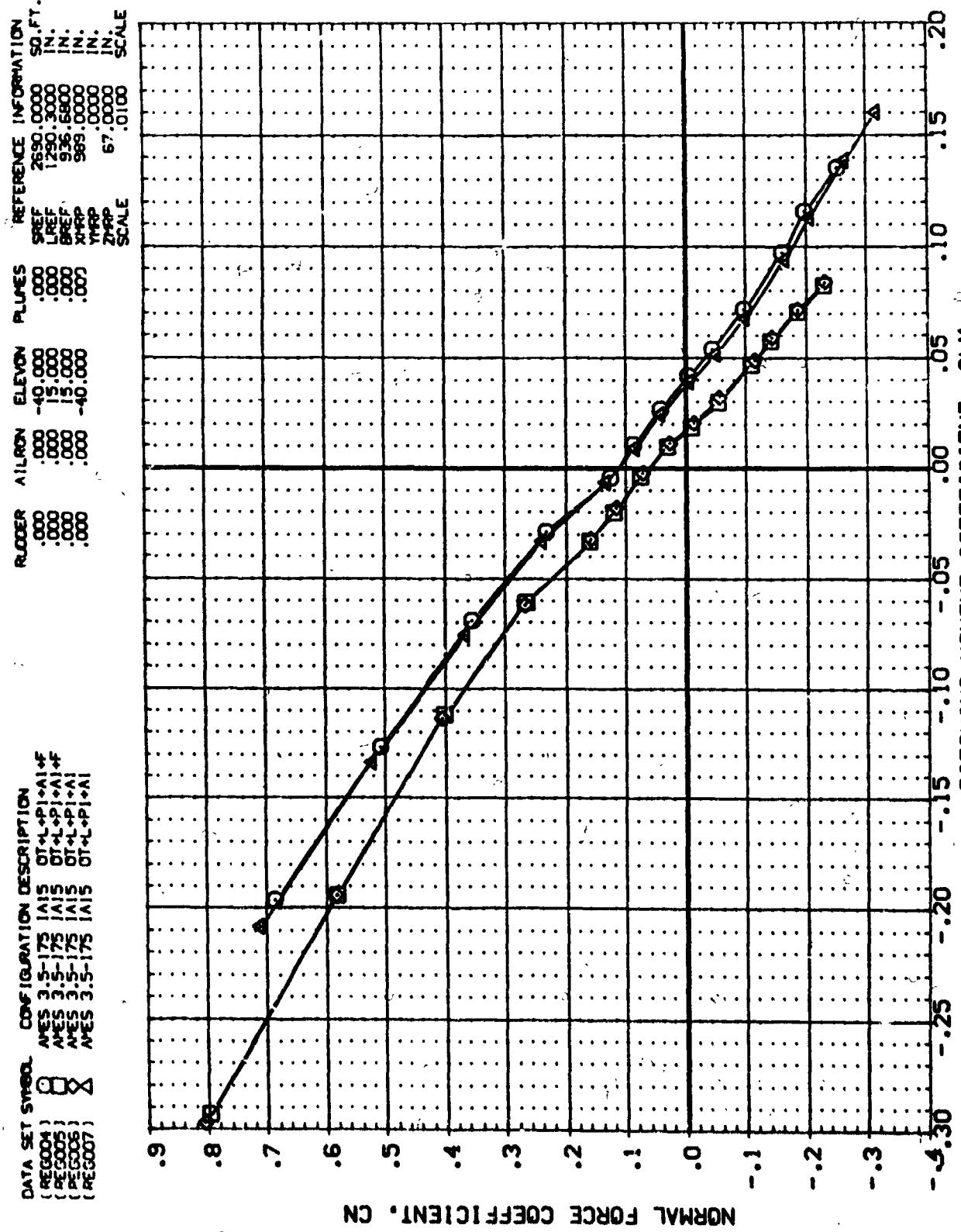


FIG. 9 WITH AND WITHOUT FAIRING, DE = 15 AND -40 DEGS., LONGITUDINAL.
 $(\text{A}) \text{MACH} = 7.32$

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DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RECD01) NES 3.5-173 1A15 DT+L+P 1+A1+F
 (RECD06) NES 3.5-173 1A15 DT+L+P 1+A1+F
 (RECD09) NES 3.5-173 1A15 DT+L+P 1+A1+F
 (RECD21) NES 3.5-173 1A15 DT+L+P 1+A1+F

RUDDER AIRLON PLUMES ELEVON PLUMES REFERENCE INFORMATION
 .000 .000 .000 .000 SREF 2690 .0000 SQ.FT.
 .000 .000 .000 .000 LREF 1290 .3000 IN.
 .000 .000 .000 .000 BREF 935 .6800 IN.
 .000 .000 .000 .000 XHPP 989 .0000 IN.
 .000 .000 .000 .000 ZHPP 67 .0000 IN.
 SCALE .0100

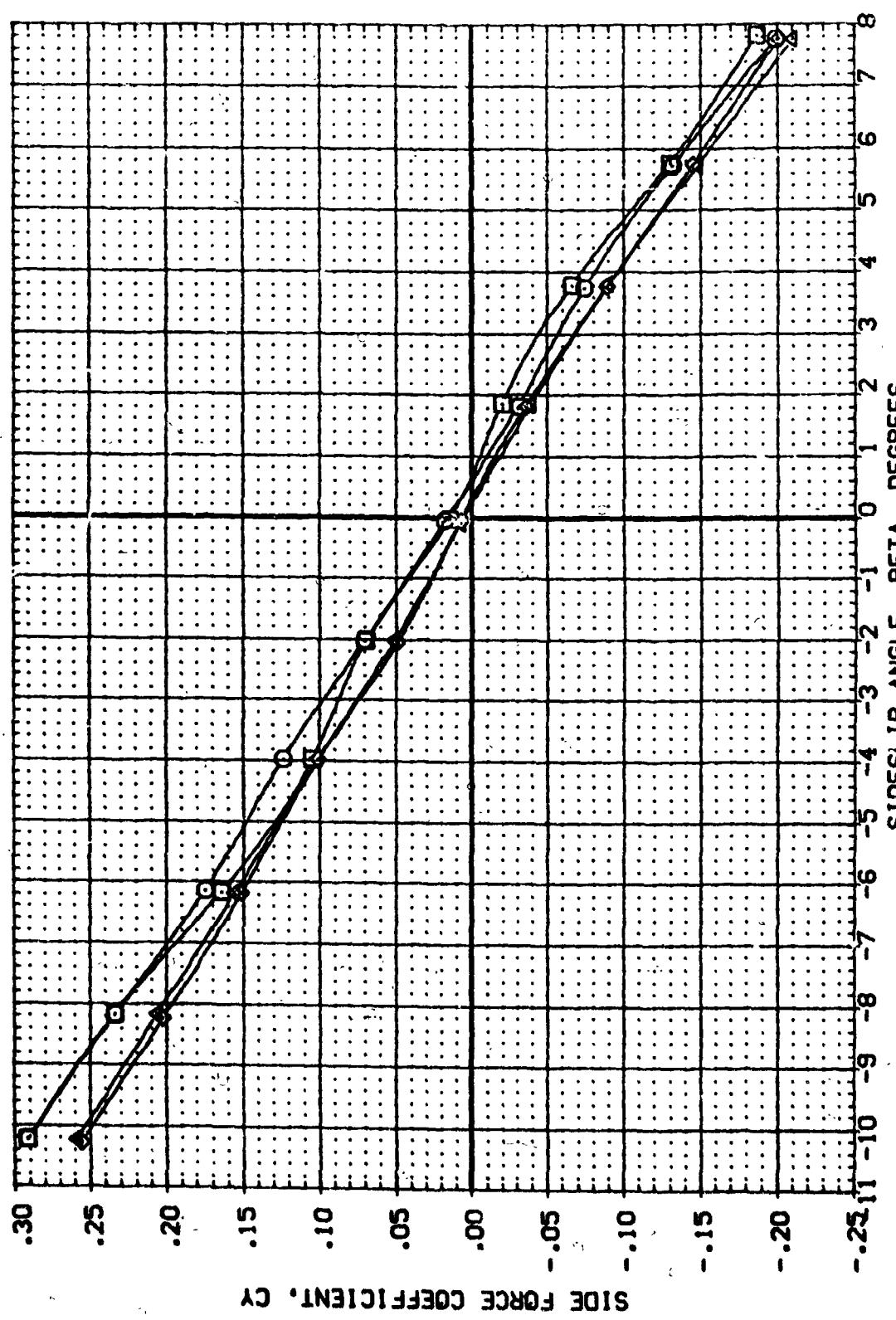


FIG. 10 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LATERAL-DIRECTIONAL.
 CA(MACH) = 7.32

DATA SET SPEED. CONFIGURATION DESCRIPTION
 (REGO14) AMES 3.5-175 TAIS DT4-SP1+A1+F
 (REGO16) AMES 3.5-175 TAIS DT4-SP1+A1
 (REGO19) AMES 3.5-175 TAIS DT4-SP1+A1+F PLATES ON
 (REGO21) AMES 3.5-175 TAIS DT4-SP1+A1

RUDDER SAILFIN ELEVON PLATES REFERENCE INFORMATION
 .000 .000 .000 SREF 2690.0000 SQ.FT.
 .000 .000 .000 LREF 1290.3000 IN.
 .000 .000 .000 BREF 936.6000 IN.
 .000 .000 .000 XRP 989.0000 IN.
 .000 .000 .000 YRP 0.0000 IN.
 .000 .000 .000 ZRP 67.0000 IN.
 SCALE .0100

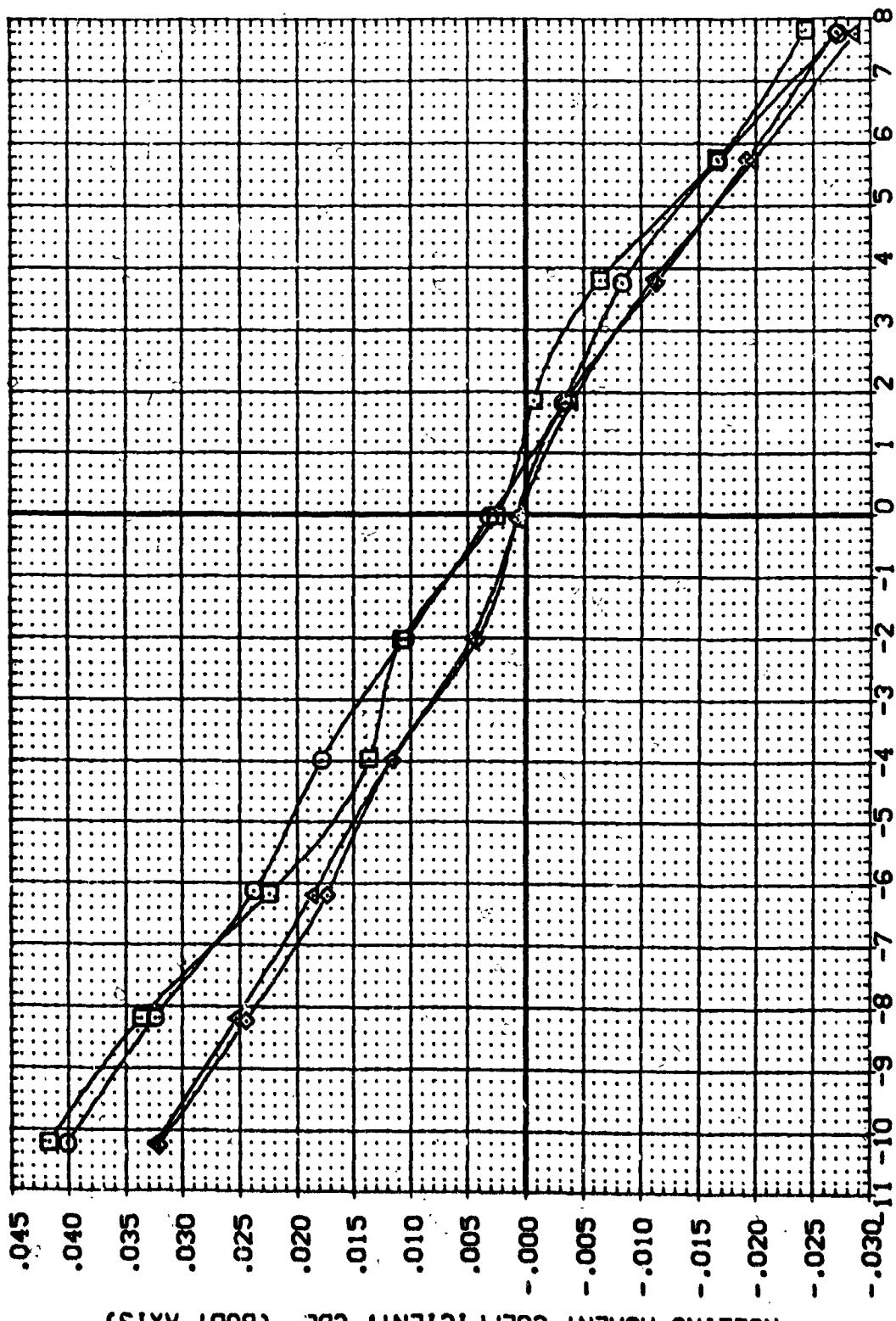


FIG. 10 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LATERAL-DIRECTIONAL.
 (AJMACH = 7.32)

| DATA SET SYMBOL | CONFIGURATION DESCRIPTION | RUDDER | AIRSON | ELEVON | PLATES | REFERENCE INFORMATION |
|-----------------|---|--------|--------|--------|--------|-----------------------|
| (REG014) | AMES 3.5-175 IAI5 OT-L- ^{1/2} -A1-A1-f | .000 | .000 | .000 | .000 | REF 2690.0000 IN. |
| (REG016) | AMES 3.5-175 IAI5 OT-L- ^{1/2} -A1-A1-f | .000 | .000 | .000 | .000 | REF 1290.3000 IN. |
| (REG019) | AMES 3.5-175 IAI5 OT-L- ^{1/2} -A1-A1-f | .000 | .000 | .000 | .000 | REF 936.6800 IN. |
| (REG021) | AMES 3.5-175 IAI5 OT-L- ^{1/2} -A1-A1-f | .000 | .000 | .000 | .000 | REF 989.0000 IN. |
| | | | | | | WIND 67.0000 IN. |
| | | | | | | SCALE .0100 |

PLATES ON

PLATES OFF

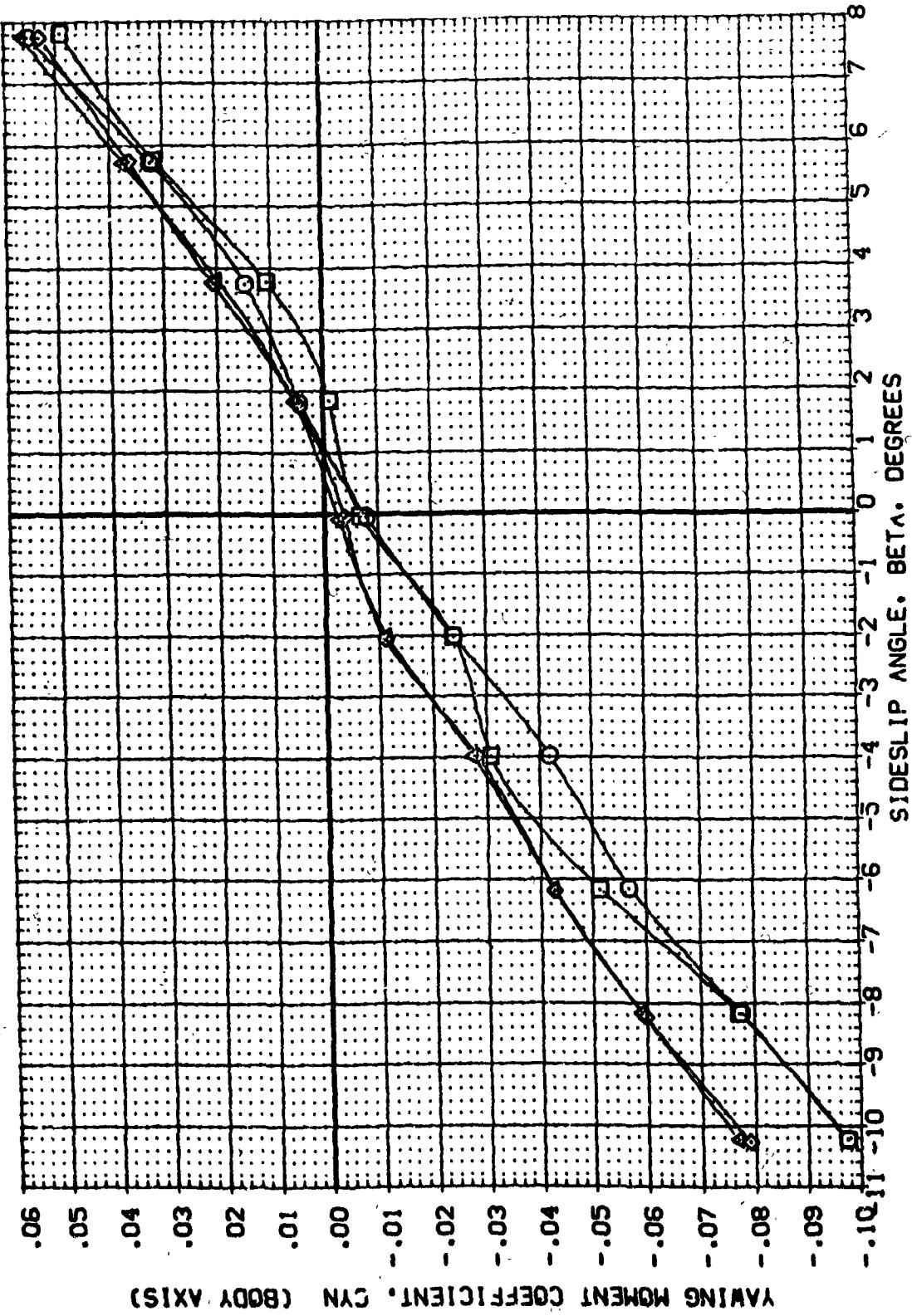


FIG. 10 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LATERAL-DIRECTIONAL.
(MACH = 7.32)

DATA SET SYMBOL. CONFIGURATION DESCRIPTION

| | |
|----------|---|
| (REGO14) | AMES 3.5-175 IAS OT-LAP-AI-4F |
| (REGO16) | AMES 3.5-175 IAS OT-LAP-AI-4F PLUMES ON |
| (REGO19) | AMES 3.5-175 IAS OT-LAP-AI-4F PLUMES ON |
| (REGO21) | AMES 3.5-175 IAS OT-LAP-AI-4F |

RUDDER AILERON ELEVON PLUMES REFERENCE INFORMATION

.000 .000 .000 .000 SREF 2690.0000 SQ.FT.

.000 .000 .000 .000 LREF 1290.3000 IN.

.000 .000 .000 .000 BREF 935.6800 IN.

.000 .000 .000 .000 XMRP 989.0000 IN.

.000 .000 .000 .000 YMRP 67.0000 IN.

SCALE .0100

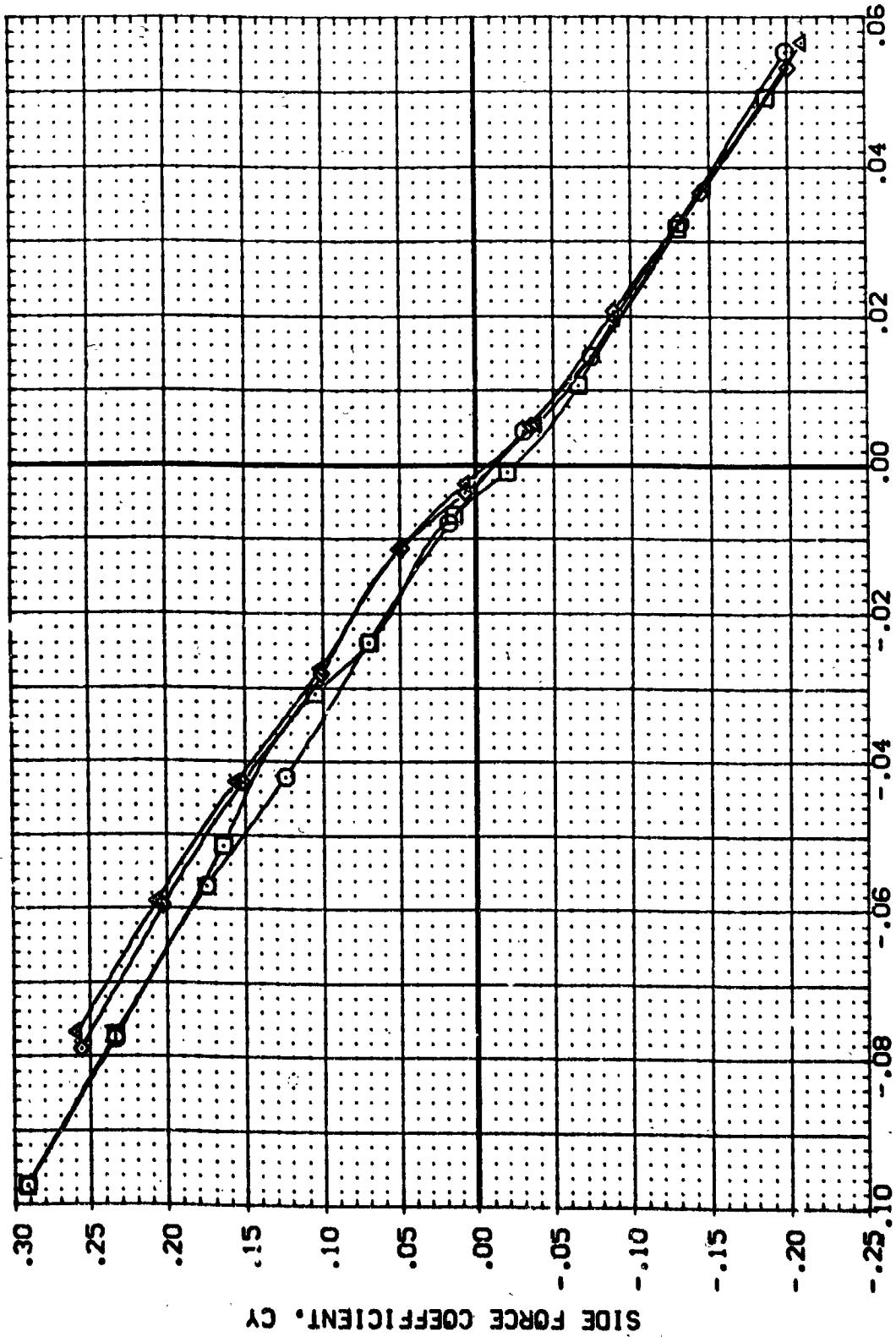


FIG. 10 POWER ON AND OFF, WITH AND WITHOUT FAIRING, LATERAL-DIRECTIONAL.
(A)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REF003) ANES 3.5-175 LARIS 3.5-175 PLATES ON
 (REF023) ANES 3.5-175 LARIS 3.5-175 PLATES ON

RUDDER AIRLOAD ELEV-N PLATES REFERENCE INFLUENTIAL.
 -20.000 .000 .000 .000 SREF 2690 0000 50.FT.
 -20.000 .000 .000 1.000 LREF 1250 3000 IN.
 BREF 936.6800 IN.
 XRP 989.0000 IN.
 YRP 67.0000 IN.
 ZRP .0000 IN.
 SCALE .0100

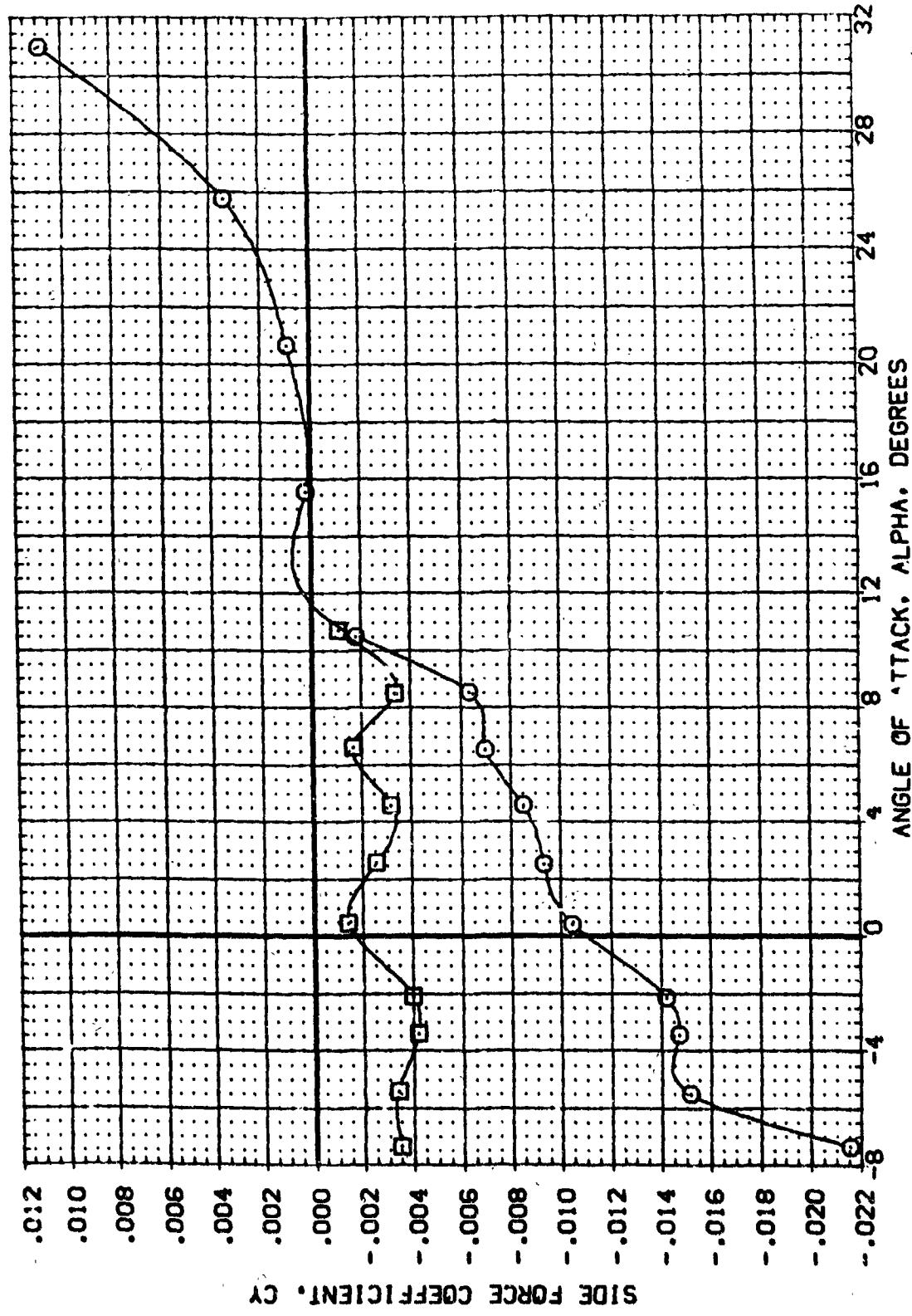


FIG. 11 RUDDER DEFLECTION, POWER ON AND OFF, LATERAL-DIRECTIONAL (PITCH).
 (MACH = 7.32

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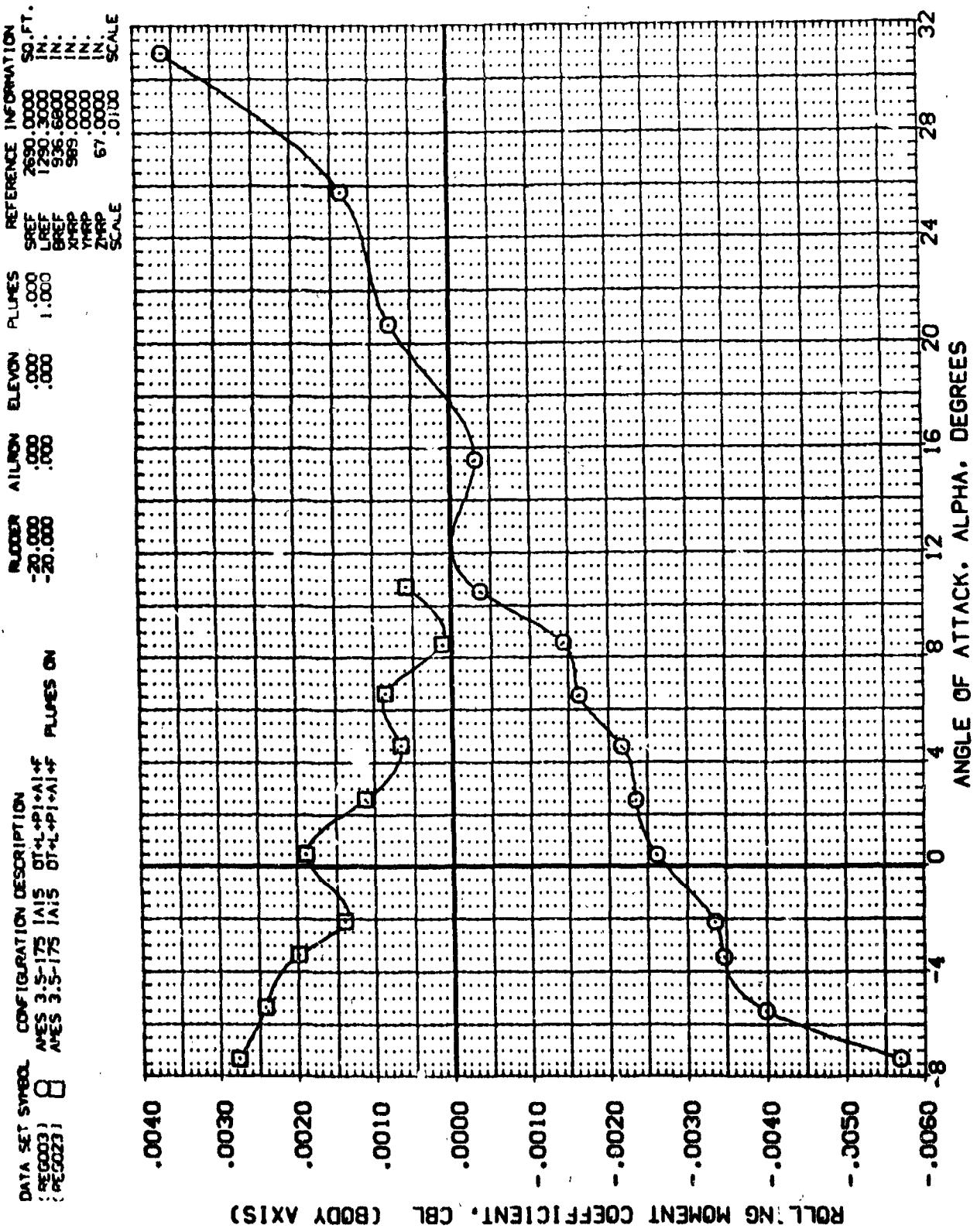


FIG. 11 RUDDER DEFLECTION, POWER ON AND OFF, LATERAL-DIRECTIONAL (PITCH).
 $(\text{A})\text{MACH} = 7.32$

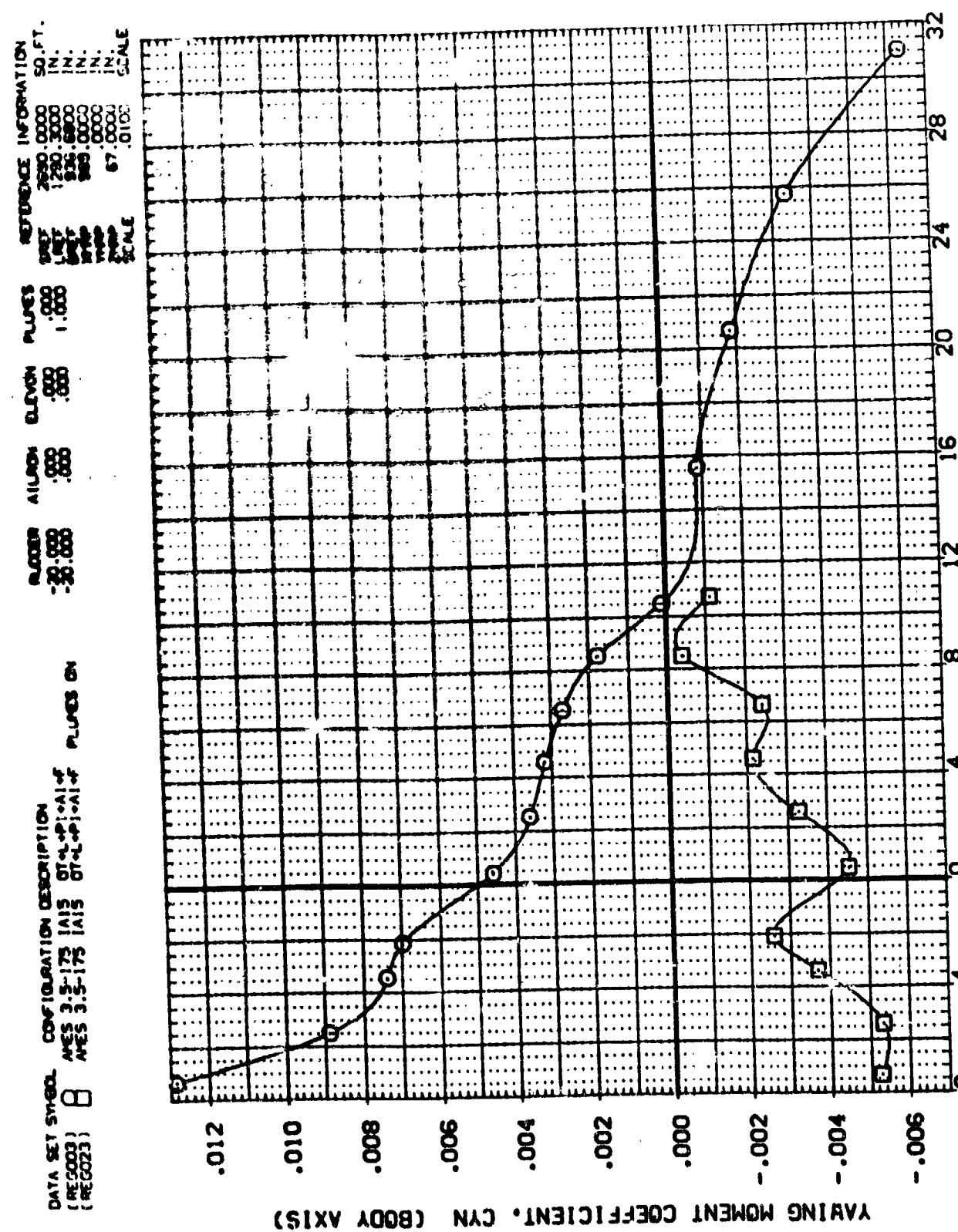


FIG. 11 RUDDER DEFLECTION, POWER ON AND OFF, LATERAL-DIRECTIONAL (PITCH).
 (A) MACH = 7.32

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DATA SET NAME: CONFIGURATION DESCRIPTION
 (RECORDS): 8 AMES 3.5-175 IAI5 OT4-L91-A14F PLUSES ON
 (RECORD): 8 AMES 3.5-175 IAI5 OT4-L91-A14F

| REFERENCE INFORMATION | | | |
|-----------------------|---------|----------|--------|
| RUDDER | AIRBORN | ELEVATOR | PLATES |
| -20,000 | .000 | .000 | 1,000 |
| -20,000 | .000 | .000 | 1,000 |

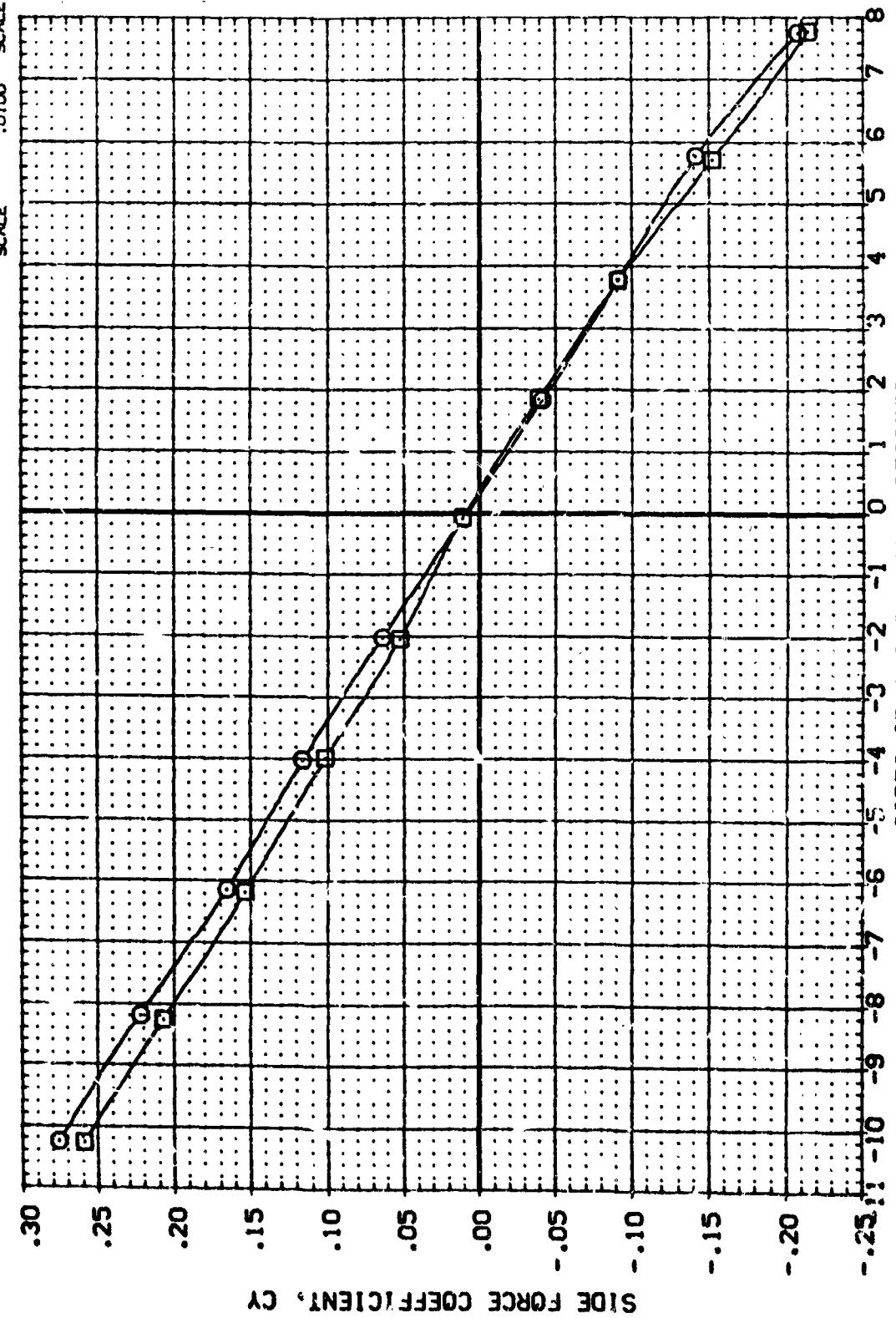


FIG. 12 RUDDER DEFLECTION. POWER ON AND OFF. LATERAL-DIRECTIONAL.

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REG015) □ MACH 3.5-175 LADS 01-L+1-X14F PLATES ON

RUDDER AIL, ROLL PLATES ELEVON PLATES
 -20.000 .000 .000 .000 .000 .000
 -20.000 .000 .000 .000 .000 .000
 REFERENCE INFORMATION
 SREF 2690.0000 SQ.FT.
 LREF 1250.3000 IN.
 BREF 935.6800 IN.
 XMPF 389.0000 IN.
 YMPF 67.0000 IN.
 ZMPF 67.0000 IN.
 SCALE .0100

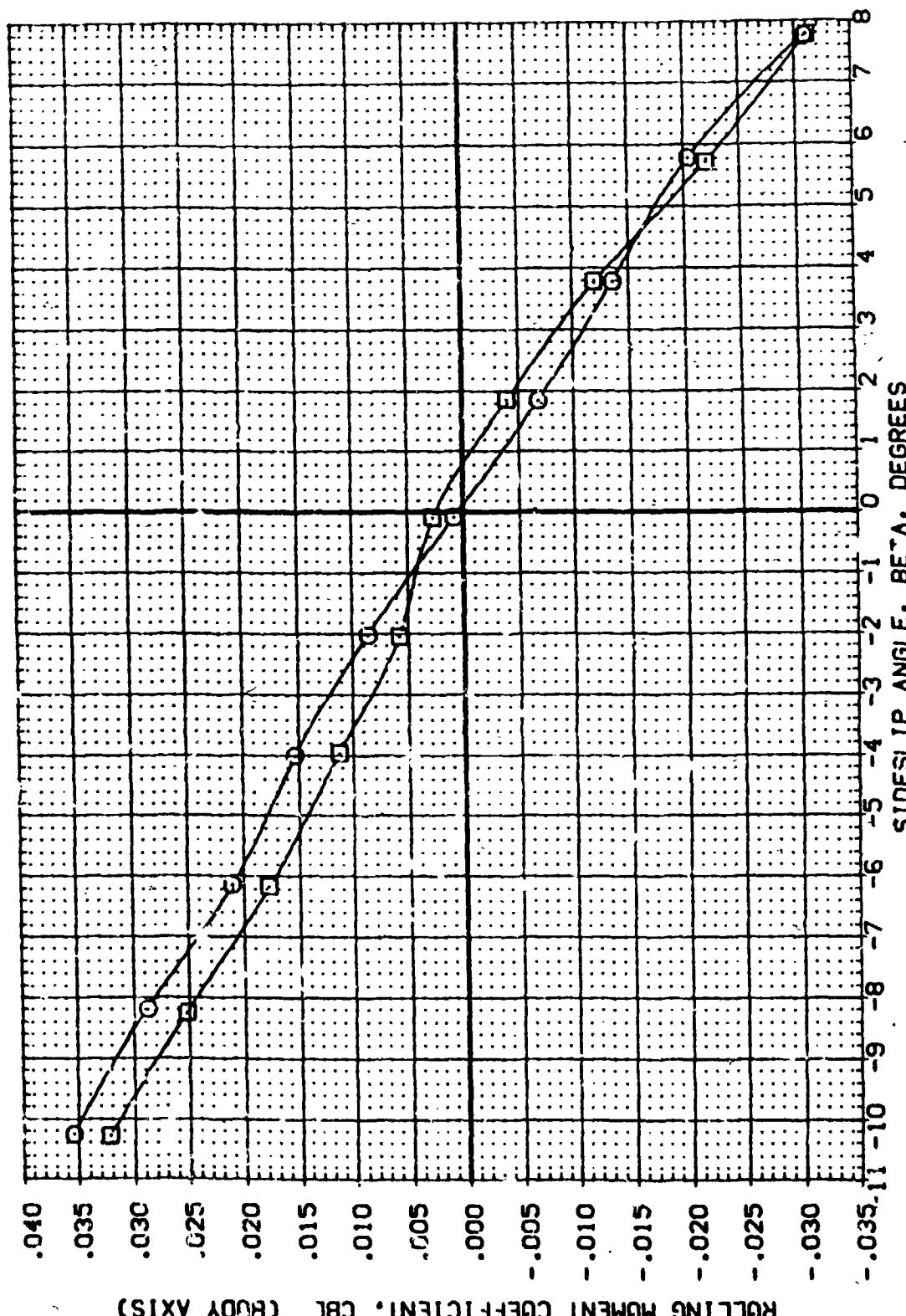


FIG. 12 RUDDER DEFLECTION. POWER ON AND OFF, LATERAL-DIRECTIONAL (YAW).
 (A)MACH = 7.32

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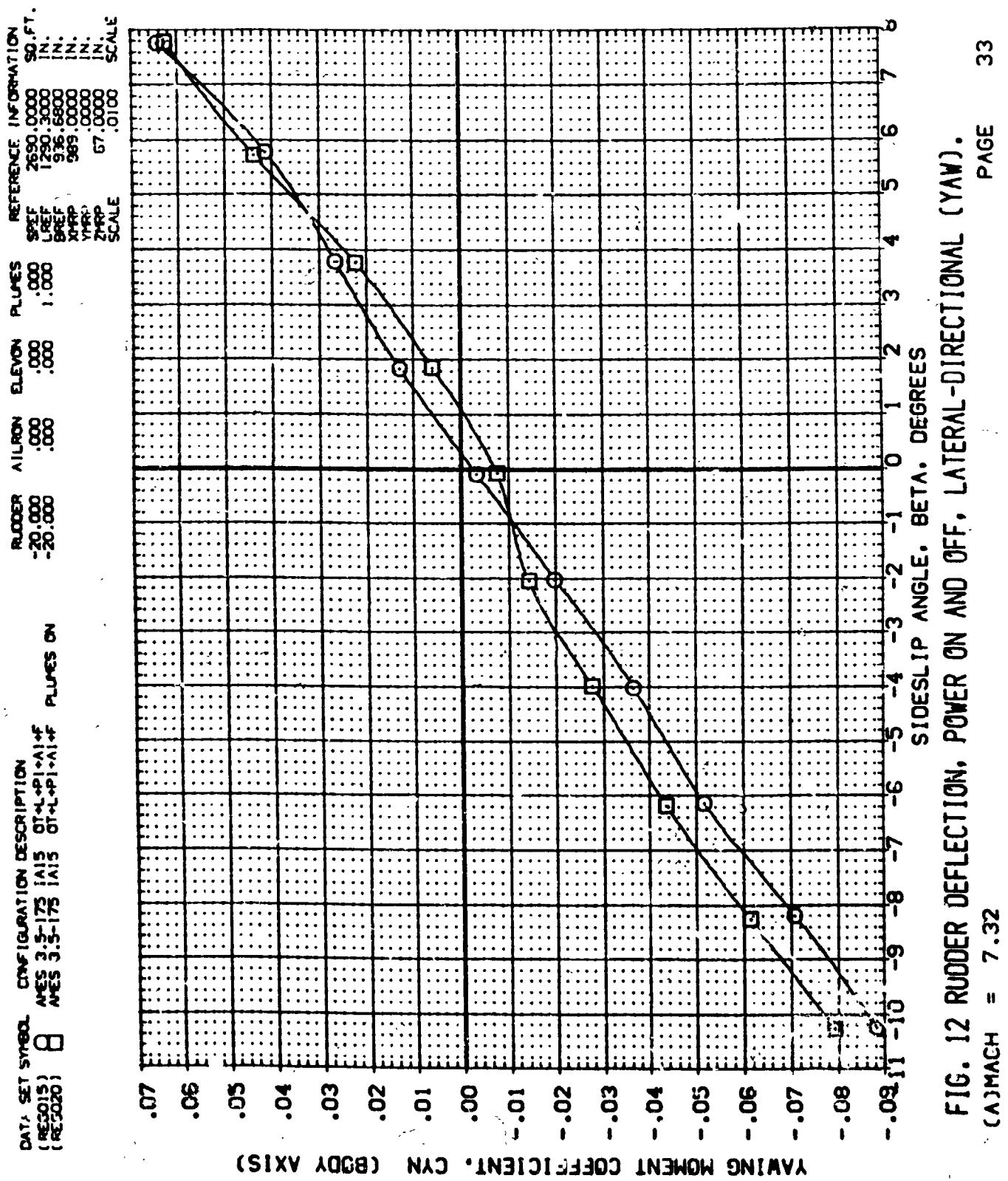


FIG. 12 RUDDER DEFLECTION, POWER ON AND OFF, LATERAL-DIRECTIONAL (YAW).
 (A)MACH = 7.32

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DATA SET SYMBOL: CONFIGURATION DESCRIPTION: MACH 3.5-173 LWS 3.5-173 LWS PLATES ON
 (REF005) 8 (REF020)

RUDDER ALTRON ELEVON PLATES REFERENCE INFORMATION
 -20.000 .000 .000 1.000 SPEC 2690.0000 SQ.FT.
 -20.000 .080 .000 1.000 LINE 1250.3000 IN.
 0.000 .000 1.000 BRFL 935.6800 IN.
 0.000 .000 1.000 XFLP 989.0000 IN.
 0.000 .000 1.000 YFLP 67.0000 IN.
 0.000 .000 1.000 ZFLP .0100 SCALE

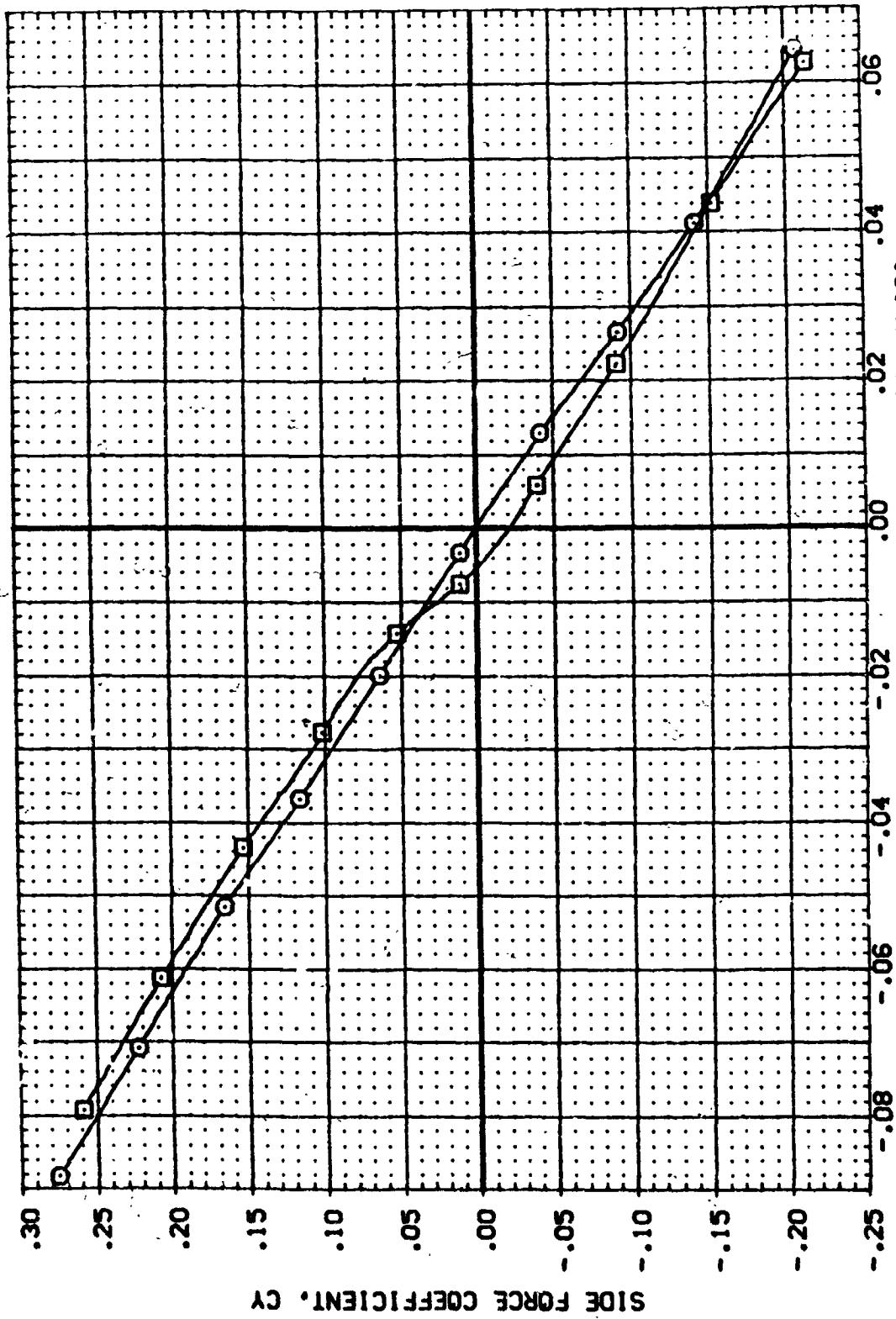


FIG. 12 RUDDER DEFLECTION, POWER ON AND OFF, LATERAL-DIRECTIONAL (YAW).
 (A)MACH = 7.32

DATA SET NUMBER: 13
(RECD 13) ANES 3.5-175 (A) STABILIZER

REFERENCE INFORMATION
SPEED: 2650.0000 SQ.FT.
LIFT COEF: 1.250.0000 IN.
BREFF: 936.8800 IN.
SHRP: 989.0000 IN.
VTRP: 67.0000 IN.
ZTRP: 67.0000 IN.
SCALE: .0100

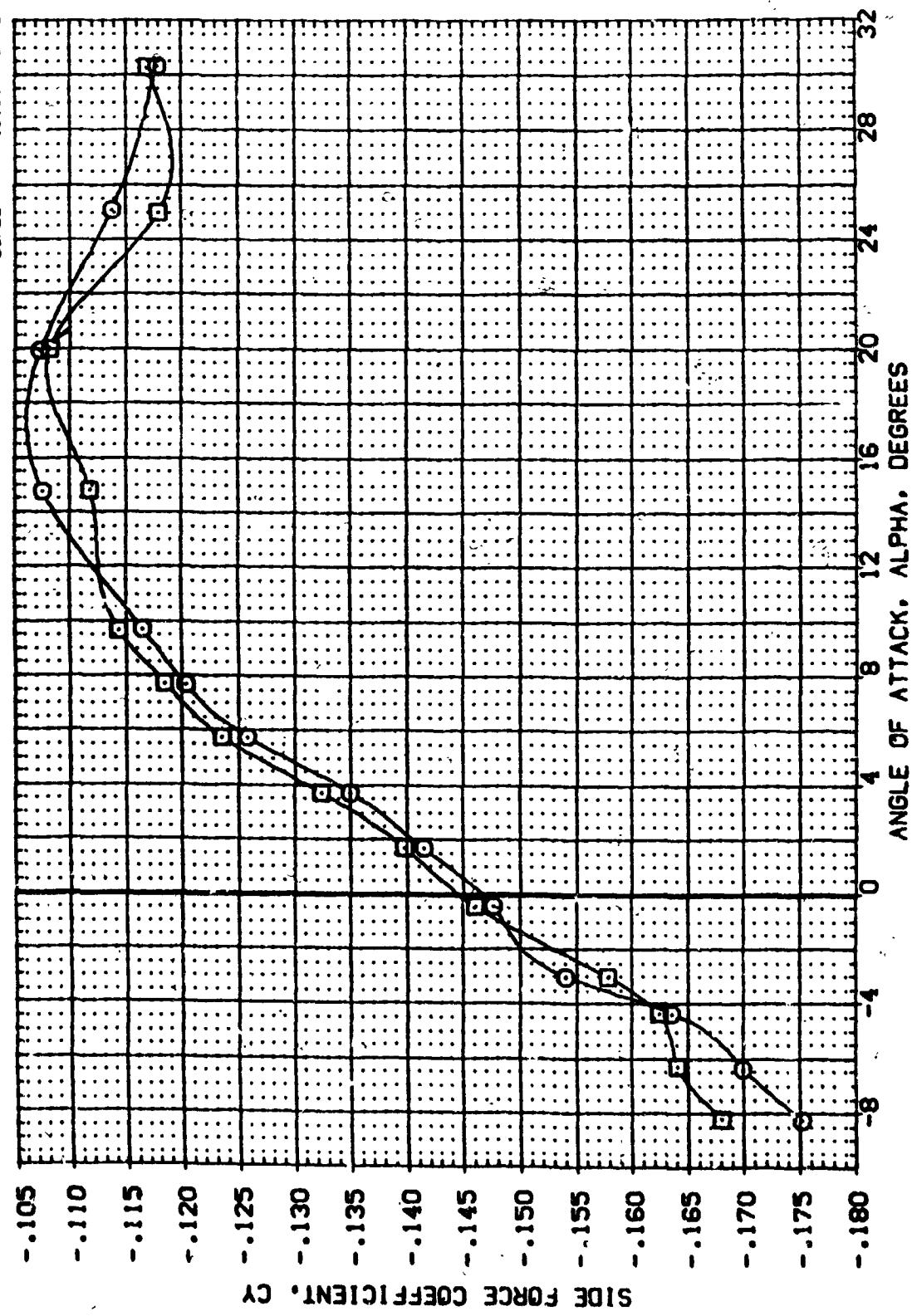


FIG. 13 RUDDER DEFLECTION WITH $\beta = 5$ DEG., LATERAL-DIRECTIONAL (PITCH).
(A)MACH = 7.32

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DATA SET NUMBER: 1 (REG012)
 CONFIGURATION DESCRIPTION:
 ANES 3.5-173 LALS 074-21-A14
 ANES 3.5-173 LALS 074-21-A14

Rudder Aileron Elevon Flaps Reference Information
 -20.000 .000 .000 .000 SPEC 2690.0000 SQ.FT.
 -20.000 .000 .000 .000 LREF 1250.3000 IN.
 0.000 0.000 0.000 0.000 BRF 936.6800 IN.
 0.000 0.000 0.000 0.000 DREF 989.0000 IN.
 0.000 0.000 0.000 0.000 TREF 714P 67.0000 IN.
 0.000 0.000 0.000 0.000 ZREF .0100 SCALE

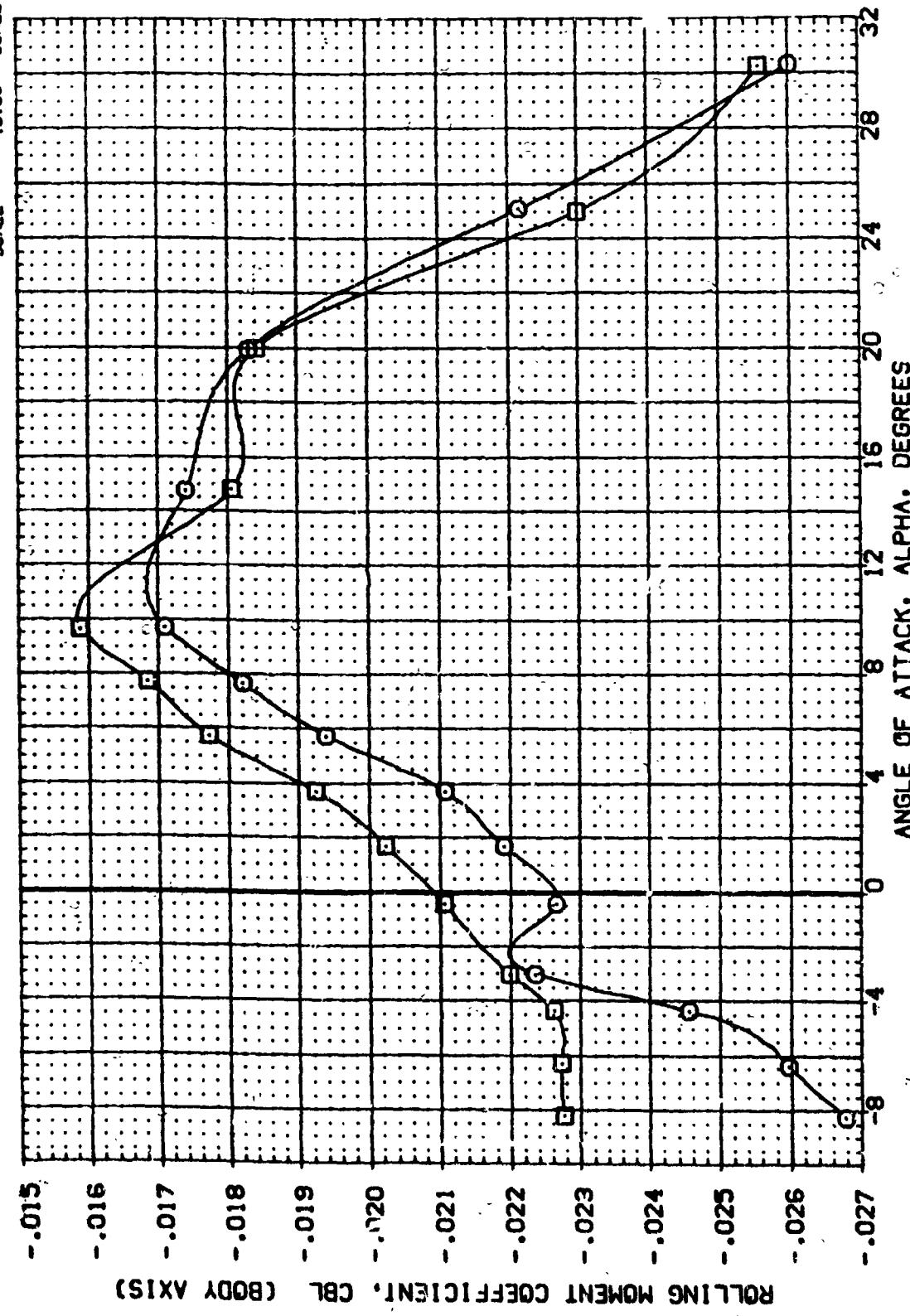


FIG. 13. RUDDER DEFLECTION WITH BETA = 5 DEG., LATERAL-DIRECTIONAL (PILOT).

(A)MACH = 7.32

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DATA SET NAME: CONFIGURATION DESCRIPTION
 (RECD012) C MEG 3-3-1715 DT-APP-A1

| ALUMINUM | FLYING PLATES | PERFORMANCE INFORMATION | SQ. FT. |
|-------------|---------------|-------------------------|---------|
| 10-20,000 | 000,000 | 2690,0000 | 120,000 |
| 20-30,000 | 000,000 | 2500,0000 | 120,000 |
| 30-40,000 | 000,000 | 2300,0000 | 120,000 |
| 40-50,000 | 000,000 | 2100,0000 | 120,000 |
| 50-60,000 | 000,000 | 1900,0000 | 120,000 |
| 60-70,000 | 000,000 | 1700,0000 | 120,000 |
| 70-80,000 | 000,000 | 1500,0000 | 120,000 |
| 80-90,000 | 000,000 | 1300,0000 | 120,000 |
| 90-100,000 | 000,000 | 1100,0000 | 120,000 |
| 100-110,000 | 000,000 | 900,0000 | 120,000 |
| 110-120,000 | 000,000 | 700,0000 | 120,000 |
| 120-130,000 | 000,000 | 500,0000 | 120,000 |
| 130-140,000 | 000,000 | 300,0000 | 120,000 |
| 140-150,000 | 000,000 | 100,0000 | 120,000 |

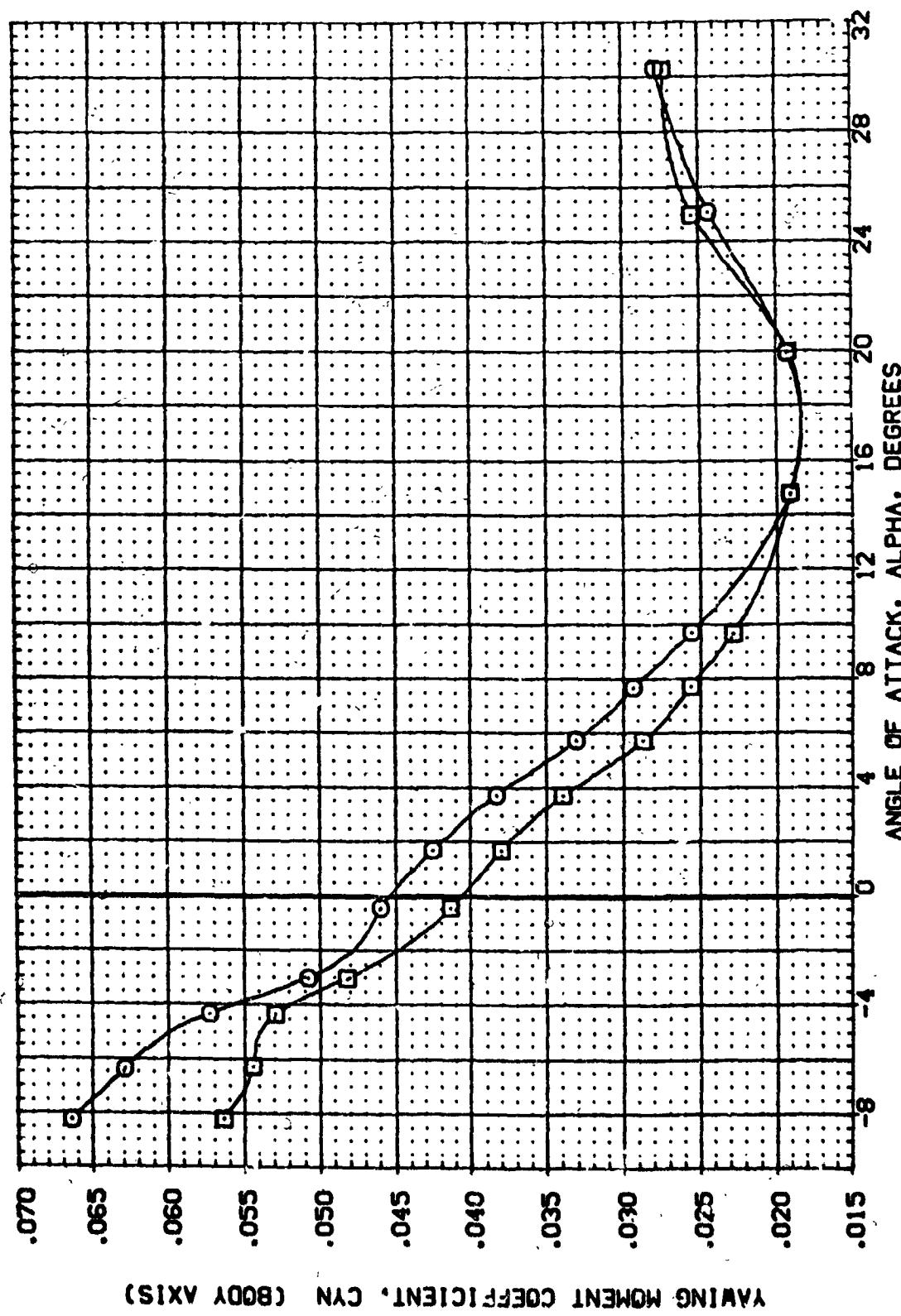


FIG. 13 RUDDER DEFLECTION WITH $\text{BET}_W = 5 \text{ DEG.}$: LATERAL-DIRECTIONAL (PITCH).

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DATA SET SYMBOL CONFIGURATION DESCRIPTION
(REF016) AMES 3.5-175 1A15 074-#1-AI
(REF017) AMES 3.5-175 1A15 074-#1-AI

REFERENCE INFORMATION
SREF 2690.0000 SQ.FT.
LREF 1250.3000 IN.
BREF 935.6800 IN.
XHPP 989.0000 IN.
YHPP 67.0000 IN.
ZHPP .0000 IN.
SCALE .0100

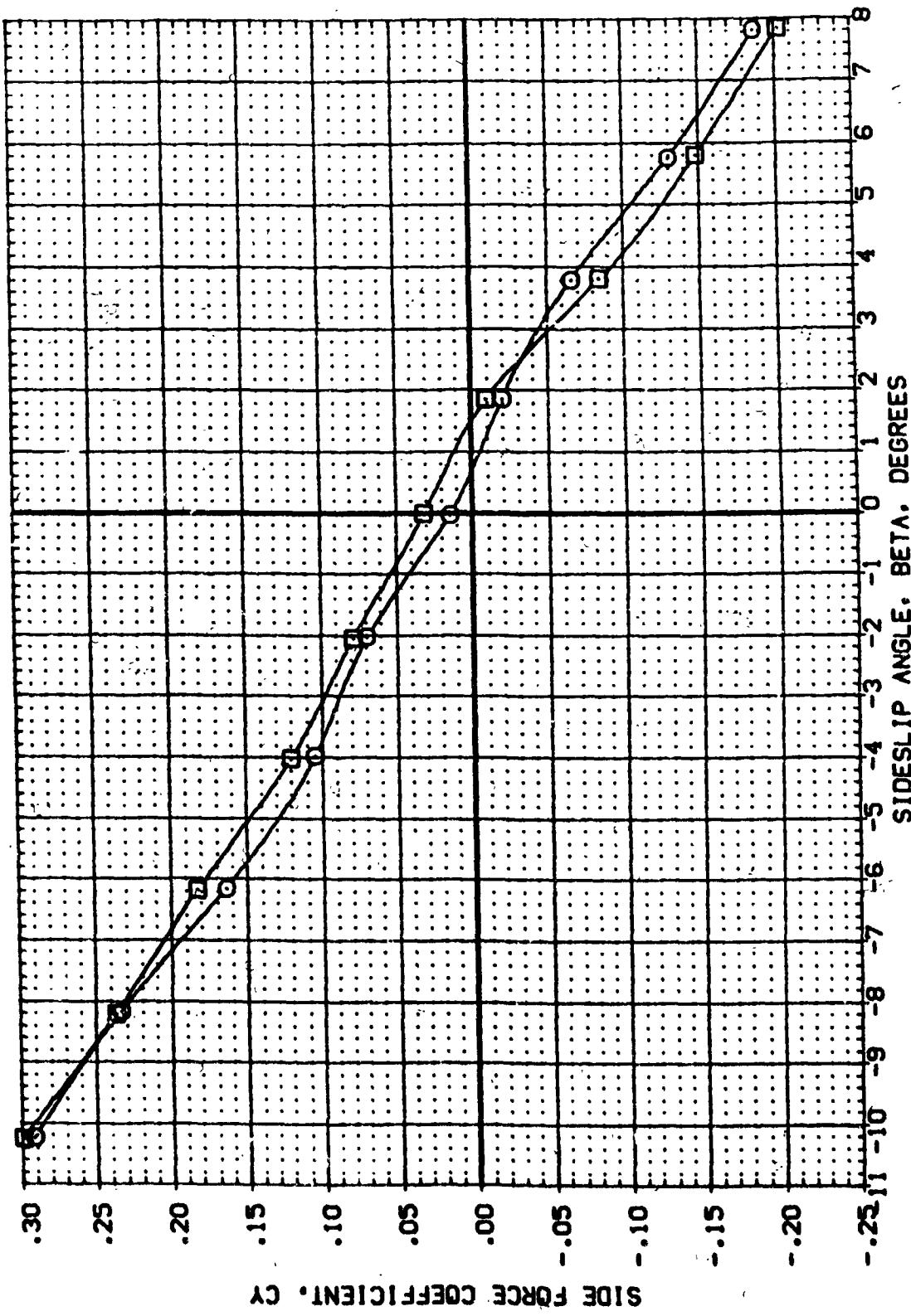


FIG. 14 YAW EFFECTS WITH ALPHA = 0 AND 30 DEG., WITHOUT FAIRING, LAT.-DIR.

C_AMACH = 7.32

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DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REC016) ARES 3.5-175 IAIIS GTR-L+P1+A1
 (REC017) ARES 3.5-175 IAIIS GTR-L+P1+A1

ALPHA AIRSON ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 .000 2690.0000 SQ.FT.
 .000 .000 .000 1290.3000 IN.
 .000 .000 .000 936.6800 IN.
 .000 .000 .000 989.0000 IN.
 .000 .000 .000 67.0000 IN.
 .0100 SCALE

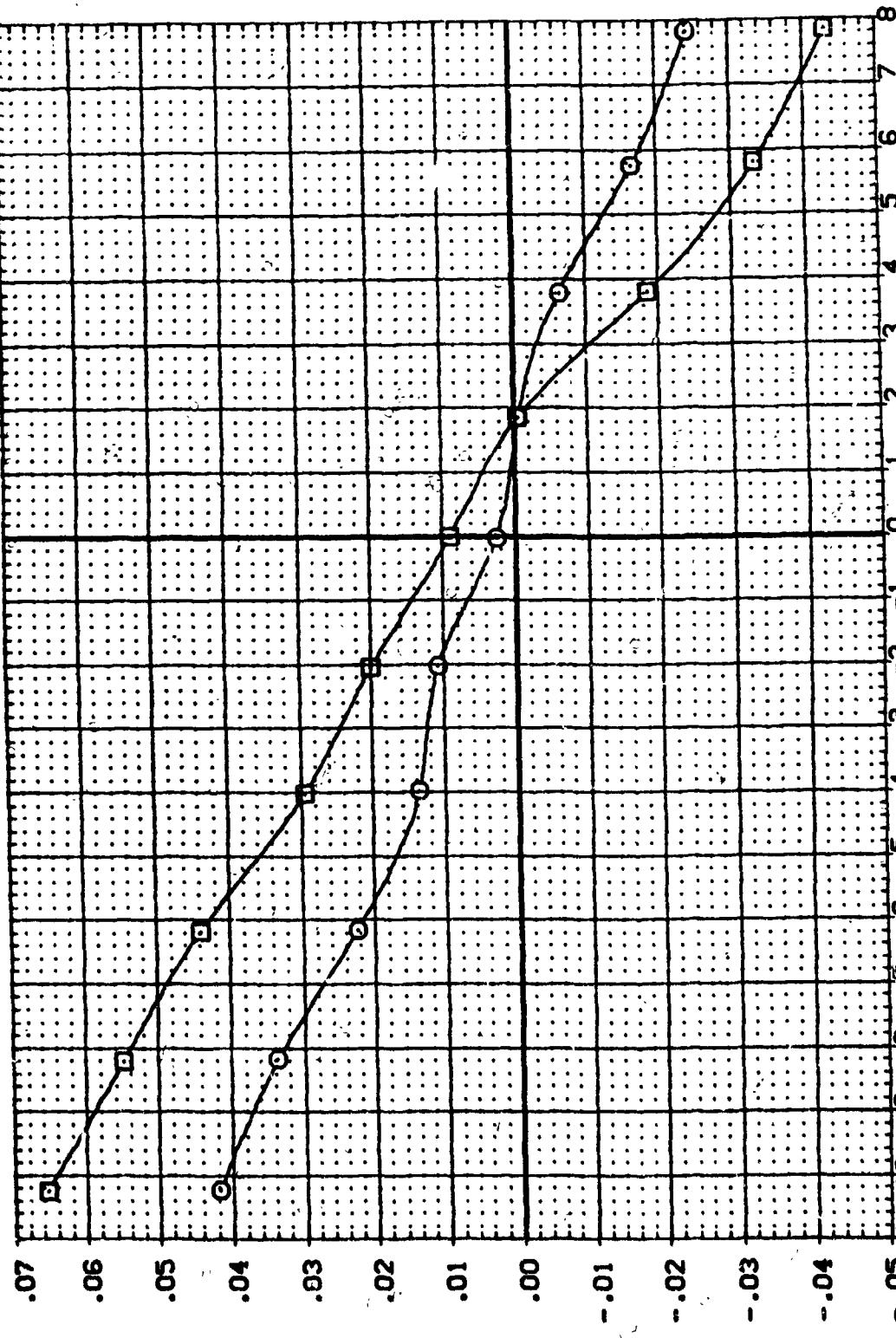


FIG. 1.1 YAW EFFECTS WITH ALPHA = 0 AND 30 DEG., WITHOUT FAIRING, LAT.-DIR.
 CA MACH = 7.32
 PAGE 39

DATA SET NAME: CONFIGURATION DESCRIPTION
 (REF3016) 8 Ames 3.5-175 TAIS OTW-
 (REF3017) 8 Ames 3.5-175 TAIS OTW-

REFERENCE INFORMATION
 SREF 2690.0000 SQ.FT.
 LREF 1250.3000 IN.
 BREF 936.6800 IN.
 XREF 989.0000 IN.
 YREF 1000.0000 IN.
 ZREF 67.0000 IN.
 SCALE 0.0100

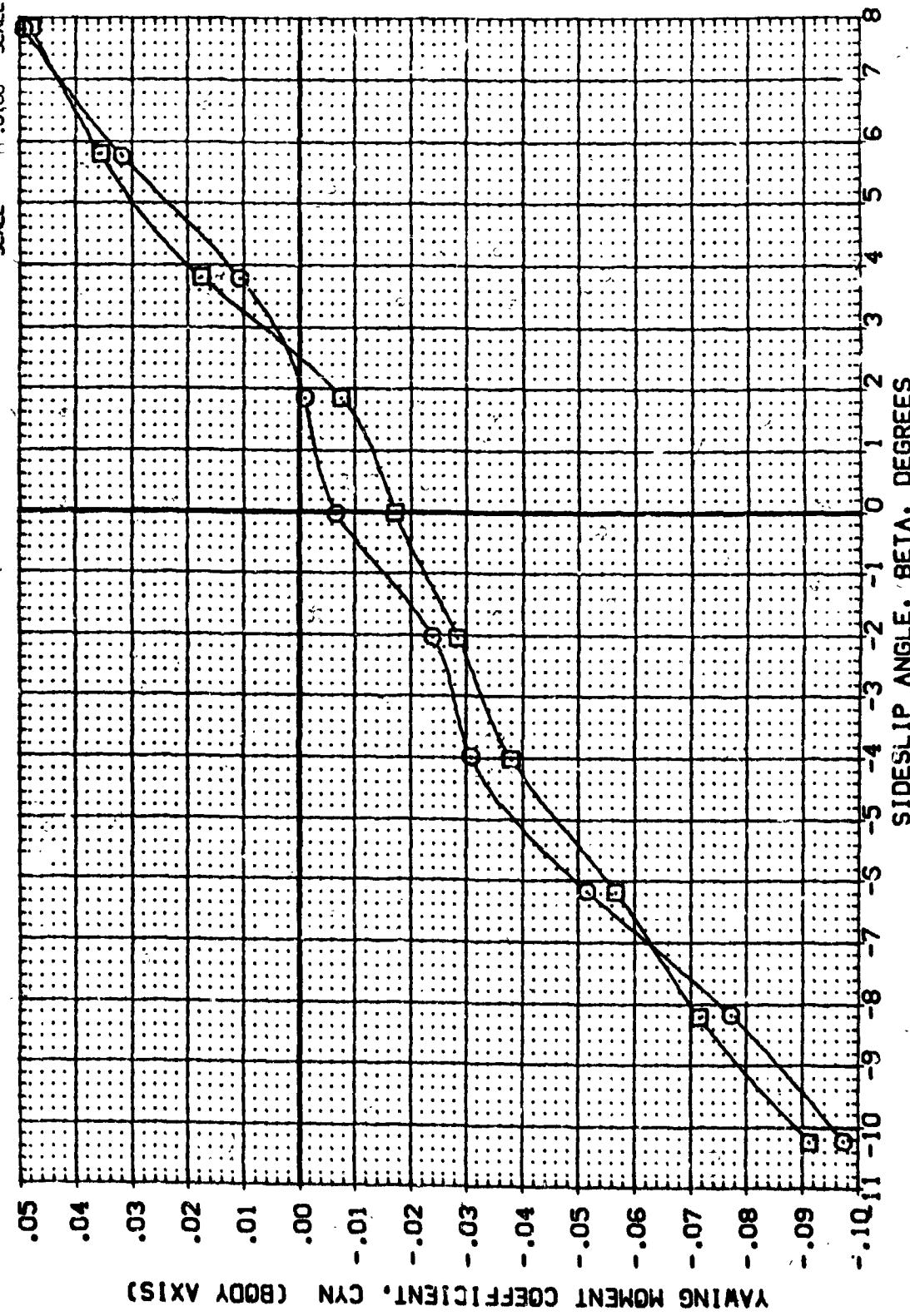


FIG. 14 YAW EFFECTS WITH ALPHA = 0 AND 30 DEG., WITHOUT FAIRING, LAT.-DIR.
 $(\text{AJMACH} = 7.32)$

DATA SET SYMBOL. CONFIGURATION DESCRIPTION
 (REGD016) AMES 3.5-175 IAI5 OT-LP1-AI
 (REGD017) AMES 3.5-175 IAI5 OT-LP1-AI

ALPHA AILERON ELEVON PLATES REFERENCE INFORMATION
 .000 :000 :000 SREF 2690.0000 SQ.FT.
 30.000 :000 :000 LREF 1250.3000 IN.
 BREF 936.6800 IN.
 XRP 986.0000 IN.
 YRP 67.0000 IN.
 ZRP .0100 SCALE

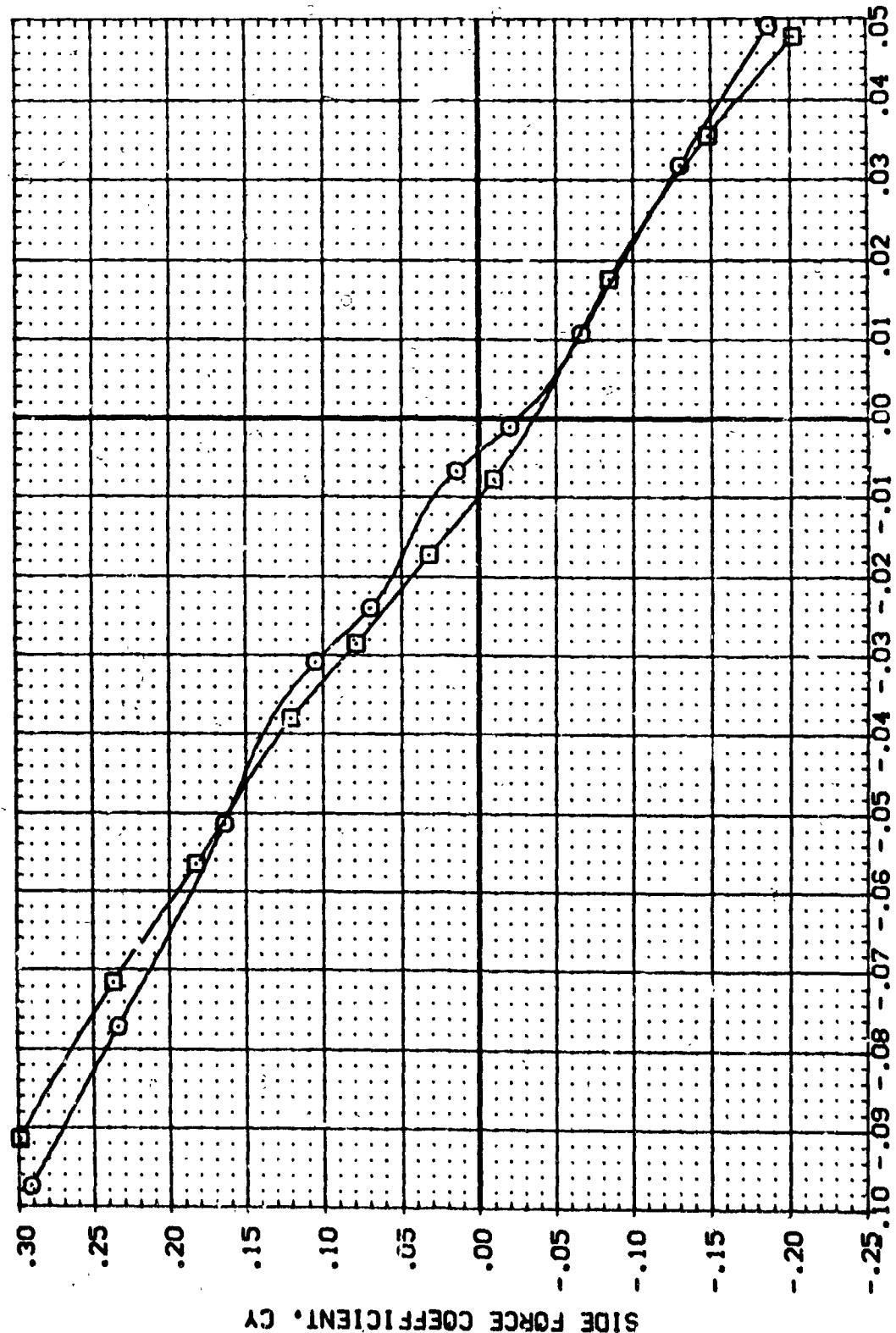


FIG. 14 YAW EFFECTS WITH ALPHA = 0 AND 30 DEG., WITHOUT FAIRING, LAT.-DIR.
 (A)MACH = 7.32

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DATA SET SUMMARY CONFIGURATION DESCRIPTION
 REGO21 ASES 3.5-175 LWS 074-491-A14 PLUMES ON
 REGO24 ASES 3.5-175 LWS 074-491-A14 PLUMES ON

RUDDER AILRDN ELEVON PLUMES REFERENCE INFORMATION
 .000 .000 .000 1.000 SREF 2690.0000 SQ.FT.
 .000 .000 1.000 LREF 1280.2000 IN.
 BREF 936.6000 IN.
 XMRP 989.0000 IN.
 YMRP 67.0000 IN.
 ZMRP .0100 SCALE

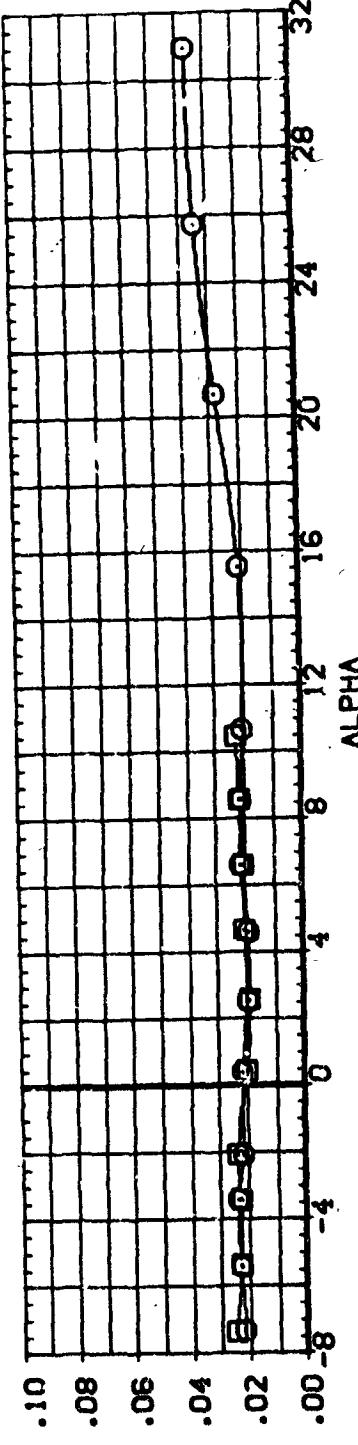
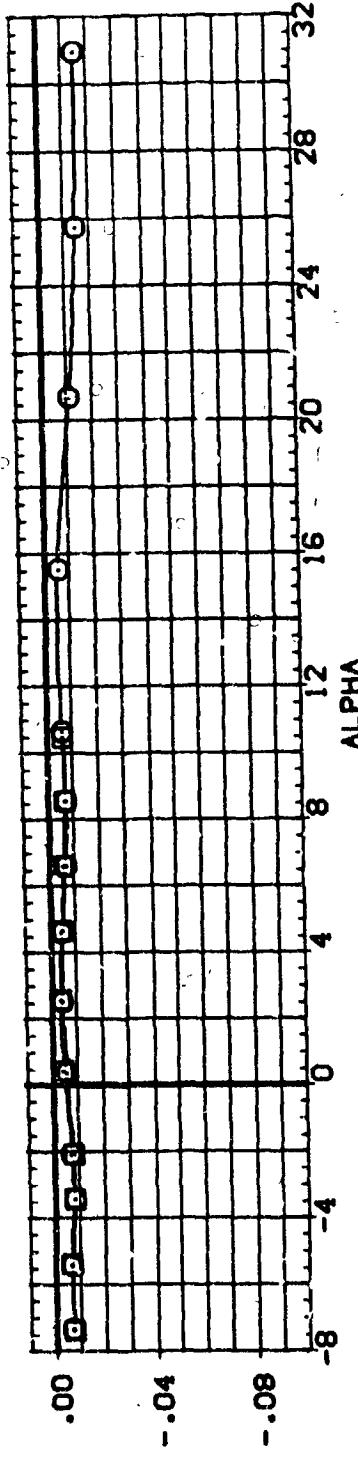
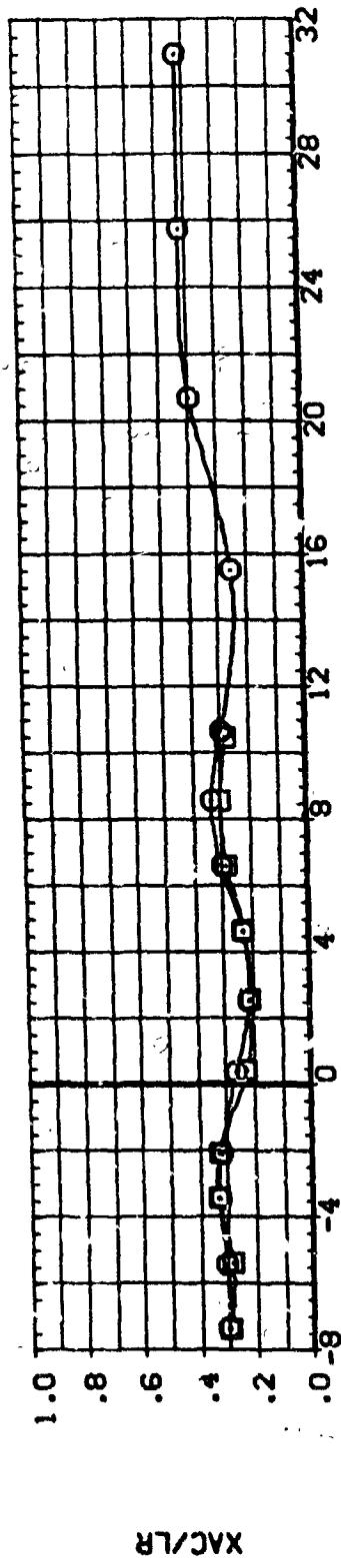


FIG. 15 SUMMARY OF SOLID PLUME PITCH EFFECTS.
 (MACH = 7.32)

DATA SET NUMBER: 00000000000000000000000000000000
 CONFIGURATION DESCRIPTION: ANES 3.5-175 TAIS OT-LP-AI-AF PLUMES ON
 (REFC014) (REFC019)

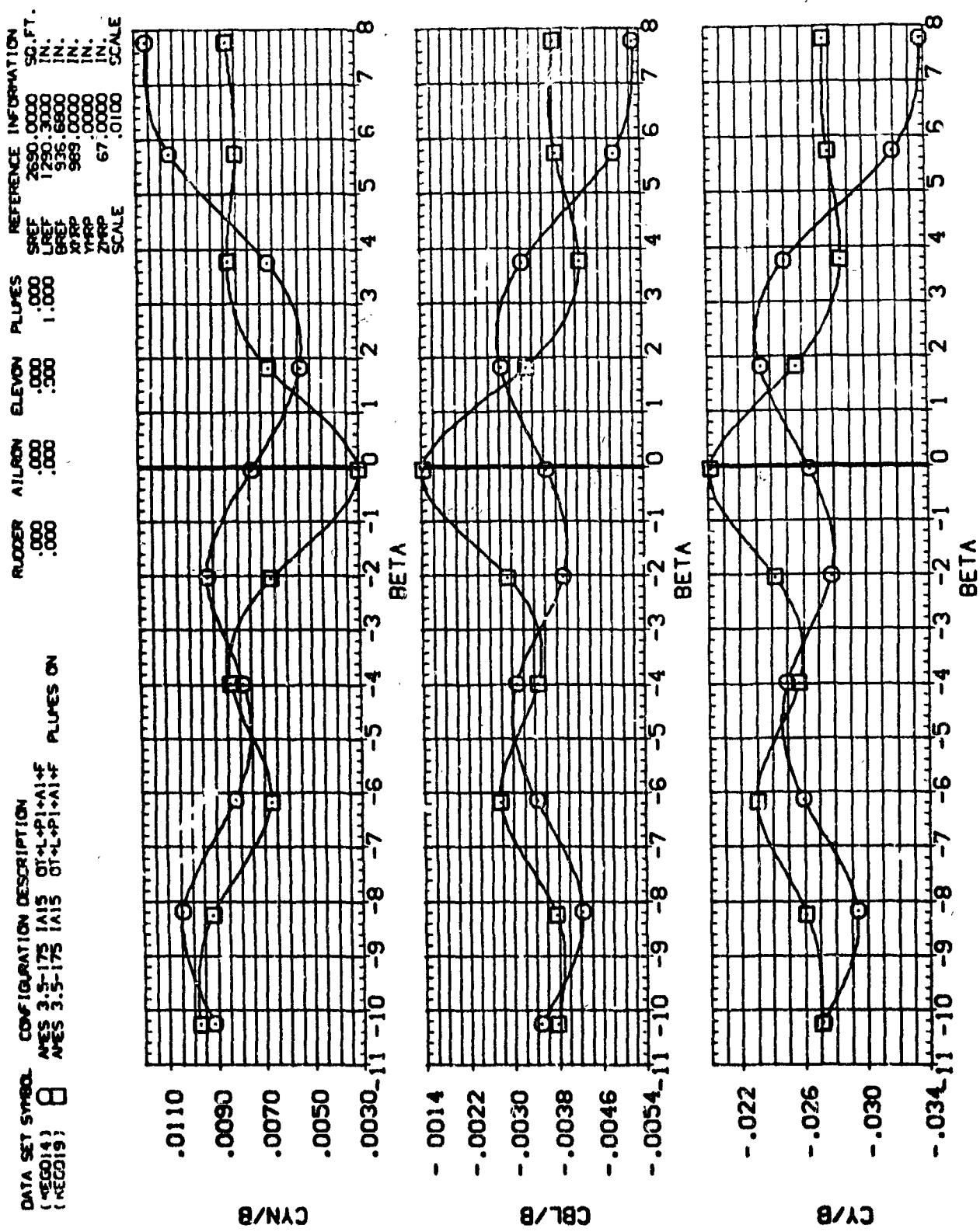


FIG. 16 SUMMARY OF SOLID PLUME YAW EFFECTS.
 (λ) MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [REG014] 8 AGES 35-75 [AIS OTG+P1+ALF
 [REG015] 9 AGES 35-75 [AIS OTG+P1+ALF PLATES ON

| NUMBER | AIRCRAFT | ELEVON | PLUNES | REFERENCE INFORMATION | | |
|--------|----------|--------|--------|-----------------------|------------|---------|
| | | | | SREF | 2690 .0000 | SO. FT. |
| .000 | .000 | .000 | .000 | LREF | 1250 .3000 | IN. |
| .000 | .000 | .000 | .000 | GREF | 936 .6800 | IN. |
| .000 | .000 | .000 | .000 | XHPP | 369 .0000 | IN. |
| .000 | .000 | .000 | .000 | ZHPP | .0000 | SCRE |

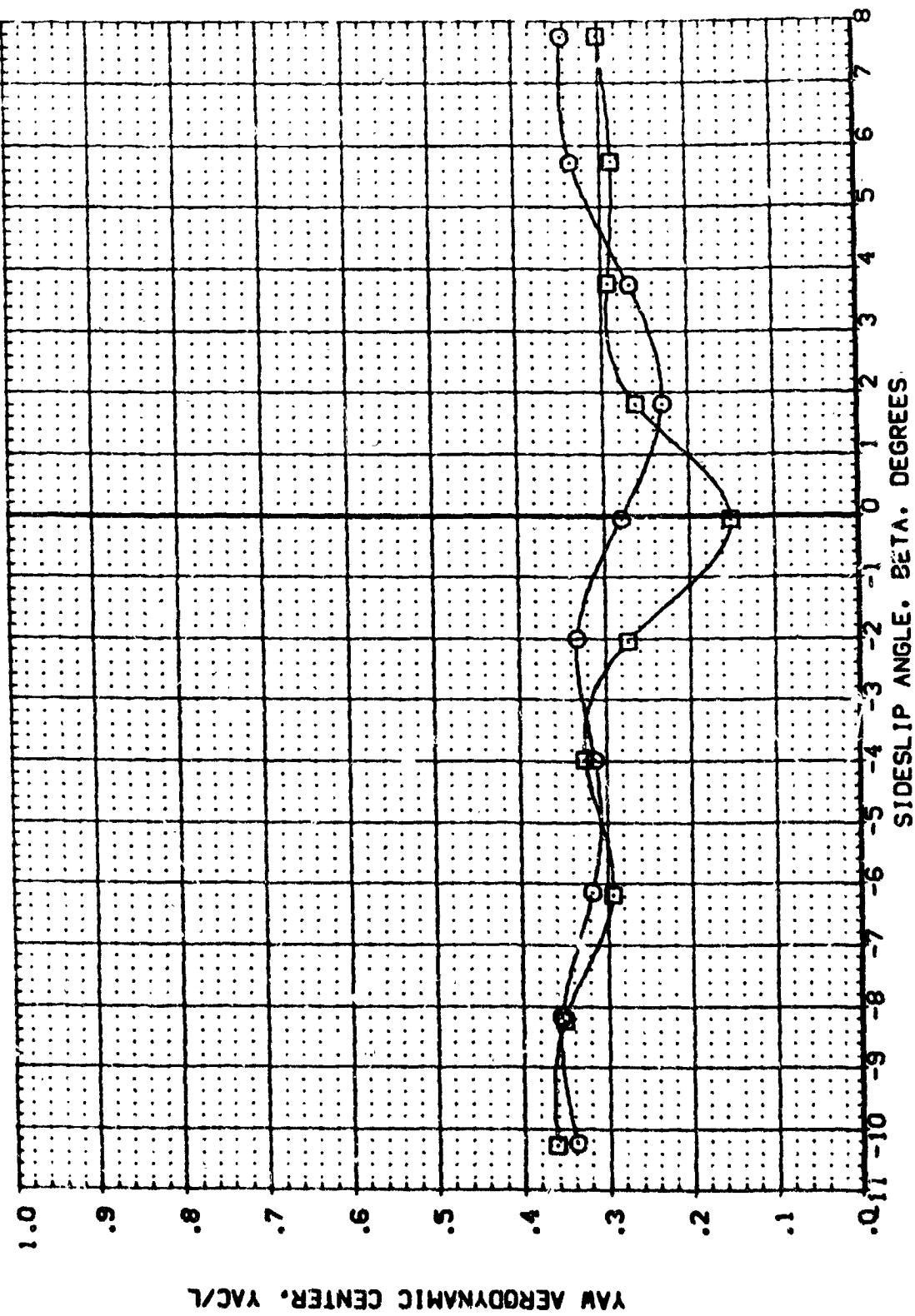


FIG. 16 SUMMARY OF SOLID PLUME YAW EFFECTS.
 $(\Delta)_{MACH} = 7.32$

$$(\text{A})\text{MACH} = 7.32$$

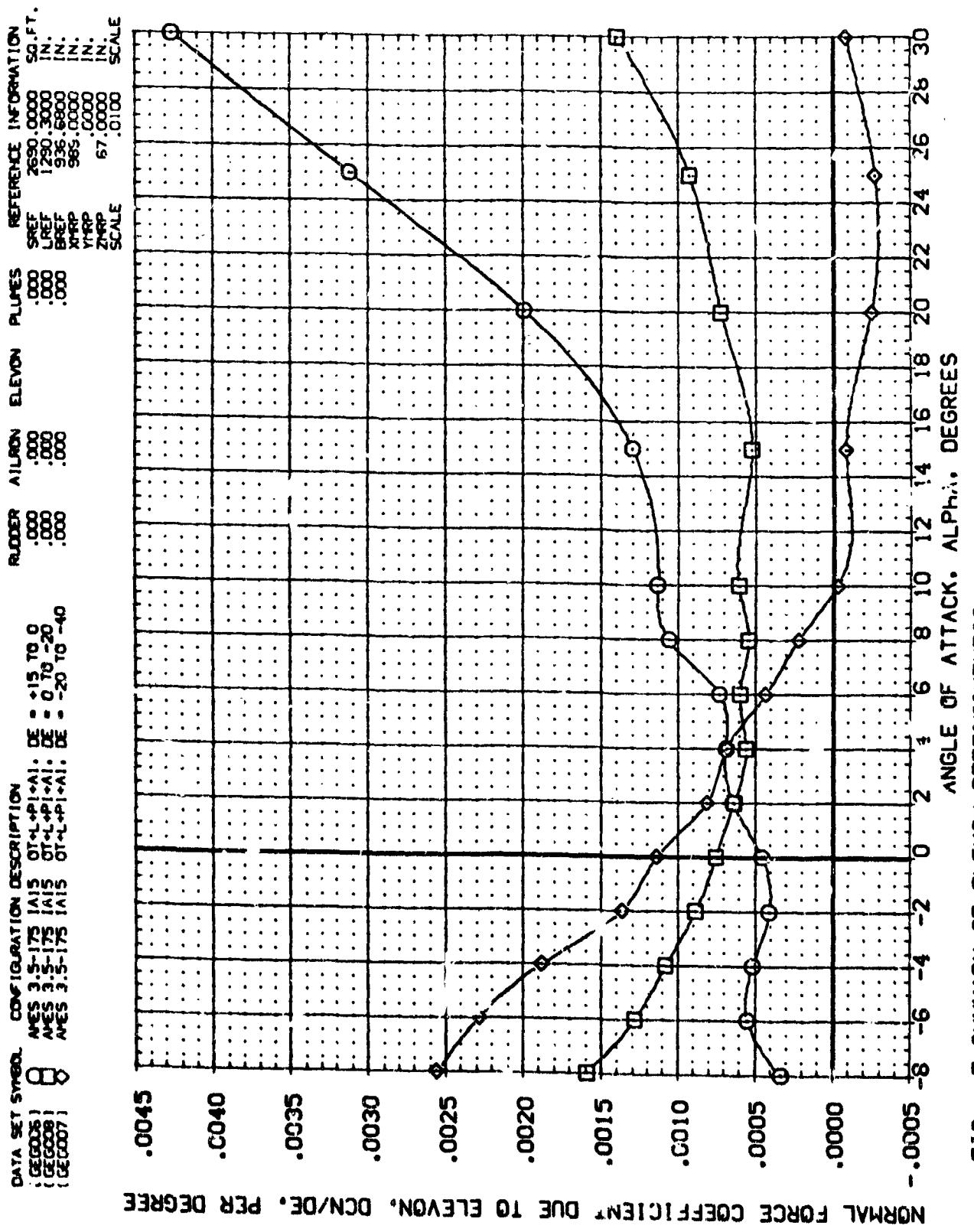


FIG. 16.17 SUMMARY OF ELEVON EFFECTIVENESS.

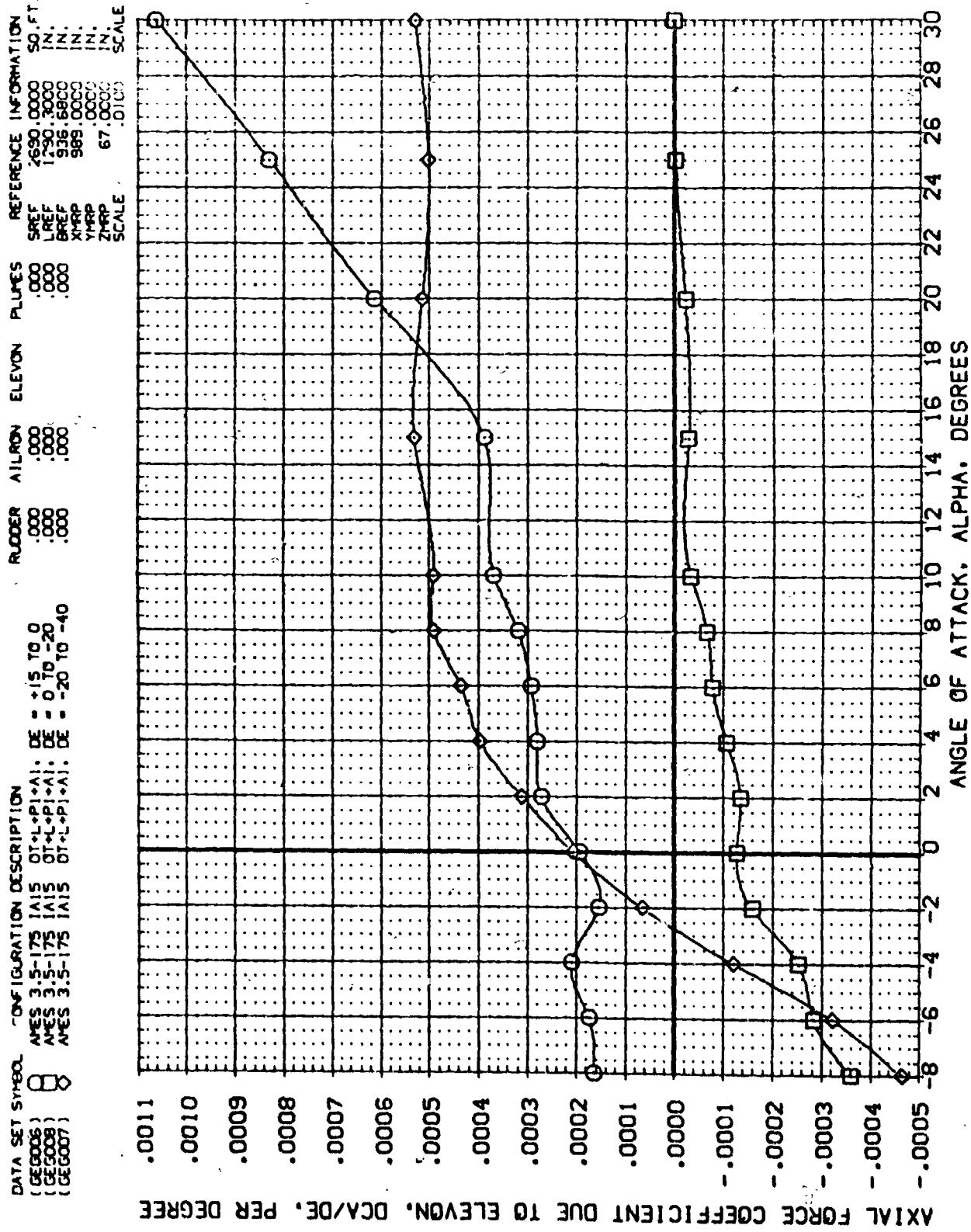


FIG. 17 SUMMARY OF ELEVON EFFECTIVENESS.
(A)MACH = 7.32

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C-2

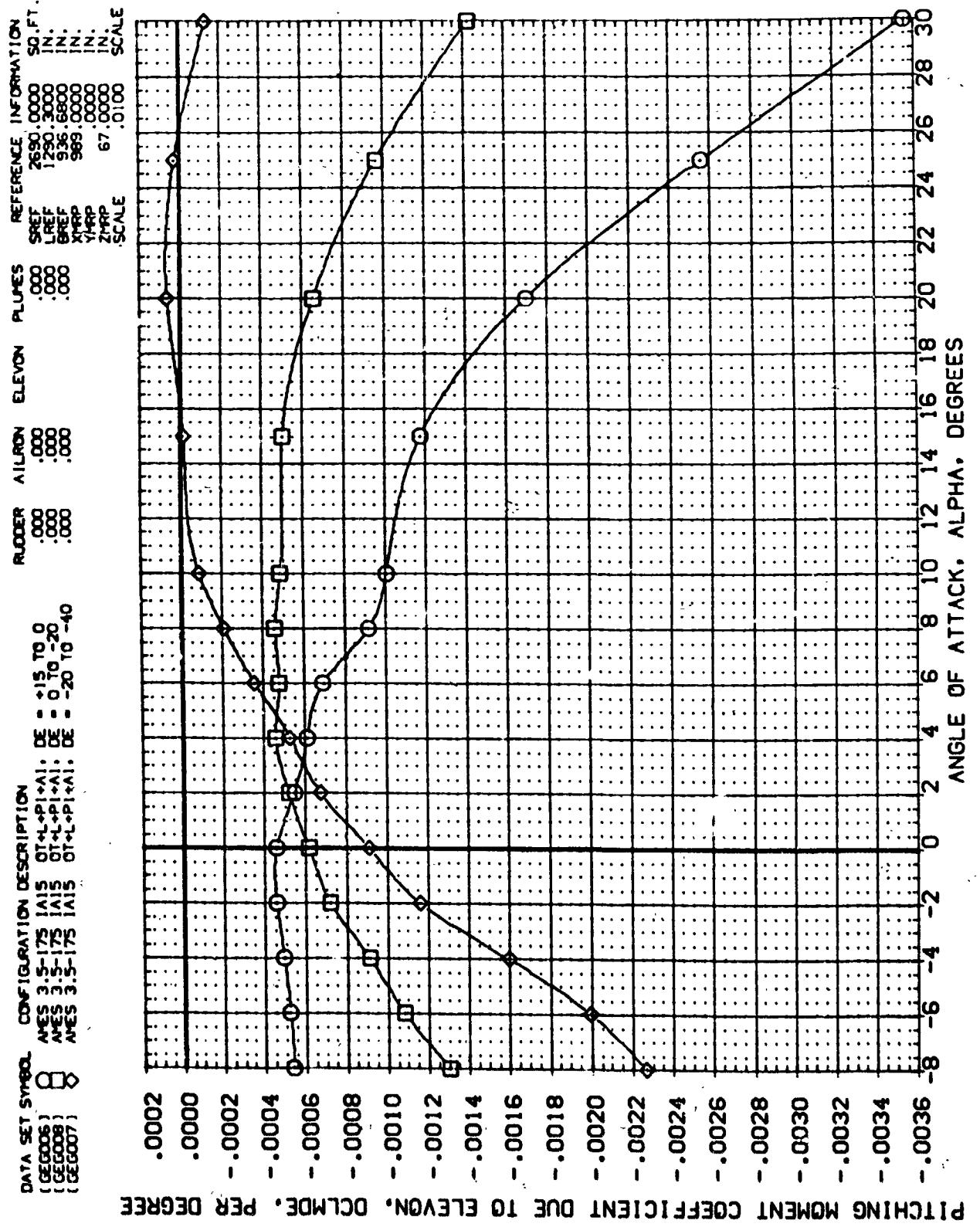


FIG. 17 SUMMARY OF ELEVON EFFECTIVENESS.

(A)MACH = 7.32

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (REFG009) O ANES 3.5-175 IANS DT+L+P1+A1

RUDDER .000 AIRON .000 ELEVON .000 PLUNES .000 REF 2630.0000 SQ.FT.
 LREF 1290.3000 IN.
 BREF 936.6800 IN.
 XMRP 589.0000 IN.
 YMRP 67.0000 IN.
 ZMRP .0100 SCALE

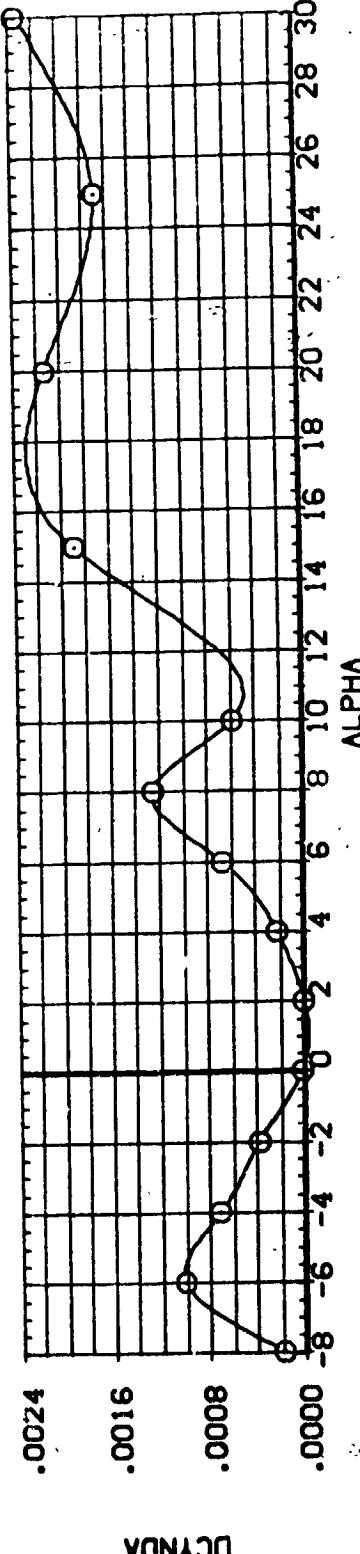
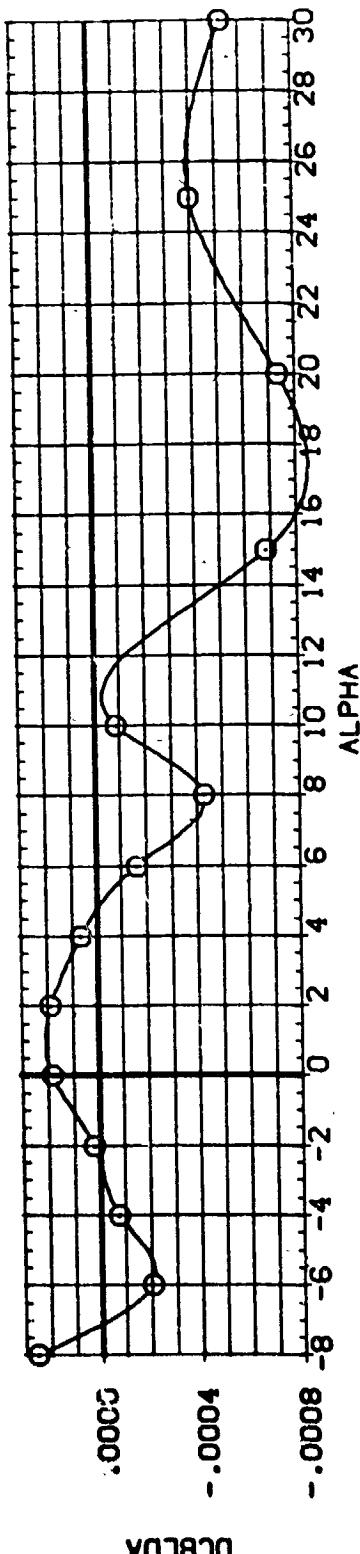
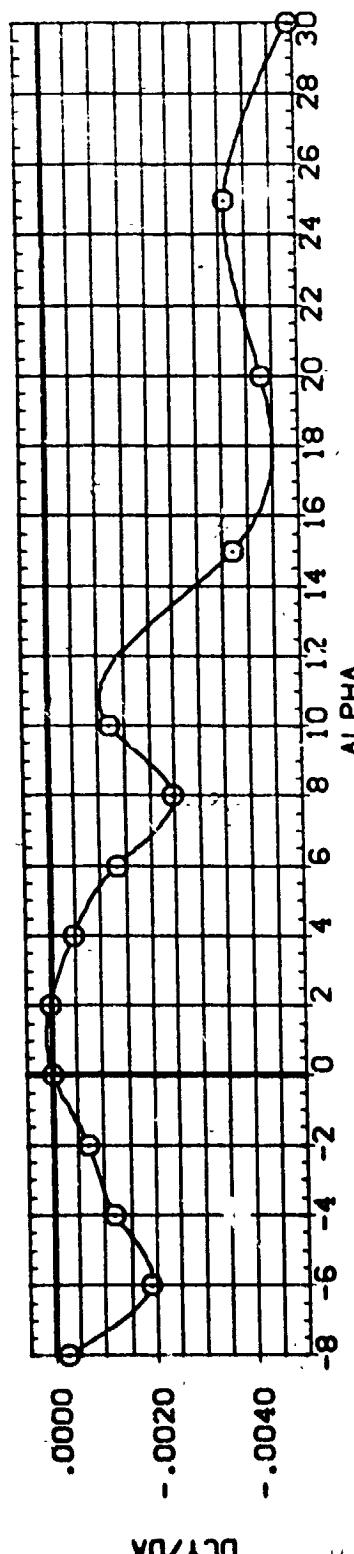


FIG. 18 SUMMARY OF AILERON EFFECTIVENESS.
 (A)MACH = 7.32

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DATA SET SYMBOL: CONFIGURATION DESCRIPTION
 (EEG015) O ANES 3.5-175 TA15 OT-LAP1+AL4F

RUDDER ALTRON ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 .000 SREF 2690.0000 50. FT.
 LREF 1290.3000 IN.
 BREF 935.6800 IN.
 XMRP 989.0000 IN.
 YMRP 67.0000 IN.
 ZMRP .0100 IN.
 SCALE .0100

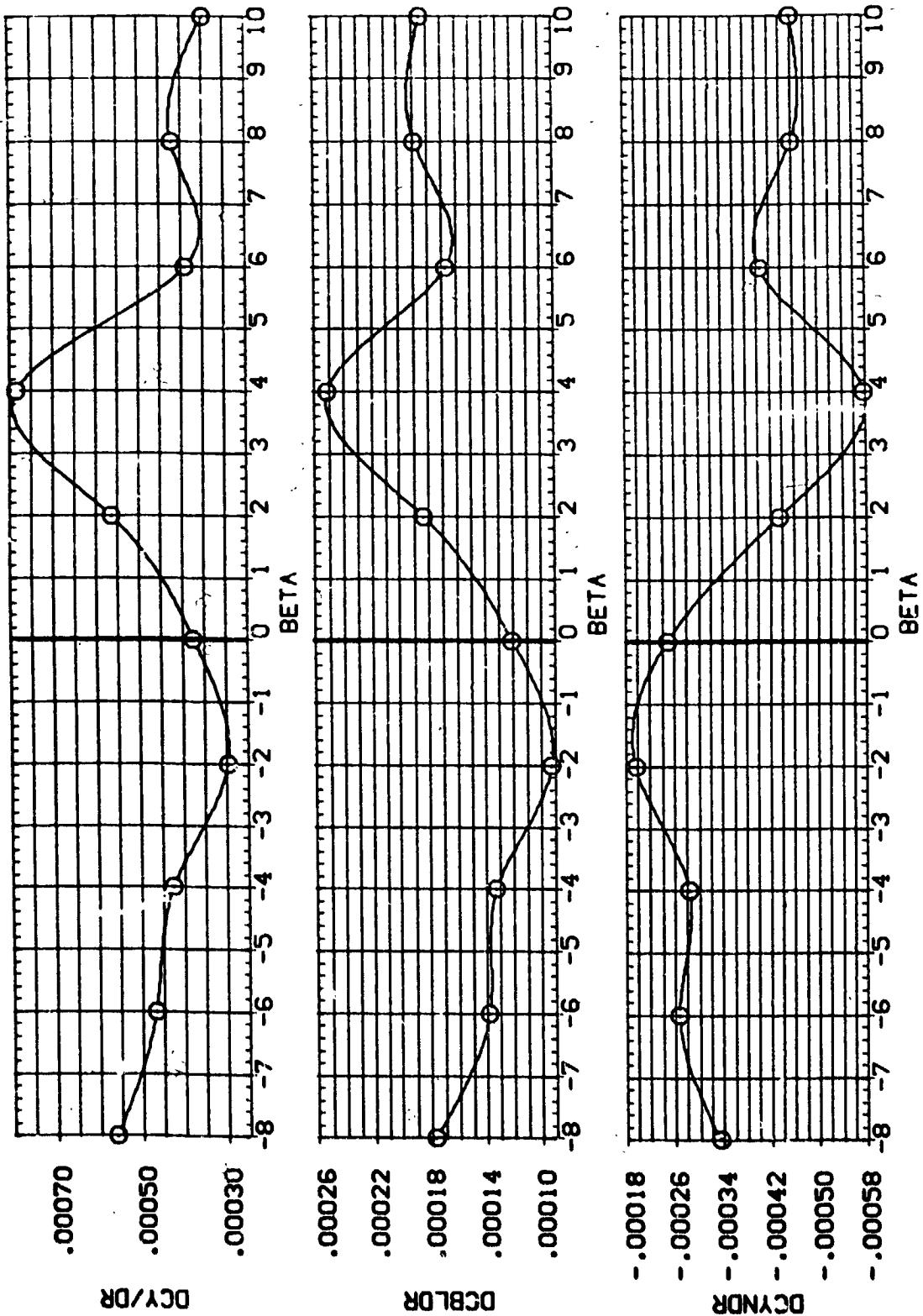


FIG. 19 SUMMARY OF RUDDER EFFECTIVENESS IN YAW.
 (A) MACH = 7.32

DATA SET SMCB. CONFIGURATION DESCRIPTION
 (REG013) O MACH 3.5-175 LADS DBT-AI+AI-REF

RUDDER AILERON ELEVON PLUNES REFERENCE INFORMATION
 .000 .000 .000 .000 SREF 2690.0000 SO.FT.
 LREF 1.290 3000 IN.
 BREF 9.36 1890 IN.
 XMRP 989.0000 IN.
 YMRP 0000 IN.
 ZMRP 67.0000 IN.
 SCALE .0100

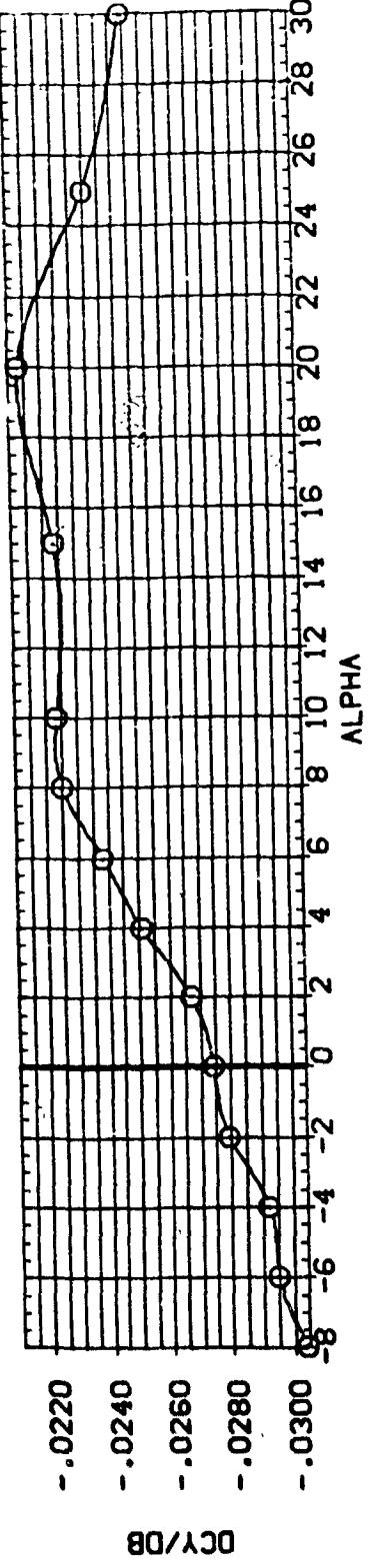
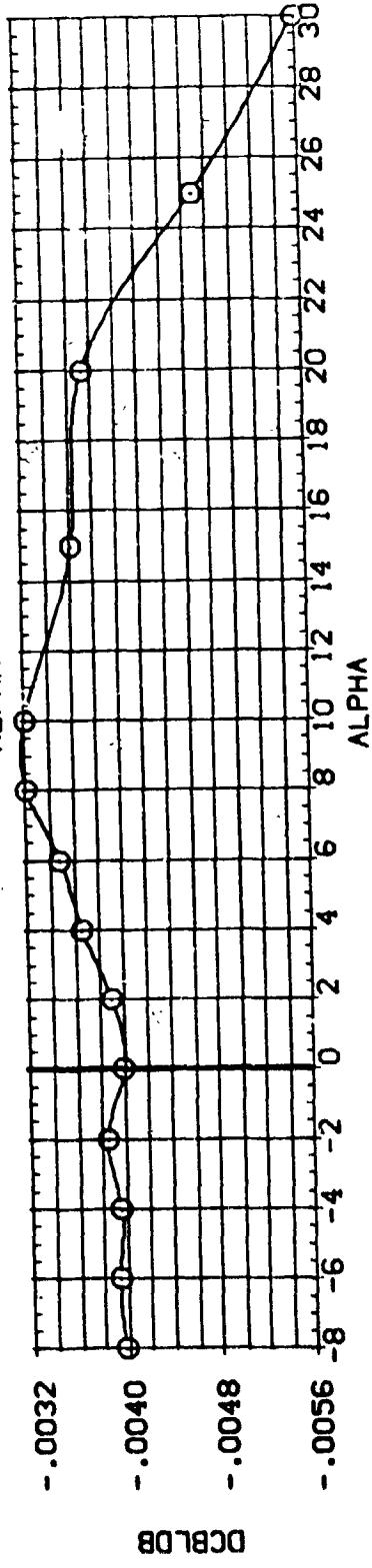
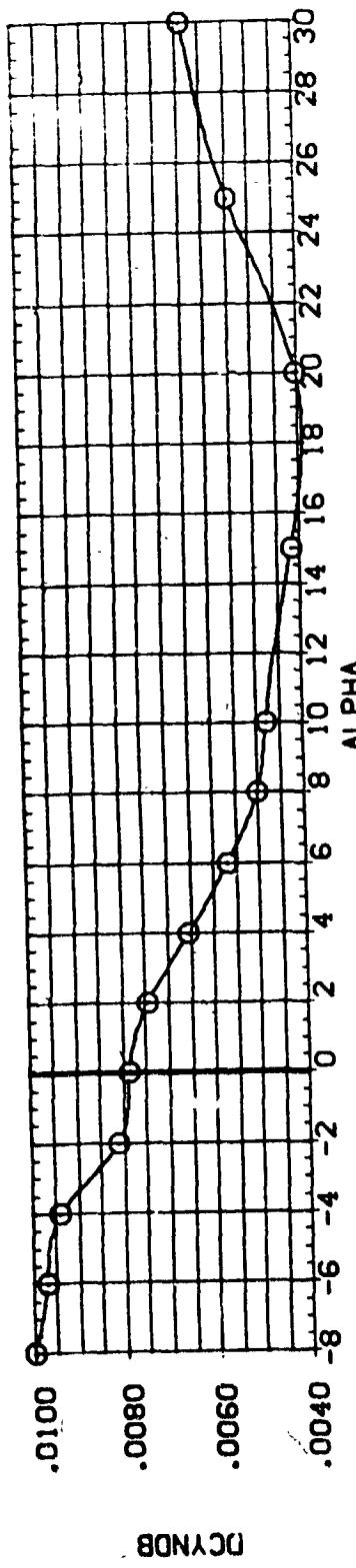
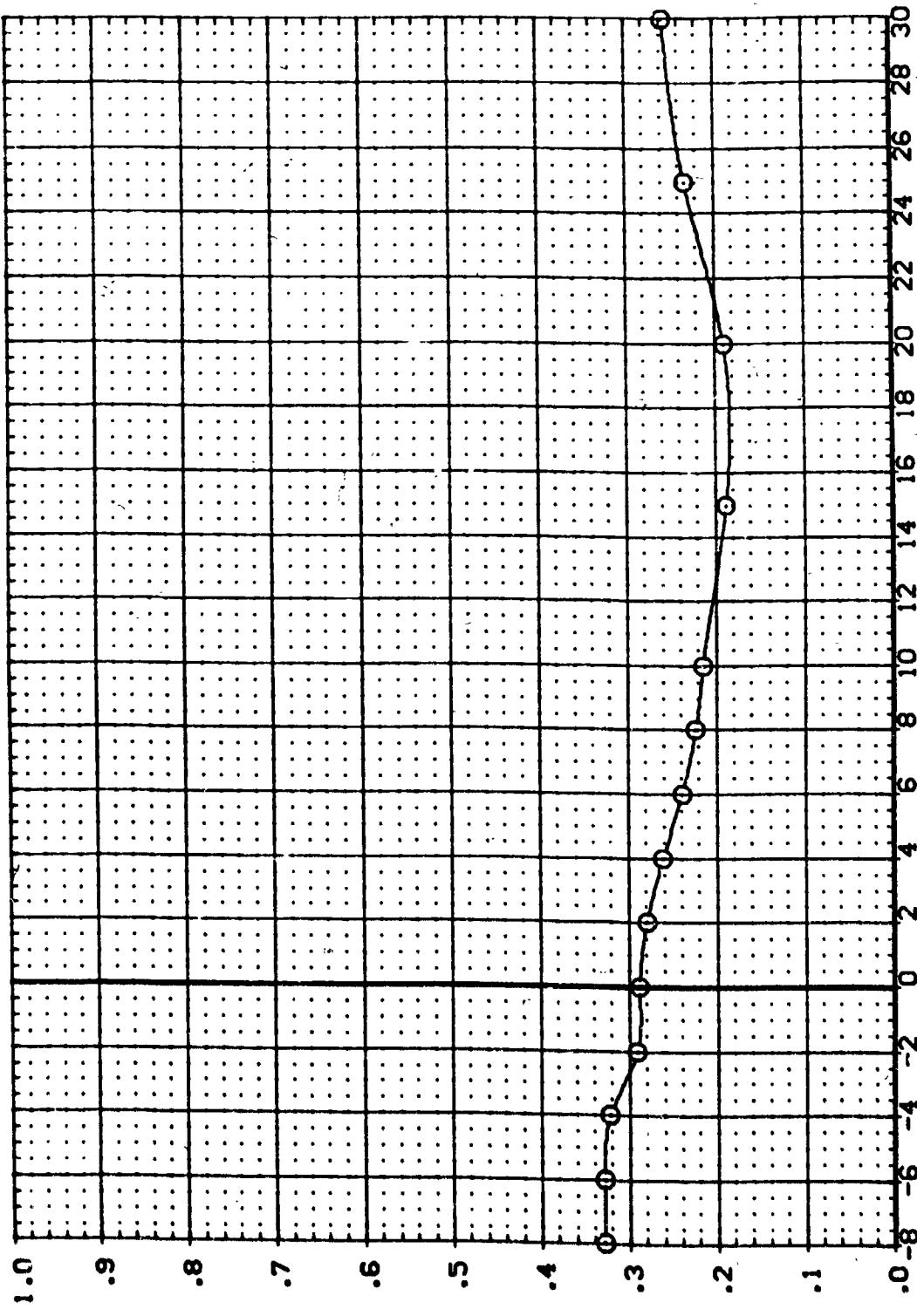


FIG. 20 SUMMARY OF BETA OFFSET EFFECTS IN PITCH, DBETA = 5 DEGS.
 (A)MACH = 7.32

DATA SET SYMBOL: CONFIGURATION DESCRIPTION
(REGO13) C AMES 3.5-175 TAIS DT-L-PI-AW

RUDER .000 AIRLON .000 ELEVON .000 PLATES .000
REFERENCE INFORMATION
REF. 2690 .0000 SQ.FT.
LREF 1290 .3000 IN.
BREF 936 .6800 IN.
XREF 989 .0000 IN.
YREF 67 .0000 IN.
ZREF .0100 SCALE



YAW AERODYNAMIC CENTER. YAC/L

FIG. 20 SUMMARY OF BETA OFFSET EFFECTS IN PITCH, DBETA = 5 DEGS.
(A)MACH = 7.32

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| DATA SET SYMBOL | CONFIGURATION DESCRIPTION |
|-----------------|----------------------------|
| {REGO16} | AMES 3.5-175 IAS OTL+P1+A1 |
| {REGO17} | AMES 3.5-175 IAS OTL+P1+A1 |

| | ALTRN | ELEVN | PLTHS | REFERENCE INFORMATION |
|--------|-------|-------|-------|-----------------------|
| .000 | .000 | .000 | .000 | SREF 2690.0000 IN. |
| 30.000 | .000 | .000 | .000 | LREF 250.3000 IN. |

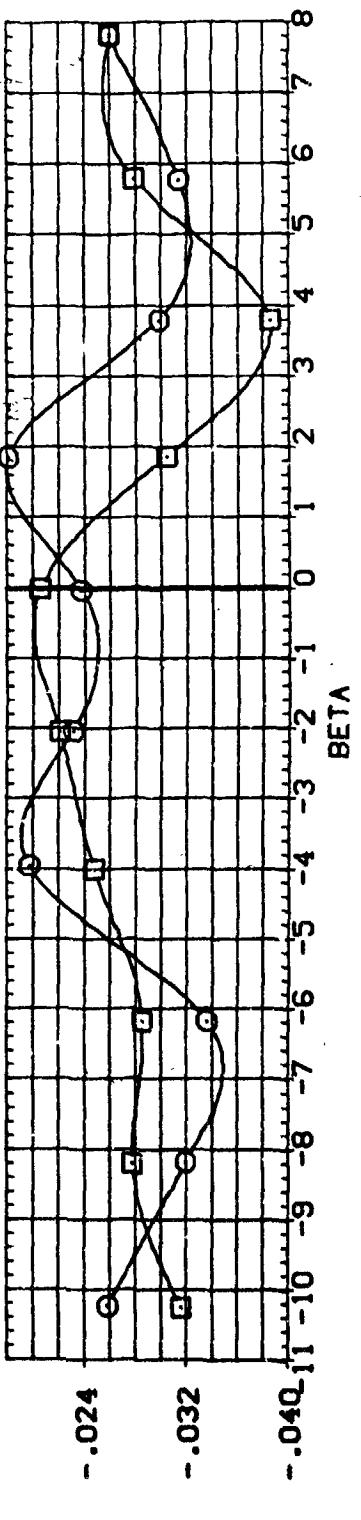
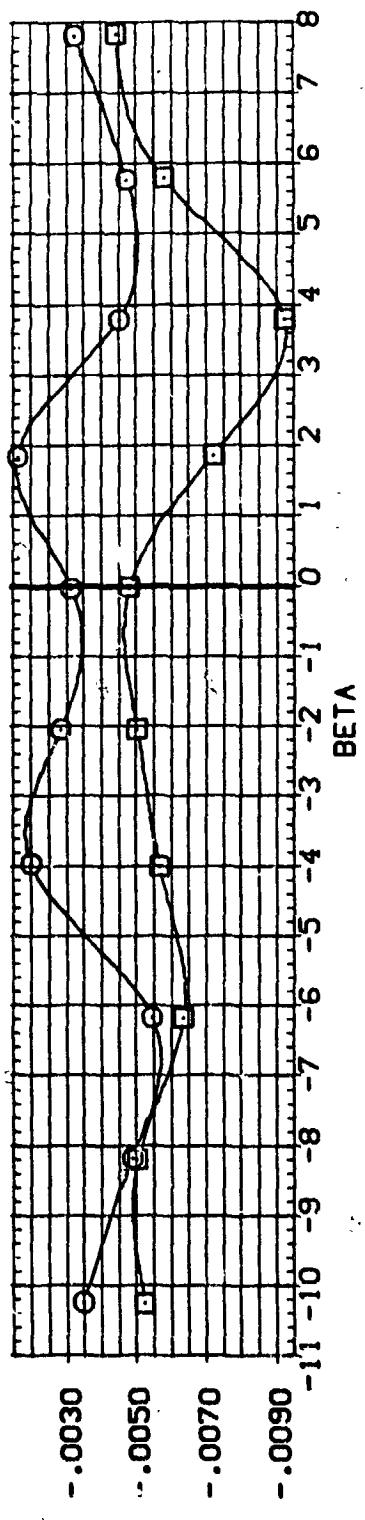
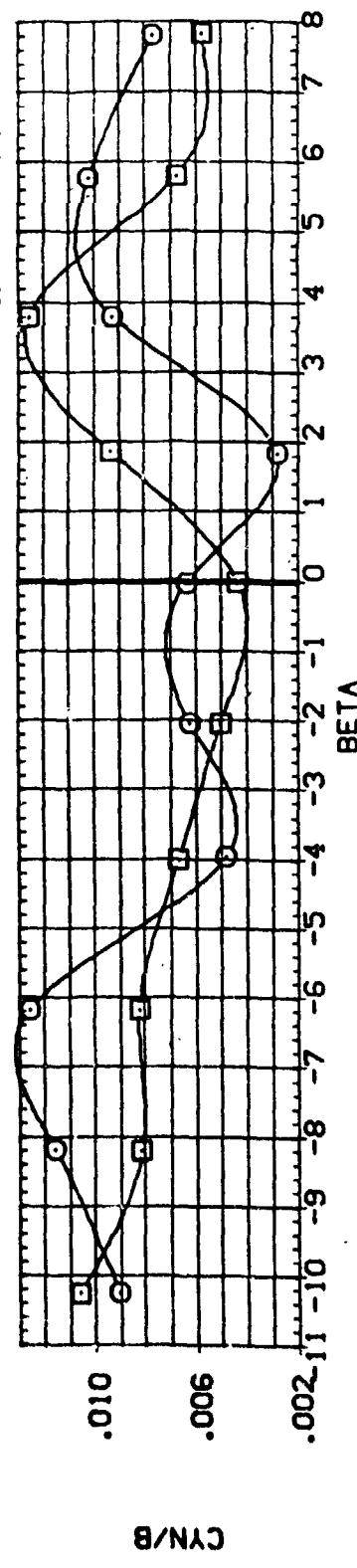


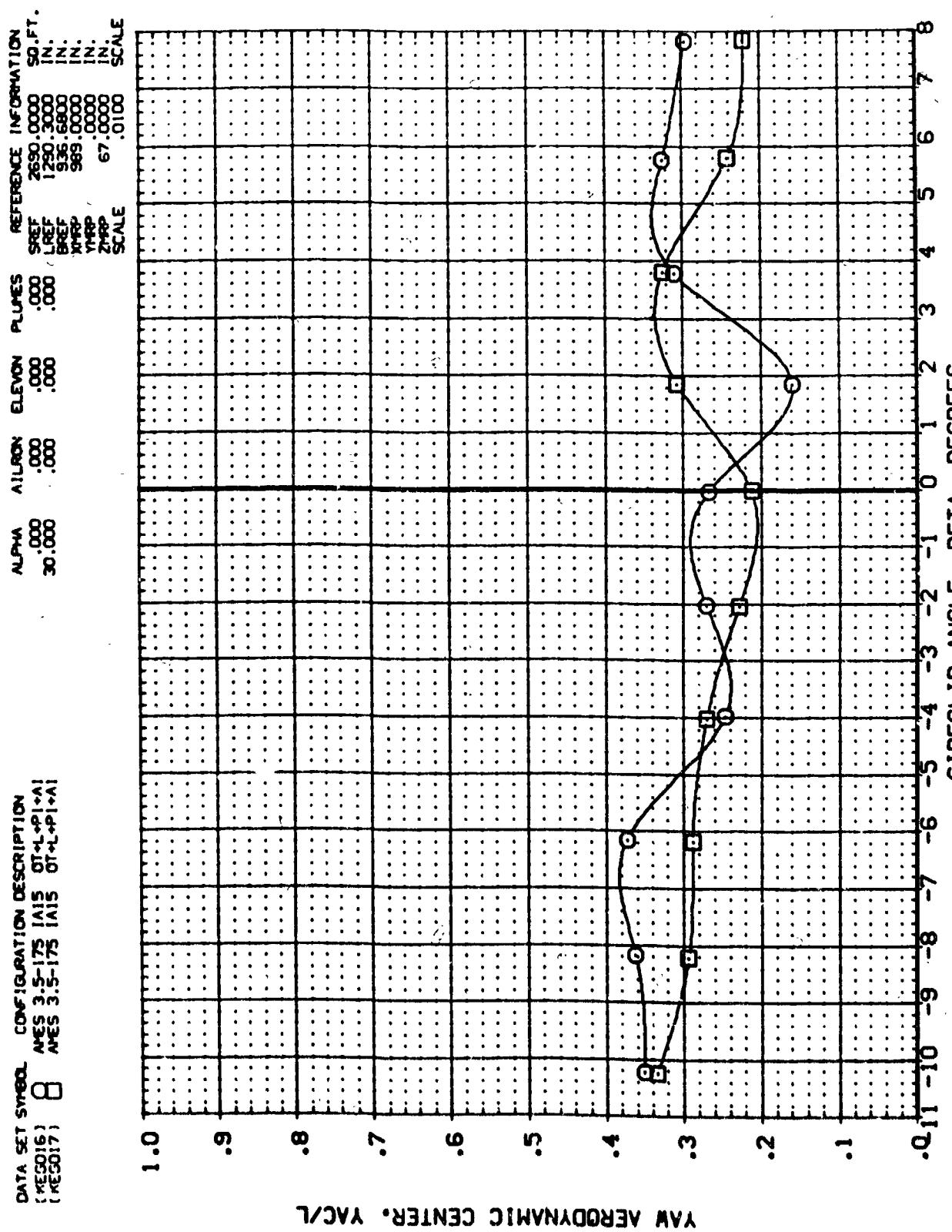
FIG. 21 SUMMARY OF ALPHA OFFSET EFFECTS IN YAW.
 $\alpha_{MACH} = 7.32$

$$(\Lambda)_{\text{MACH}} = 7.32$$

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PAGE 53

FIG. 21 SUMMARY OF ALPHA OFFSET EFFECTS IN YAW.
(MACH = 7.32)



DATA SET SYMBOL: CONFIGURATION DESCRIPTION
 EEE003, O MACH 3.5-175 TA15 GTHL+PI+A14F

REFERENCE INFORMATION
 RUDER .000 AILRDN .000 ELEVON .000 PLUNES .000
 SO.FT.
 SREF 7690.0000 IN.
 LREF 1290.3000 IN.
 BREF 1935.6800 IN.
 YMRF 989.0000 IN.
 ZMRF 67.0000 IN.
 SCALE .0100

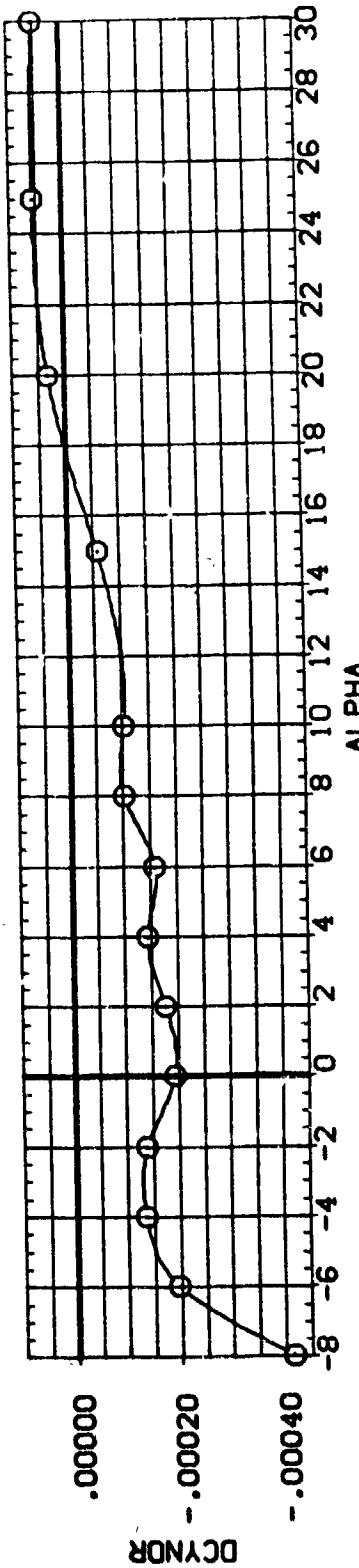
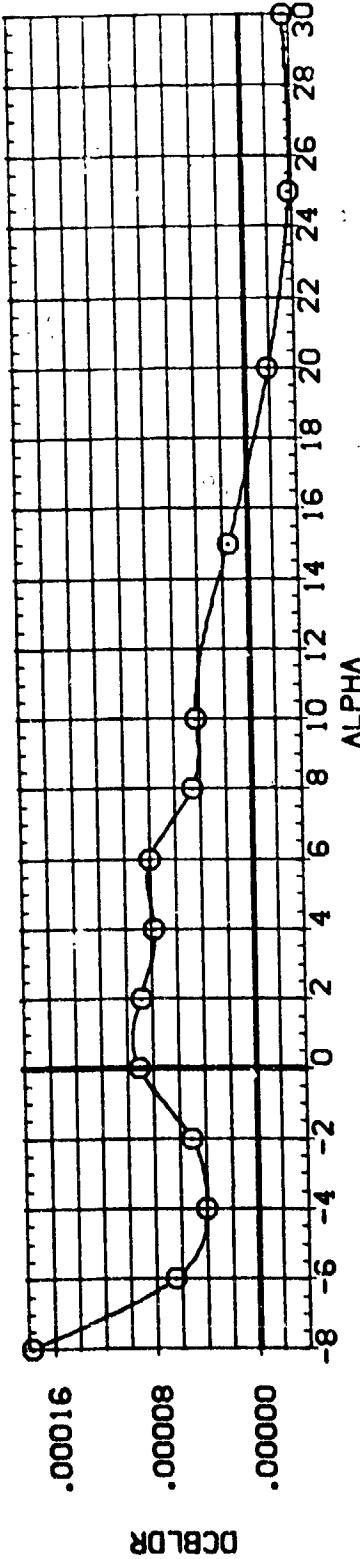
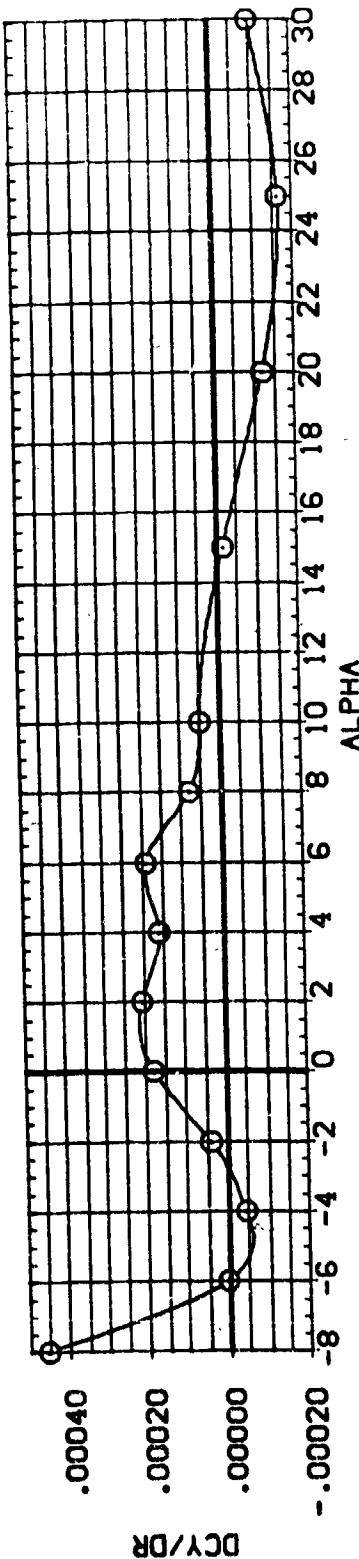


FIG. 22 SUMMARY OF RUDDER EFFECTIVENESS IN PITCH.
 CA/MACH = 7.32

APPENDIX
TABULATED SOURCE DATA

Tabulations of plotted data are available on request from
Data Management Services.

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 28 MAR 74

TAIS ARC 3.5 179

PAGE 1

AMES 3.5-179 TAIS OT-LP1+A1+F

(REG002) (12 FEB 74)

REFERENCE DATA

| | | | |
|---------|------------------|--------|--------------|
| SREF = | 2890.0000 SF.FT. | XREF = | 969.0000 IN. |
| LREF = | 1220.3000 IN. | YREF = | .0000 IN. |
| SREF = | 938.8600 IN. | ZREF = | .67.0000 IN. |
| SCALE = | .0100 SCALE | | |

PARAMETRIC DATA

| | | | |
|----------|------|----------|------|
| BETA = | .000 | ELEVON = | .000 |
| AILRON = | .000 | RUDDER = | .000 |
| PLUMES = | .000 | | |

| RUN NO. | 2 / 0 | RNL = | 2.29 | GRADIENT INTERVAL = | -5.00/ 5.00 |
|----------|--------|---------|---------|---------------------|-------------|
| MACH | ALPHA | ON | CY | CLM | CIN |
| 7.320 | -7.330 | -.23776 | -.01367 | .20595 | .09111 |
| 7.320 | -5.407 | -.19556 | -.01626 | .19419 | .07625 |
| 7.320 | -3.411 | -.14817 | -.01490 | .16124 | .06334 |
| 7.320 | -2.000 | -.11177 | -.01335 | .17183 | .05320 |
| 7.320 | -.411 | -.08523 | -.00645 | .16145 | .03836 |
| 7.320 | 2.327 | -.02015 | -.00542 | .15135 | .02710 |
| 7.320 | 4.814 | -.01808 | -.00526 | .14329 | .01939 |
| 7.320 | 6.815 | -.05987 | -.00331 | .13905 | .00779 |
| 7.320 | 8.563 | -.10119 | -.00550 | .12931 | -.00533 |
| 7.320 | 10.658 | -.14526 | -.00035 | .13041 | -.01991 |
| 7.320 | 15.375 | -.24384 | -.00050 | .11482 | -.04249 |
| 7.320 | 20.726 | -.36872 | -.00161 | .10892 | -.08497 |
| 7.320 | 25.794 | -.52837 | -.00024 | .10426 | -.13084 |
| 7.320 | 31.014 | -.71482 | -.01946 | .11346 | -.22698 |
| GRADIENT | .02080 | .00132 | -.00464 | -.00549 | -.00062 |

GRADIENT

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 26 MAR 74

TA15 ARC 3.5 175

PAGE 3

AMES 3.9-179 TA15 O7L4F1+A1+F

(RECD4) (26 MAR 74)

REFERENCE DATA

SREF = 2691.0000 SQ.FT.
 LREF = 1290.3000 IN.
 BREF = 936.6875 IN.
 SCALE = .1100 SCALE

RNL/L = 2.02 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CY | CA | CLW | CTN | CBL |
|----------|---------|---------|---------|---------|---------|---------|-----|
| 7.320 | | | | | | | |
| -5.427 | -25360 | .01915 | .22163 | .13510 | .00932 | -.00240 | |
| -3.488 | -15946 | -.01972 | .21435 | .11923 | .00649 | -.00247 | |
| -2.150 | -16124 | -.01722 | .20121 | .09700 | .00714 | -.00231 | |
| -.423 | -.05713 | -.01363 | .16343 | .07125 | .00455 | -.00215 | |
| 2.526 | -.04583 | -.01986 | .16973 | .05340 | .00211 | -.00127 | |
| 4.605 | -.01404 | -.00944 | .15872 | .04346 | .00215 | -.00151 | |
| 6.596 | .01146 | -.00937 | .14860 | .1 649 | .00227 | -.00168 | |
| 8.553 | .00673 | -.00573 | .14127 | .01102 | .00233 | -.00164 | |
| 10.495 | .12635 | -.00120 | .13693 | -.00440 | -.00108 | .00131 | |
| 15.614 | .23264 | -.00139 | .12383 | -.02894 | -.00166 | -.00026 | |
| 20.744 | .35421 | -.00115 | .11653 | -.06918 | -.00150 | .00169 | |
| 25.832 | .510932 | .00260 | .11911 | -.12692 | -.00317 | .00135 | |
| 31.016 | .666682 | .01134 | .12364 | -.19639 | -.00594 | .00373 | |
| GRADIENT | .02423 | .02138 | -.01681 | -.00912 | -.00085 | .00014 | |

PARAMETRIC DATA

BETA = .0000 ELEVON = -40.000
 AIRON = .0000 RUDDER = .000
 PLUMES = .000

DATE TO MAR 74

PAGE 9

IA15 ARC 3.5 175

AMES 3.5-175 IA15 OTOL+P1+A1

REFERENCE DATA

SUPER = 2000.0000 34.97.
CP = 1290.2000 IN.
DREF = 936.6000 IN.
SCALE = .0100 SCALE

RUN NO. E/D RNL/Z 1.35 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CY | CA | CLM | CYN | CBL |
|-------|----------|---------|---------|----------|----------|---------|---------|
| 7.320 | -7.450 | -.23273 | -.01651 | .21090 | .08394 | .00377 | -.00316 |
| 7.320 | -5.520 | -.18746 | -.00844 | .20031 | .07134 | .00149 | .00111 |
| 7.320 | -3.571 | -.14521 | -.00551 | .18664 | .05869 | -.00051 | -.00009 |
| 7.320 | -2.231 | -.11513 | -.00844 | .17974 | .04867 | .00124 | -.00111 |
| 7.320 | .340 | -.05769 | -.00845 | .16866 | .03185 | .00147 | -.00146 |
| 7.320 | 2.427 | -.01363 | .002238 | .15998 | .02041 | -.00332 | .00361 |
| 7.320 | 4.476 | .32750 | -.00452 | .15254 | .01099 | -.00093 | -.00166 |
| 7.320 | 6.519 | .57195 | -.00560 | .14559 | -.00241 | .00073 | -.00193 |
| 7.320 | 8.474 | .11613 | -.00162 | .14062 | -.01615 | .01284 | .01023 |
| 7.320 | 10.413 | .15682 | .03771 | .14113 | -.03254 | .00714 | .00273 |
| 7.320 | 15.564 | .26329 | -.00120 | .12879 | -.06205 | .01234 | -.01210 |
| 7.320 | 20.736 | .41382 | .00374 | .12506 | -.11406 | .03451 | .00166 |
| 7.320 | 25.766 | .58511 | .00631 | .12841 | -.119437 | .02614 | .01255 |
| 7.320 | 31.976 | .80548 | .01457 | .14040 | -.29724 | .00889 | .00459 |
| | GRADIENT | .12147 | .007154 | -.033441 | -.01596 | .00126 | .00214 |

(REG000) (12 FEB 74)

PARAMETRIC DATA

BETA = .0000
AILRDN = .0000
PLUMES = .0000
ELEVON = .0000
RUDDER = .0000

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 26 MAR 74

1A19 ARC 3.5 175

APES 3.3-175 1A19 OT+L+P1+A1

REFERENCE DATA

SREF = 2800.0000 50.FFT. XREF = 369.0000 IN.
 LREF = 1251.3073 IN. YREF = .0000 IN.
 BREF = 336.0000 IN. ZREF = 67.0000 IN.
 SCALE = .1150 SCALE

PAGE 6

(REGD7) (12 FEB 74)

PARAMETRIC DATA

| RUN NO. | T / D | RN/L = | 1.98 | GRADIENT INTERVAL = | -3.00/ 3.00 |
|----------|--------|---------|---------|---------------------|-------------|
| MACH | ALPHA | CN | CY | CA | CL,M |
| 7.320 | -7.408 | -311.20 | .008278 | .222340 | .16021 |
| 7.320 | -5.514 | -26430 | .00170 | .20881 | .13905 |
| 7.320 | -3.327 | -20724 | -.00392 | .19150 | .11200 |
| 7.320 | -2.187 | -16637 | -.01167 | .17944 | .09381 |
| 7.320 | -.385 | -10032 | -.00802 | .16339 | .06787 |
| 7.320 | 2.451 | -.09~12 | .00364 | .15166 | .03126 |
| 7.320 | 4.532 | -.00828 | -.00810 | .14168 | .03698 |
| 7.320 | 6.557 | .04132 | .00436 | .13343 | .02393 |
| 7.320 | 8.497 | .06549 | .03876 | .12865 | .01861 |
| 7.320 | 10.486 | .13268 | .03135 | .12622 | -.00870 |
| 7.320 | 15.627 | .24014 | .03255 | .11254 | -.03367 |
| 7.320 | 20.831 | .36860 | .00964 | .10346 | -.07555 |
| 7.320 | 25.834 | .52354 | .03971 | .10842 | -.13422 |
| 7.320 | 31.197 | .71319 | .02179 | .11309 | -.21022 |
| GRADIENT | | .02469 | .02762 | -.01609 | -.00911 |

BETA = .000 ELEVON = -40.000

AILMOM = .000

PLUMES = .000

RUDDER = .000

CBL = .00263

CYN = -.00381

CL,M = .00304

CYN = .00271

CBL = .00157

CYN = .00229

CL,M = .00156

CYN = -.00086

CBL = -.000321

CYN = .00234

CL,M = -.000314

CYN = .00215

CBL = -.000372

CYN = -.00016

CL,M = .00191

CYN = -.001965

CBL = .001263

CYN = -.01715

CL,M = .00125

CYN = -.01407

CBL = .00123

CYN = .002073

CL,M = .00426

CYN = .00362

CBL = .001762

CYN = .00339

CL,M = .00271

CYN = .003677

CBL = -.000253

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 26 MAR 74 TAIS ARC 3.3 175
AMES 3.5-175 TAIS OTL-4P10A1

PAGE 7

(RECD#) (12 FEB 74)

REFERENCE DATA

| | | | | | |
|-------|---|-------------------|------|---|---------------|
| SREF | = | 2600.0000 56.4FT. | XHDP | = | .999.0000 IN. |
| LREF | = | 1250.3000 IN. | YHDP | = | .0000 IN. |
| BREF | = | 936.8000 IN. | ZHDP | = | .67.0000 IN. |
| SCALE | = | .0100 SCALE | | | |

PARAMETRIC DATA

| RUN NO. | E/D | RW/L | = | 1.35 | GRADIENT INTERVAL = | -5.00/ 5.00 |
|----------|--------|---------|---------|---------|---------------------|-------------|
| MACH | ALPHA | CN | CV | CA | CLW | CYN |
| 7.320 | -7.402 | -26760 | -.00499 | .21468 | .11636 | -.00015 |
| 7.320 | -5.507 | -.22036 | .00482 | .20293 | .09986 | -.00373 |
| 7.320 | -3.508 | -.17186 | .01130 | .16992 | .06293 | -.00396 |
| 7.320 | -2.181 | -.13616 | .00262 | .18042 | .06981 | -.00497 |
| 7.320 | .378 | -.07687 | .00022 | .16799 | .05155 | -.00398 |
| 7.320 | 2.520 | -.03410 | .01086 | .15892 | .03840 | -.00282 |
| 7.320 | 4.541 | .02683 | -.00397 | .15059 | .02911 | -.00163 |
| 7.320 | 6.540 | .04608 | .003174 | .14243 | .01790 | -.00193 |
| 7.320 | 8.556 | .06026 | .01327 | .136.6 | .03465 | -.00992 |
| 7.320 | 10.494 | .13103 | .01280 | .13596 | -.00614 | -.01023 |
| 7.320 | 15.651 | .23954 | .00967 | .12315 | -.03416 | -.00252 |
| 7.320 | 20.900 | .36025 | .00729 | .11164 | -.07378 | -.00271 |
| 7.320 | 25.636 | .51963 | .00251 | .11654 | -.13419 | -.01426 |
| 7.320 | 31.177 | .71172 | .01510 | .12379 | -.21146 | -.02612 |
| GRADIENT | | | -.00097 | -.10468 | -.01666 | -.00449 |
| | | | | | | -.00437 |

DATE 20 MAR 74

TASG ARC 7.5 175

PAGE 0

AMES 3.5-175 1A15 OT+L+P1+A1

(RECODE) (12 FEB 74)

REFERENCE DATA

| | |
|-------------------------|----------------------|
| SLEP = 2690.0000 SQ.FT. | XHSP = .989.0000 IN. |
| LECF = 1290.3000 IN. | YHSP = .0000 IN. |
| SLEF = 936.8600 IN. | ZHSP = .67.0000 IN. |
| SCALE = .0100 SCALE | |

PARAMETRIC DATA

| RUN NO. | 9/0 | RNL = | 2.03 | GRADIENT INTERVAL = -5.00/ 5.00 |
|----------|--------|---------|---------|---------------------------------|
| MACH | ALPHA | CN | CA | CLM CYN CBL |
| 7.320 | -7.430 | .24654 | .01061 | .21213 .04616 -.00615 .00701 |
| 7.320 | -5.519 | .20362 | .00239 | .20083 .04510 -.00390 .00464 |
| 7.320 | -3.339 | .15480 | .00255 | .18683 .08929 -.00425 .00439 |
| 7.320 | -2.195 | .12347 | .01038 | .17987 .08620 -.00360 .00595 |
| 7.320 | -.371 | .08623 | .01749 | .16797 .04032 -.01291 .00717 |
| 7.320 | 2.465 | -.02237 | .01672 | .15887 .02696 -.01392 .00741 |
| 7.320 | 4.536 | .01774 | .00906 | .15075 .02044 -.00866 .00467 |
| 7.320 | 6.545 | .08039 | .01157 | .14387 .00861 -.01016 .00955 |
| 7.320 | 8.515 | .10243 | .03657 | .13611 -.00444 -.00663 .00519 |
| 7.320 | 10.483 | .14780 | .01956 | .13796 -.01987 -.00986 .00502 |
| 7.320 | 15.595 | .25322 | .01052 | .12468 -.04660 -.00935 .01559 |
| 7.320 | 20.812 | .36314 | .00932 | .11626 -.09149 -.00913 .01727 |
| 7.320 | 25.026 | .54996 | .00466 | .11983 -.16096 -.00750 .00761 |
| 7.320 | 31.163 | .75426 | .01677 | .12774 -.26892 -.01276 .01532 |
| GRADIENT | .02142 | .02196 | -.00463 | -.00806 -.00211 .00211 |

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 24 MAR 74

1A15 ARC 3.3 175

AMES 3.3-175 1A15 OTL OPT101

REFERENCE DATA

SQFT = 2850.0000 SQ.FT.
 LBF = 1250.0000 IN.
 SQFT = 936.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 10/0 PNL = 2.09 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CY | CA | CLW | CM | CBL |
|----------|--------|---------|--------|---------|---------|---------|--------|
| 7.320 | -7.450 | -.23076 | .01634 | .20043 | .00195 | -.01257 | .00593 |
| | -5.537 | -.19634 | .02210 | .19766 | .07817 | -.01449 | .00899 |
| 7.320 | -3.510 | -.15103 | .01203 | .18517 | .06566 | -.01006 | .00480 |
| | -2.196 | -.12050 | .01766 | .17717 | .05531 | -.01259 | .00376 |
| 7.320 | .374 | -.06434 | .01675 | .16542 | .03864 | -.01270 | .00316 |
| | 2.454 | -.02334 | .01935 | .15965 | .02676 | -.01435 | .00581 |
| 7.320 | 4.507 | .01734 | .01532 | .14826 | .02015 | -.01200 | .00471 |
| | 6.594 | .05946 | .02615 | .14102 | .00632 | -.01655 | .00793 |
| 7.320 | 8.507 | .09993 | .03263 | .13563 | -.00393 | -.02101 | .00924 |
| | 10.443 | .14251 | .01954 | .13546 | -.01786 | -.01392 | .00596 |
| 7.320 | 15.590 | .24846 | .05210 | .12272 | -.04398 | -.01076 | .01354 |
| | 20.812 | .37591 | .04974 | .11526 | -.03782 | -.02800 | .01375 |
| 7.320 | 25.878 | .53934 | .04179 | .11659 | -.15546 | -.02484 | .01174 |
| | 31.164 | .74208 | .07011 | .12375 | -.24215 | -.03854 | .01925 |
| GRADIENT | | .02096 | .00036 | -.00459 | -.00566 | -.00025 | .00000 |

PAGE

(RECD10) (12 FEB 74)

PARAMETRIC DATA

| BETA | ELEVON | AILLON | RUDDER | PLUNES |
|------|--------|--------|--------|--------|
| .000 | .000 | .000 | .000 | .000 |

DATE 26 MAR 74

TA19 ARC 3.5 175

AMES 3.5-175 TA15 OT+L+P1+A1+F

PAGE 11

(REG013) (12 FEB 74)

REFERENCE DATA

| | | | |
|---------|------------------|--------|--------------|
| SREF = | 2690.0000 SQ.FT. | XHFP = | 969.0000 IN. |
| LREF = | 1290.3000 IN. | YHFP = | .0000 IN. |
| BREF = | 936.6000 IN. | ZHFP = | .67.0000 IN. |
| SCALE = | .0100 SCALE | | |

RUN NO. 13/ 0 RN/L = 2.06 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CY | CLM | CYN | CBL |
|-------|----------|----------|---------|----------|---------|----------|
| 7.320 | -6.245 | -1.25107 | -1.1603 | .23119 | .09664 | .05639 |
| 7.320 | -6.269 | -.20237 | -.16413 | .21971 | .08053 | .05442 |
| 7.320 | -4.366 | -.15765 | -.16246 | .21006 | .08745 | .05294 |
| 7.320 | -5.019 | -.13035 | -.15781 | .20144 | .08010 | .04817 |
| 7.320 | -.467 | -.07236 | -.14605 | .16591 | .04025 | .04132 |
| 7.320 | 1.690 | -.02636 | -.13977 | .17432 | .02650 | .03799 |
| 7.320 | 5.703 | .13243 | .16493 | .01586 | .03391 | .01924 |
| 7.320 | 5.732 | .05476 | .12393 | .19726 | .00593 | .02868 |
| 7.320 | 7.704 | .09881 | -.11844 | .14932 | .00764 | .02559 |
| 7.320 | 9.670 | .14405 | -.11435 | .14140 | .02252 | .02287 |
| 7.320 | 14.811 | .24698 | -.11177 | .13328 | .04918 | .01903 |
| 7.320 | 19.961 | .38693 | -.10817 | .12756 | .10114 | .01914 |
| 7.320 | 23.006 | .54706 | -.11797 | .12689 | .16533 | .02537 |
| 7.320 | 30.281 | .741394 | -.11692 | .13176 | .2711 | .02732 |
| | GRADIENT | .02157 | .00375 | -.011561 | .016558 | -.012231 |

AMES 3.5-175 TA15 OT+L+P1+A1+F

REFERENCE DATA

| | | | |
|---------|------------------|--------|--------------|
| SREF = | 2690.0000 SQ.FT. | XHFP = | 969.0000 IN. |
| LREF = | 1290.3000 IN. | YHFP = | .0000 IN. |
| BREF = | 936.6000 IN. | ZHFP = | .67.0000 IN. |
| SCALE = | .0100 SCALE | | |

RUN NO. 14/ 0 RN/L = 1.96 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CLM | CYN | CBL |
|-------|----------|---------|---------|---------|--------|---------|
| 7.320 | -10.243 | -.07851 | .29158 | .20511 | .03676 | .09737 |
| 7.320 | -8.193 | -.07844 | .23415 | .19654 | .03766 | .07762 |
| 7.320 | -6.130 | -.07892 | .17468 | .19030 | .03970 | .05669 |
| 7.320 | -3.983 | -.08363 | .12395 | .16575 | .04142 | .04210 |
| 7.320 | -2.013 | -.08182 | .06982 | .17973 | .04184 | .02382 |
| 7.320 | -.049 | -.07453 | .01723 | .17650 | .03987 | .00311 |
| 7.320 | 1.828 | -.07634 | -.03070 | .17885 | .04157 | .00453 |
| 7.320 | 5.757 | -.07601 | -.07457 | .16212 | .04320 | .01466 |
| 7.320 | 5.729 | -.07446 | -.13167 | .18650 | .04246 | .03246 |
| 7.320 | 7.759 | -.07293 | -.19316 | .19306 | .04114 | .05337 |
| | GRADIENT | .00253 | -.02575 | -.00117 | .00175 | -.00343 |

(REG014) (12 FEB 74)

PARAMETRIC DATA

| | | | |
|----------|------|----------|-------|
| ALPHA = | 0.00 | ELEVON = | .0000 |
| AIRON = | .000 | RUDER = | .0000 |
| PLUMES = | .000 | | |

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 26 MAR 74 1A15 ARC 3.5 175

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AMES 3.5-175 1A15 OT+L+P1+A1+F

(RECD 1) (12 FEB 74)

REFERENCE DATA

SREF = 2890.0000 SQ.FT. XREF = 969.0000 IN.
L*EF = 1290.3000 IN. YREF = .0000 IN.
S*EF = 936.6802 IN. ZREF = 67.0000 IN.
SCALE = .0100 SCALE

RUN NO. 15 / 0 RNL/L = 2.07 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|-------|----------|----------|---------|---------|--------|---------|----------|
| 7.320 | -10.256 | -0.07551 | .27561 | .19992 | .03595 | -.00630 | .03591 |
| 7.320 | -8.168 | -0.07501 | .22270 | .19240 | .03660 | -.01717 | .02878 |
| 7.320 | -6.141 | -0.07346 | .16552 | .16570 | .03772 | -.02163 | .02997 |
| 7.320 | -4.013 | -0.07498 | .11810 | .17944 | .03680 | -.03685 | .01527 |
| 7.320 | -2.017 | -0.07473 | .06395 | .17561 | .03910 | -.01991 | .00658 |
| 7.320 | -1.085 | -0.07133 | .01076 | .17259 | .03801 | -.00323 | .01186 |
| 7.320 | 1.846 | -0.07487 | -.04193 | .17807 | .04163 | -.01295 | .01680 |
| 7.320 | 3.794 | -0.07021 | -.09155 | .17874 | .04059 | .02658 | -.01362 |
| 7.320 | 5.783 | -0.06837 | -.14198 | .10380 | .04187 | .04135 | -.02145 |
| 7.320 | 7.765 | -0.06392 | -.20751 | .19016 | .03952 | .06337 | -.03087 |
| 7.320 | GRADIENT | .00039 | -.02676 | -.00X05 | .0X032 | .0X1816 | -.0X0376 |

AMES 3.5-175 1A15 OT+L+P1+A1

(RECD 1) (12 FEB 74)

REFERENCE DATA

SREF = 2890.0000 SQ.FT. XREF = 969.0000 IN.
L*EF = 1290.3000 IN. YREF = .0000 IN.
S*EF = 936.6802 IN. ZREF = 67.0000 IN.
SCALE = .0100 SCALE

RUN NO. 16 / 0 RNL/L = 2.08 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|-------|----------|----------|---------|---------|--------|---------|---------|
| 7.320 | -10.230 | -0.07981 | .29111 | .19289 | .03751 | -.09737 | .04171 |
| 7.320 | -8.185 | -0.07923 | .23412 | .14864 | .03814 | -.07715 | .03361 |
| 7.320 | -6.163 | -0.07948 | .16427 | .18036 | .04020 | -.01339 | .02242 |
| 7.320 | -3.972 | -0.07778 | .10442 | .17550 | .04047 | -.03178 | .01364 |
| 7.320 | -2.030 | -0.08000 | .06921 | .17203 | .04174 | -.02392 | .01071 |
| 7.320 | -1.021 | -0.07485 | -.01483 | .17056 | .04047 | -.00668 | .00255 |
| 7.320 | 1.834 | -0.07618 | -.01950 | .17207 | .04208 | -.03199 | .00165 |
| 7.320 | 3.800 | -0.07500 | -.06597 | .17405 | .04196 | .01073 | .013649 |
| 7.320 | 5.771 | -0.07161 | -.13021 | .17811 | .04149 | .03181 | .01663 |
| 7.320 | 7.751 | -0.07022 | -.16708 | .18369 | .04019 | .04923 | -.02431 |
| 7.320 | GRADIENT | .00249 | -.02214 | -.00D15 | .00017 | .00546 | -.01266 |

(RECD 1) (12 FEB 74)

DATE 24 MAR 74

TA15 ARC 3.5 175

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AMES 3.5-175 TA15 OT+L+P1+A1

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XREF = 989.0000 IN.
 LREF = 1290.3000 IN. YREF = .0000 IN.
 BREF = 936.6000 IN. ZREF = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 17/ 0 RN/L = 2.02 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|-------|----------|--------|---------|---------|---------|----------|--------|
| 7.320 | -10.254 | .85562 | .29891 | .13349 | -.30828 | -.109146 | .06546 |
| 7.320 | -8.210 | .84410 | .23704 | .12942 | -.32020 | -.07154 | .03487 |
| 7.320 | -6.174 | .82236 | .18259 | .12215 | -.28605 | -.05637 | .04384 |
| 7.320 | -4.039 | .81046 | .12024 | .12393 | -.27504 | -.03796 | .02936 |
| 7.320 | -2.039 | .79765 | .07834 | .11654 | -.26947 | -.02845 | .02101 |
| 7.320 | -.019 | .79346 | .03124 | .11896 | -.26535 | -.01721 | .01896 |
| 7.320 | 1.865 | .79372 | -.00970 | .11727 | -.26772 | -.01779 | .01061 |
| 7.320 | 3.822 | .81201 | -.18455 | .12389 | -.27789 | -.01766 | .01852 |
| 7.320 | 5.796 | .83298 | -.14841 | .12497 | -.29194 | -.03560 | .03321 |
| 7.320 | 7.834 | .85691 | -.20263 | .12997 | -.31598 | -.04317 | .04317 |
| | GRADIENT | .03033 | -.02545 | -.00019 | .00676 | -.00596 | |

AMES 3.5-175 TA15 OT+L+P1+A1

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XREF = 989.0000 IN.
 LREF = 1290.3000 IN. YREF = .0000 IN.
 BREF = 936.6000 IN. ZREF = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 18/ 0 RN/L = 2.49 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|-------|----------|---------|----------|---------|--------|----------|---------|
| 7.320 | -10.238 | .25849 | -.07778 | .23884 | .10043 | .27187 | -.16587 |
| 7.320 | -8.207 | -.27736 | 4.09108 | .25027 | .11175 | -.14450 | 1.08805 |
| 7.320 | -6.165 | -.27513 | 3.87041 | .24632 | .10925 | -1.08523 | 1.02357 |
| 7.320 | -4.022 | -.27290 | 3.45612 | .24126 | .10628 | -.97124 | .93877 |
| 7.320 | -2.070 | -.27219 | 1.46150 | .22965 | .10281 | -.40369 | .38667 |
| 7.320 | -.061 | -.25916 | -.793176 | .20212 | .09674 | -.22843 | -.20355 |
| 7.120 | 1.810 | -.29327 | 5.11385 | .20288 | .11378 | -.14524 | 1.31408 |
| 7.320 | 3.736 | -.29247 | -.03473 | .22218 | .10425 | .001516 | -.02382 |
| 7.320 | 5.668 | -.31287 | 4.74227 | .24782 | .12246 | -.13502 | 1.21084 |
| 7.320 | 7.759 | -.31725 | 3.36493 | .24905 | .12308 | -.96888 | .83671 |
| | GRADIENT | -.01026 | -.17435 | -.00089 | .00033 | .04769 | -.01912 |

(REGD17) (12 FEB 74)

(REGD18) (12 FEB 74)

(REGD17) (12 FEB 74)

(REGD18) (12 FEB 74)

AMES 3.5-175 TA15 OT+L+P1+A1+F PLUMES ON

(REC019) (12 FEB 74)

REFERENCE DATA

SREF = 2690.0000 50.FT.
 LREF = 1290.3200 IN.
 BREF = 936.6600 IN.
 SCALE = .0100 SCALE

RUN NO. 19/ 0 RN/L = 2.01 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|----------|---------|----------|---------|--------|--------|---------|--------|
| 7.320 | -10.265 | -0.07448 | .25638 | .16852 | .03084 | -.07904 | .03207 |
| 7.320 | -8.245 | -.07755 | .20235 | .16336 | .03308 | -.05966 | .02447 |
| 7.320 | -6.183 | -.07656 | .15125 | .15940 | .03492 | -.04277 | .01741 |
| 7.320 | -3.978 | -.07725 | .10026 | .15534 | .03637 | -.02617 | .01157 |
| 7.320 | -2.041 | -.07754 | .04851 | .15246 | .0352 | -.01108 | .00420 |
| 7.320 | -.058 | -.06333 | .00721 | .14421 | .03179 | -.00366 | .00178 |
| 7.320 | 1.826 | -.07414 | -.03567 | .15267 | .03526 | -.00159 | .00152 |
| 7.320 | 3.764 | -.07749 | -.08935 | .13713 | .03830 | -.02174 | .01131 |
| 7.320 | 5.744 | -.07440 | -.14505 | .15990 | .03724 | -.01922 | .00854 |
| 7.320 | 7.766 | -.07411 | -.20105 | .16533 | .03628 | -.02697 | .00531 |
| GRADIENT | .02922 | -.02394 | .00216 | .00216 | .00590 | -.01276 | |

AMES 3.5-175 TA15 OT+L+P1+A1+F PLUMES ON

(REC020) (12 FEB 74)

REFERENCE DATA

SREF = 2690.0000 50.FT.
 LREF = 1290.3200 IN.
 BREF = 936.6600 IN.
 SCALE = .0100 SCALE

RUN NO. 20/ 0 RN/L = 1.83 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|----------|---------|---------|---------|--------|--------|---------|--------|
| 7.320 | -10.263 | -.07414 | .25952 | .18050 | .03290 | -.07935 | .03216 |
| 7.320 | -8.254 | -.07814 | .20736 | .17441 | .03479 | -.05132 | .02527 |
| 7.320 | -6.174 | -.07825 | .15316 | .17010 | .03711 | -.04347 | .01769 |
| 7.320 | -3.974 | -.06108 | .10116 | .16578 | .03970 | -.02763 | .01125 |
| 7.320 | -2.052 | -.07634 | .05253 | .16229 | .03766 | -.01425 | .00578 |
| 7.320 | -.082 | -.06610 | .01127 | .15557 | .03471 | -.00749 | .00266 |
| 7.320 | 1.852 | -.07865 | -.05887 | .16280 | .03987 | -.01610 | .00415 |
| 7.320 | 3.774 | -.07783 | -.09115 | .16722 | .04109 | -.02230 | .01199 |
| 7.320 | 5.719 | -.07915 | -.15271 | .17224 | .04354 | -.02200 | .01387 |
| 7.320 | 7.773 | -.07554 | -.21445 | .17689 | .04014 | -.02771 | .01112 |
| GRADIENT | .000032 | -.02453 | .00216 | .00216 | .00619 | -.00290 | |

AMES 3.5-175 TA15 OT+L+P1+A1+F PLUMES ON

(REC021) (12 FEB 74)

REFERENCE DATA

ALPHA = .000
 AIRON = .000
 PLUMES = 1.000

RUN NO. 20/ 0 RN/L = 1.83 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|----------|---------|---------|---------|--------|--------|---------|--------|
| 7.320 | -10.263 | -.07414 | .25952 | .18050 | .03290 | -.07935 | .03216 |
| 7.320 | -8.254 | -.07814 | .20736 | .17441 | .03479 | -.05132 | .02527 |
| 7.320 | -6.174 | -.07825 | .15316 | .17010 | .03711 | -.04347 | .01769 |
| 7.320 | -3.974 | -.06108 | .10116 | .16578 | .03970 | -.02763 | .01125 |
| 7.320 | -2.052 | -.07634 | .05253 | .16229 | .03766 | -.01425 | .00578 |
| 7.320 | -.082 | -.06610 | .01127 | .15557 | .03471 | -.00749 | .00266 |
| 7.320 | 1.852 | -.07865 | -.05887 | .16280 | .03987 | -.01610 | .00415 |
| 7.320 | 3.774 | -.07783 | -.09115 | .16722 | .04109 | -.02230 | .01199 |
| 7.320 | 5.719 | -.07915 | -.15271 | .17224 | .04354 | -.02200 | .01387 |
| 7.320 | 7.773 | -.07554 | -.21445 | .17689 | .04014 | -.02771 | .01112 |
| GRADIENT | .000032 | -.02453 | .00216 | .00216 | .00619 | -.00290 | |

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

(REC021) (12 FEB 74)

IA15 ARC 3.5 175

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XHREF = 989.0000 IN.
 LREF = 1280.3000 IN. YHREF = .0000 IN.
 BREF = 936.6800 IN. ZHREF = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 21/ 0 RN/L = 2.03 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | BETA | CN | CY | CA | CLM | CYN | CBL |
|----------|---------|---------|---------|--------|--------|---------|---------|
| 7.320 | -10.239 | -.18019 | .218901 | .17541 | .03532 | -.07683 | .03241 |
| 7.320 | -8.185 | -.08343 | .20657 | .17162 | .03648 | -.10591 | .02527 |
| 7.320 | -6.176 | -.06346 | .15559 | .16751 | .03985 | -.04273 | .01863 |
| 7.320 | -3.985 | -.18430 | .18115 | .16382 | .04142 | -.02749 | .01168 |
| 7.320 | -1.997 | -.08243 | .15912 | .16140 | .04025 | -.01141 | .00463 |
| 7.320 | -.184 | -.06918 | .16700 | .15421 | .03499 | -.00172 | .001252 |
| 7.320 | 1.832 | -.18298 | .03879 | .16082 | .04242 | -.00404 | .00150 |
| 7.320 | 3.821 | -.17819 | .15937 | .16466 | .04165 | .01984 | -.01114 |
| 7.320 | 5.724 | -.17665 | -.14812 | .16787 | .04228 | .03752 | -.01971 |
| 7.320 | 7.756 | -.17349 | -.21916 | .17173 | .03911 | .05654 | -.02861 |
| GRADIENT | .01760 | -.02428 | .02916 | .02913 | .02574 | -.01280 | |

AMES 3.5-175 IA15 OT+L+P1+A1 FLUMES ON

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XHREF = 989.0000 IN.
 LREF = 1280.3000 IN. YHREF = .0000 IN.
 BREF = 936.6800 IN. ZHREF = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 22/ 0 RN/L = 2.29 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CY | CA | CLM | CYN | CBL |
|----------|--------|---------|---------|---------|----------|---------|---------|
| 7.320 | -7.340 | -.23926 | -.02153 | .19917 | .08773 | .01636 | -.00354 |
| 7.320 | -5.380 | -.19143 | -.02173 | .18901 | .07356 | .01064 | -.00324 |
| 7.320 | -3.410 | -.14826 | -.01883 | .17816 | .06043 | .00633 | -.00321 |
| 7.320 | -2.051 | -.11569 | -.01812 | .17044 | .04993 | .00455 | -.00272 |
| 7.320 | -.458 | -.06165 | -.01789 | .15859 | .03466 | .00391 | -.00313 |
| 7.320 | 2.584 | -.01987 | -.01850 | .15151 | .02600 | .00442 | -.00358 |
| 7.320 | 4.646 | .02168 | -.01353 | .14469 | .01740 | .01336 | -.00251 |
| 7.320 | 6.630 | .05927 | -.01378 | .13784 | .01777 | .01293 | -.00269 |
| 7.320 | 8.561 | -.10163 | -.01335 | .13306 | -.018472 | .01362 | -.00268 |
| 7.320 | 10.464 | .14162 | -.01190 | .12879 | -.01620 | .01280 | -.00214 |
| GRADIENT | .02187 | .03049 | -.02442 | -.03028 | -.03028 | -.01002 | |

(REC022) (12 FEB 74)

PARAMETRIC DATA

ALPHA = .000
 AIRON = .000
 PLUMES = 1.000

ELEVON = .000

RUDDER = .000

FLUMES = 1.000

(REC022) (12 FEB 74)

PARAMETRIC DATA

BETA = .000
 AIRON = .000
 PLUMES = 1.000

IA15 ARC 3.5 175

(REG023) (12 FEB 74)

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XHGP = 969.0000 IN.
 LREF = 1290.3000 IN. YHGP = .0000 IN.
 BREF = 936.6000 IN. ZHGP = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 23/ G RNL = 2.20 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CT | CA | CLW | CYN | CBL |
|-------|----------|---------|---------|---------|---------|----------|--------|
| 7.320 | -7.352 | -23476 | -.00347 | .20125 | .08467 | -.001227 | .00277 |
| 7.320 | -5.394 | -19122 | -.00336 | .19034 | .07193 | -.001935 | .00243 |
| 7.320 | -5.373 | -14296 | -.00420 | .17966 | .05757 | -.001371 | .00199 |
| 7.320 | -2.067 | -11291 | -.00397 | .17207 | .04758 | -.001259 | .00139 |
| 7.320 | -4.66 | -26141 | -.00141 | .16101 | .03349 | -.001452 | .00189 |
| 7.320 | 2.580 | -16162 | -.00257 | .15250 | .02342 | -.001327 | .00112 |
| 7.320 | 4.626 | -12265 | -.00319 | .14573 | .01525 | -.001214 | .00166 |
| 7.320 | 6.634 | -65539 | -.00164 | .13816 | .00373 | -.001242 | .00185 |
| 7.320 | 8.514 | -117359 | -.00341 | .13321 | -.00734 | -.001339 | .00153 |
| 7.320 | 10.712 | -15.61 | -.00311 | .12697 | -.02141 | -.00113 | .00157 |
| 7.320 | GRADIENT | .02167 | -.00317 | -.00421 | .02011 | -.00114 | .00114 |

AMES 3.5-175 IA15 OT+L+P1+A1+F PLUMES ON

(REG024) (12 FEB 74)

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XHGP = 969.0000 IN.
 LREF = 1290.3000 IN. YHGP = .0000 IN.
 BREF = 936.6000 IN. ZHGP = 67.0000 IN.
 SCALE = .0100 SCALE

RUN NO. 24/ G RNL = 2.17 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CN | CT | CA | CLW | CYN | CBL |
|-------|----------|--------|---------|---------|---------|----------|---------|
| 7.320 | -7.350 | -23749 | -.01216 | .20077 | .08629 | -.00297 | -.00115 |
| 7.320 | -5.372 | -16941 | -.00820 | .19006 | .07214 | -.001332 | .00241 |
| 7.320 | -3.431 | -14645 | -.01460 | .17914 | .05930 | -.001478 | -.00204 |
| 7.320 | -2.077 | -11433 | -.01097 | .17134 | .04624 | -.001212 | -.00199 |
| 7.320 | -4.452 | -5883 | -.00549 | .15993 | .03322 | -.001194 | .00212 |
| 7.320 | 2.562 | -16165 | -.00773 | .15159 | .02453 | -.001174 | -.00186 |
| 7.320 | 4.657 | -23132 | -.00947 | .14484 | .01935 | -.001193 | -.00135 |
| 7.320 | 6.624 | -16362 | -.00626 | .13764 | .00512 | -.001211 | -.00118 |
| 7.320 | 8.556 | -10469 | -.00471 | .13263 | -.00723 | -.001087 | -.00139 |
| 7.320 | 10.487 | -14737 | -.00217 | .12895 | -.01972 | -.00109 | -.00144 |
| 7.320 | GRADIENT | .02046 | -.00385 | -.00422 | -.00533 | -.00032 | -.00117 |

PARAMETRIC DATA

BETA = .000 ELEVON = .000
 AILRON = .000 RUDDER = .000
 FLUMES = 1.000

PARAMETRIC DATA

BETA = .000 ELEVON = .000
 AILRON = .000 RUDDER = .000
 FLUMES = 1.000

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR

DATE 26 MAR 74

TA15 ARC 3.5 175

AMES 3.5-175 TA15 OTL+P1+H1

PLUMES ON

PAGE 17

(RECORD) (12 FEB 74)

REFERENCE DATA

| | | | | |
|-------|---|------------------|--------|--------------|
| SECF | = | 2690.0000 SQ.FT. | XMRF = | 989.0000 IN. |
| LREF | = | 1290.3000 IN. | YMRF = | .0000 IN. |
| BREF | = | 9: .4800 IN. | ZMRF = | 67.0000 IN. |
| SCALE | = | .0100 SCALE | | |

PARAMETRIC DATA

| | | | | |
|---------------------------------|--------|---------|----------|---------|
| BETA | = | .000 | ELEVON = | 15.000 |
| AILRON | = | .000 | RUDER = | .000 |
| PLUMES | = | 1.0100 | | |
| GRADIENT INTERVAL = -5.00/ 5.00 | | | | |
| RUN NO. | 25/ 0 | RHL/L = | 1.00 | |
| MACH | ALPHA | CN | CY | CLM |
| 7.321 | -7.434 | -.23100 | -.01112 | .20010 |
| 7.321 | -5.522 | -.18626 | -.01333 | .19050 |
| 7.321 | -3.515 | -.14242 | -.00882 | .17920 |
| 7.321 | -2.171 | -.11103 | -.00730 | .17182 |
| 7.321 | .380 | -.05335 | -.02476 | .16358 |
| 7.321 | 2.496 | -.01210 | -.00432 | .15328 |
| 7.321 | 4.549 | -.00805 | -.001724 | .14740 |
| 7.321 | 6.541 | -.07355 | -.001793 | .14133 |
| 7.321 | 8.501 | -.11453 | -.002203 | .13728 |
| 7.321 | 10.467 | -.15625 | -.00157 | .13334 |
| 7.321 | 15.533 | -.26539 | -.00199 | .12605 |
| 7.321 | 20.722 | .41313 | .00584 | .11974 |
| 7.321 | 25.796 | .58005 | .00610 | .11260 |
| 7.321 | 31.562 | .80370 | .00808 | .11315 |
| GRADIENT | | .02114 | .00129 | .00393 |
| | | | | -.00555 |
| | | | | -.00017 |
| | | | | -.00001 |